


## STANDARD AIRCRAFT CHARACTERISTICS AD-4B "SKYRAIDER"

DOUGLAS



MISSION AND DESCRIPTION
The primary mission of the $A D-4 B$ is the destruction of sea and ground targets by dive bombing attacks. The airplane is also capable of torpedo, glide bombing, rocket attacks and tactical support missions.

The airplane is conventional in design and structure. The landing gear, canopy, flaps, wing folding, and three fuselage dive brakes are hydraulically operated. Flaps are of the single slotted trailing edge type. The pres-sure-balance type allerons are operated by power boost, the rudder is equipped with a spring tab system and the longitudinal trim is achieved by an electrically adjustable stabilizer.
Oxygen for five hours is supplied. Twenty gallons of $A D 1$ fluid are supplied for injection

## DEVELOPMENT

First flight....August 1952
Service use.......July 1953

| WEIGHTS |  |
| :---: | :---: |
| LOADINGS | IBS E.F. |
| EMPTY. | .11,837.... |
| BASIC. | .13.364..... |
| DESIGT. | .15.595.07.0 |
| combat. | .15.952..6.8 |
| MAX.T.O. | $\begin{array}{r} 25,000 \ldots 1 \\ .25,000 \ldots \ldots \end{array}$ |
| MAX.LANSD. | $\left\{\begin{array}{l} 21,000 . \ldots .1 \\ .17,500.6 .1 \end{array}\right.$ |
| All weight | calculated. |


| FUEL AND OIL |  |
| :---: | :---: |
| No. Tanks | Total Gal |

FUEL GRADE....115/145
FVEL SPEC.MIL-F-5572

## $01 L$

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CAPACITY(Gal s)........... }3
GRADE....................1120
SPIC................MIL-0-6082A
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| PERFORMANCE SUMMARY |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: |
| TAKE-OTF LOADING CONDITION | $\begin{aligned} & \text { (1) artack } \\ & 1-1660 \mathrm{Ib} \\ & \text { store } \end{aligned}$ | $\begin{aligned} & \text { (3.) ATTACK } \\ & 1-16601 \text { s.ore sex. } \\ & 2=150 \text { gal. ext. } \\ & \text { tanks } \end{aligned}$ | $\left\lvert\, \begin{gathered} 5) \text { MI ACK } \\ 1-1660 \text { Ib store } \\ 2-300 \text { Gal. ext. } \\ \text { tanks } \end{gathered}\right.$ | $\begin{aligned} & (\mathrm{b}) \text { ATMACK } \\ & 1-32501 \mathrm{~b} \\ & \text { store } \end{aligned}$ | $\begin{array}{\|l\|} \hline(7) \text { ATMACK } \\ 1-3250 \mathrm{Ib} \text { store } \\ 2-300 \text { gal. ext } \\ \text { tanks } \end{array}$ |
| TAKIm OFF WHiC Gill | 18.524 | 20,504 | 22,494 | 20,1.20 | 24,090 |
| Frel Ib. | 2,280 | 4,080 | 5,880 | 2,280 | 5,880 |
| Payload 1bo | 1,660 | 1,660 | 1,660 | 3,250 | 3,2.50 |
| Wing loading 1bofsqoft. | 46.3 | 51.3 | 56.? | 50.3 | 60.2 |
| Stall speed - powermoff lon. | 81.6 | 85.9 | 89.9 | 85.0 | 93.1 |
| Takeoff run at S.I. - calm ft. | 840 | 1,140 | 1.770 | 1,060 | 2,510 |
| Takeoff mun at SoL. 25 kn , wind fto | 395 | 570 | 905 | 525 | 1.300 |
| Take-off to clear 50 ft . - calm ft. | - | - | - | - | 1. |
| Max speed/altitude (A) lono/ft. | 288/19,600 | 274/19,300 | 264/19,200 | 287/19,600 | 261/19,200 |
| Rate of climb at S.L. (A) fpm | 2,060 | 1,680 | 1,380 | 1,800 | 1.190 |
| T1me: S.I. to 10,000 fit. (A) min. | 5.2 | 1.6 | 8.4 | 6.0 | 9.7 |
| Time: S.L. to $20,000 \mathrm{ft}$. (A) min. | 12.8 | 17.3 | 24.3 | 15.4 | 32.4 |
| Service ceiling (100 fpm) (A) fit. | 27.400 | 24,300 | 21,700 | 25,600 | 20,400 |
| Combat range nomio | 725 | 1,360 | 1,875 . | 670 | 1,740 |
| Average cruising speed kn. | 175 | 175 | 175 | 180 | 185 |
| Cruising altitude(s) ft. | 5,000 | 5,000 | 5,000 | 5,000 | 5,000 |
| Combat radius nomi. | 270 | 610 | 670(B) | 255 | 665(B) |
| Avorage cruising speed ka. | 170 | 170 | 175 | 170 | 175 |
| Mission time hr. | 3.4 | 7.4 | 7.9 | 3.2 | 7.8 |
| COMBAT LOADING CONDITION | (2) CLIEAN | (4) CLIEAT |  |  |  |
| COMEAT WHIGHT Ib. | 15.952 | 15.952 |  |  |  |
| figgine power | Combat | Military |  |  |  |
| Fuel 1b. | 1,368 | 1,368 |  |  |  |
| Combat speed/corabat altitude kno/ft. | 297/S.I. | 281/S.L. |  |  |  |
| Rate of climb/combat altitude fpm/ft. | $3.940 / \mathrm{SoL}$. | 3,260/S.I. |  |  |  |
| Combat ceiling (500 fom) ft. | - | 29,100 |  |  |  |
| Pate of climb at S.I. fpm | 3.940 | 3.260 |  |  |  |
| Max. speed at S.I. kn. | 297 | 281 |  |  |  |
| Max. speed/altitude keo/ft. | - | 301/17.200 | . |  |  |
|  |  |  |  |  |  |
| LANDING WFICHMT In. | 14,798 |  |  |  |  |
| Fual $1 \mathrm{~b}_{0}$ | 214 |  |  |  |  |
| Stall speed - powermoff kne | 72.9 |  |  |  |  |
| Stall speed - with approach power kne | 69.8 |  |  |  |  |

## NOTES

(A) Normal rated power
(B) Muel dropped with external tanks prior to combat: 1152 1b。 in loading (5) and io18 1b in loading (7)

Performance Basis: Calculations and contractors flight tests on the $A D-2$ and $A D-4 B$ airplane

- Mange and radius are based on engine specification fuel consumption data increased by $\overline{5} \%$
- Alil loadings inciude all main racks, ía Aero ILAracks.


OLOADING CONDITION COLUMN NUMBER

## NOTES

SPOTIING: A maximum operating spot aboard a CVA-19 (canted deck) class carrier consists of 42 aircraft
on the flight deck with elevators and landing area clear and 41 aireraft on the hangar deck. with hangar bay five doors and elevators clear. Total 83 aircraft.

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LOW ALTITUDE ATTACK AND GROUND SUPPORT PROBLEM
WARM-UP, TAKH-OFF, ACCEL FRATE: 10 minutes at normal rated power at sea level. CLIMB: to 5000 feet at normal rated power.
CRUISEMCUT: At speed for long range at 5000 feet.
DESCEND: To sea level (No distance gained, no fuel used).
DROP BOMBS AND FIRE ROCKETS
COMBAT: 5 minutes at maximum power plus 10 minutes at normal rated power at sea level. CLIMB: To 5000 feet with normal rated power.
CRUISE-BACK: At speed for long range at 5000 feet.
RESHRV: 20 minutes at speed for long range at sea level plus $5 \%$ of initial fuel load.
COMBAT RADIUS $=$ CLIMB + CRUISE OUT $=$ CLIMB + CRUISE BACK
MISSION TIME = TIME REQUIRED FOR CLIMB + CRUISE OUT + COMBAT + CLIMB + CRUISE BACK


## CARRIER SUITABILITY

MIIIMUM WITD OVER DECK REGUIRBD FOR CATAPULITIVG
VS. GROSS WEIGHT


MINIUUM WIIND OVER DECK REGUIRED FOR IANDING VS. GROSS WEIGHT

Based on approach speeci of 1.2 power-off stall speed


## NOTES

(A) These curves should be used for planning purposes only. Actual catapult and arresting gear operation should be in accordance with applicable Aircraft Technical Orders, and Catapult and Arresting Gear Eulletins.
(B) Based on NAMC flight test.

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