

NAVAIR 00-110AA1-3

Standard Aircraft Characteristics

NAVY MODEL EA-1F AIRCRAFT

THIS PUBLICATION SUPERSEDES NAVAIR 00-110A-1 DATED
1 MAY 1955 IN PART AND ALL ADDENDA THERETO

PUBLISHED BY DIRECTION OF THE
COMMANDER OF THE NAVAL AIR SYSTEMS COMMAND

1 JULY 1967

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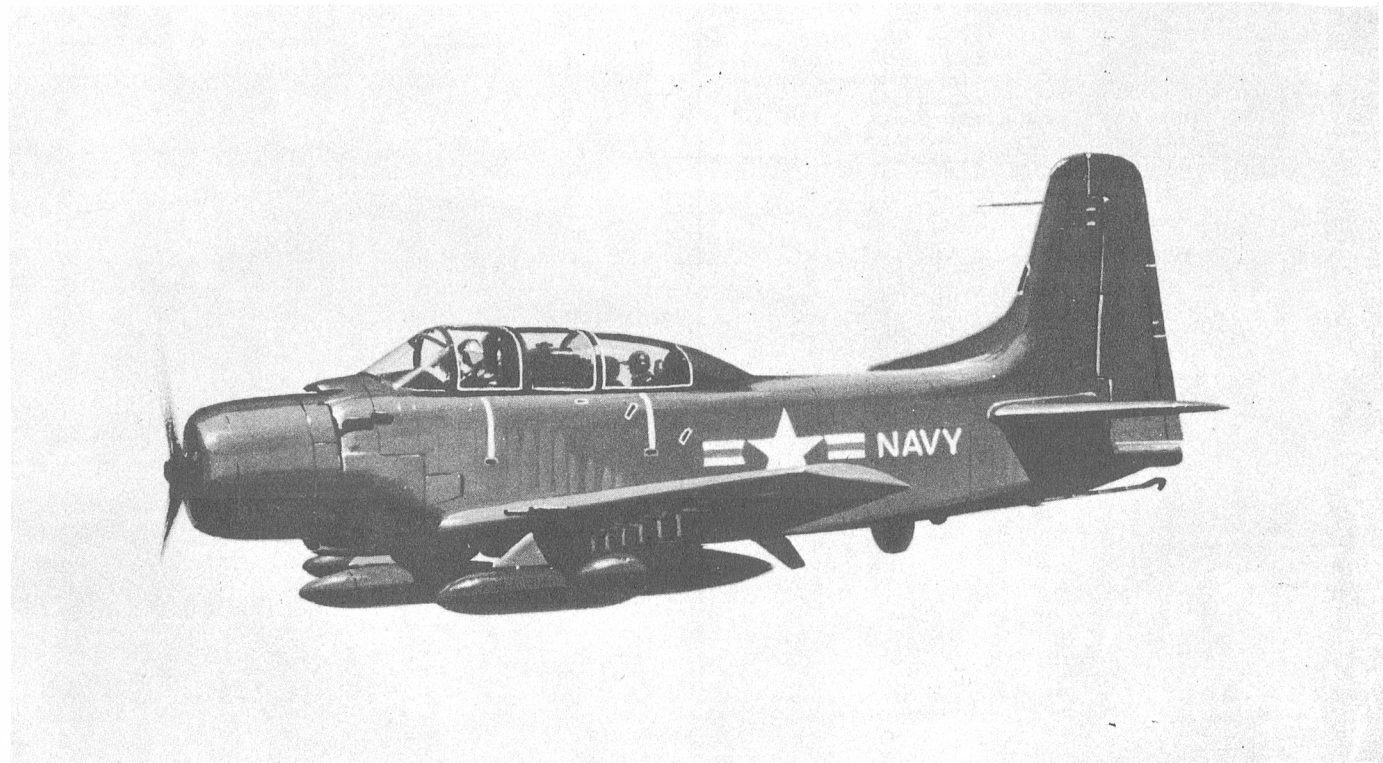
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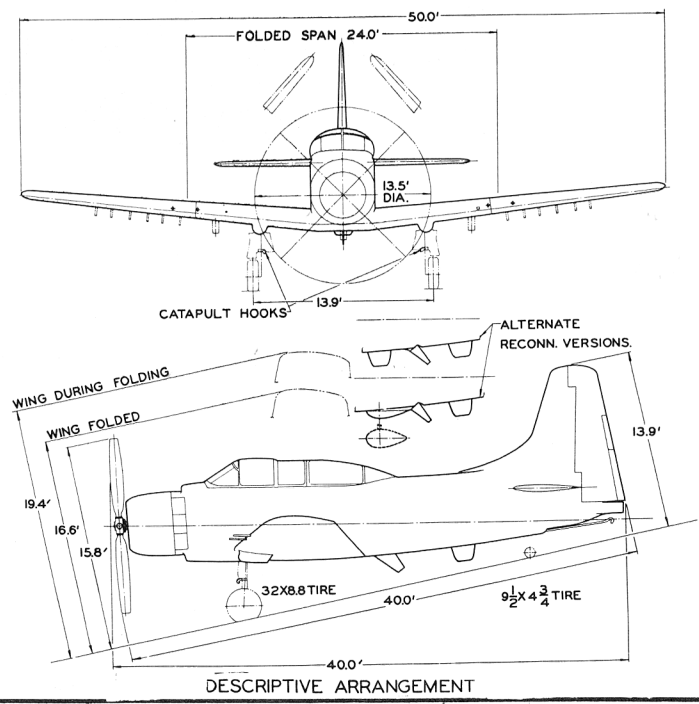
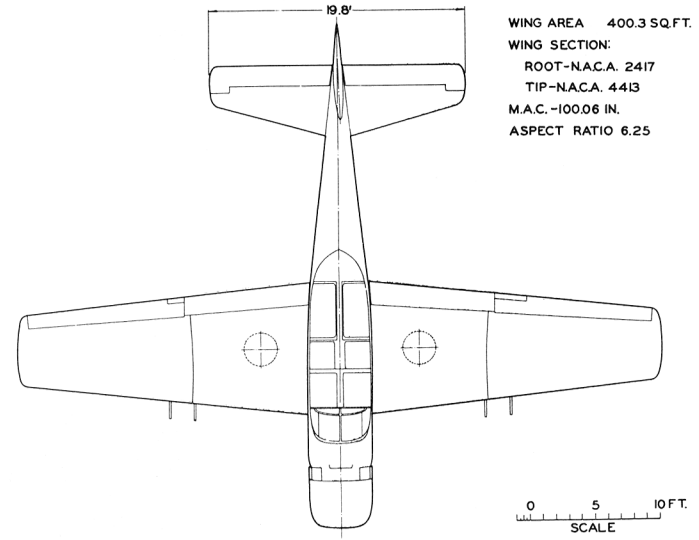
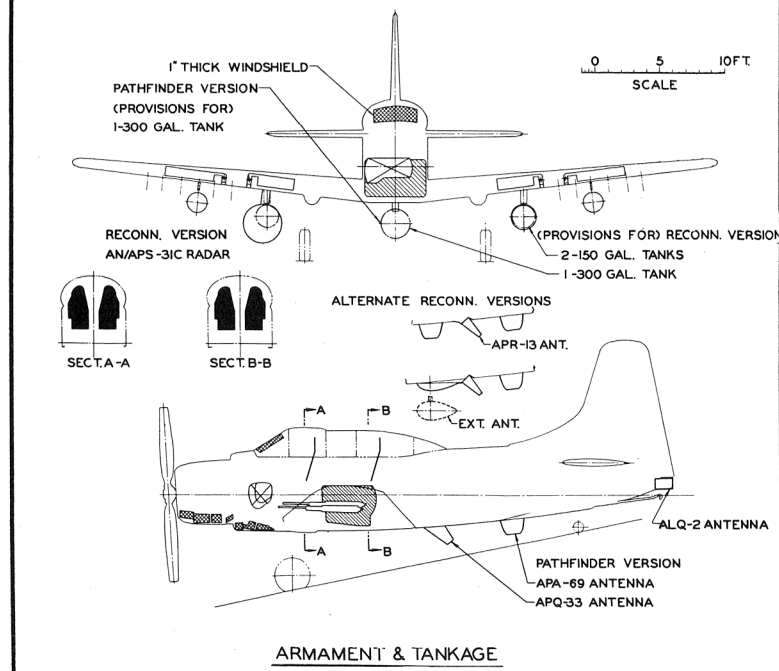
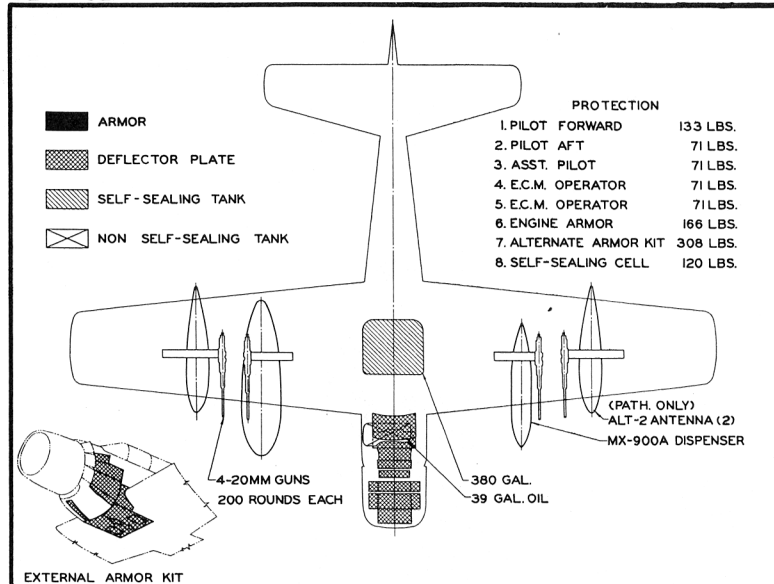
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STANDARD AIRCRAFT CHARACTERISTICS
EA-1F SKYRAIDER

SERVICE

NAVAIR 00-110AAL-3



POWER PLANT

NO. & MODEL.....(1)R-3350-26-WA
 MFR.....Wright Aero
 SUPERCHG.....Single Stage Two Speed
 REDUCT.GEAR RATIO......4375:1
 PROP. MFR.....Aero Products
 BLADE DES. NO.....M20A2-162-0
 NO. BLD./DIA.....4/13'6"

RATINGS

| | BHP | @ RPM | @ ALT. |
|-------|-------|-------|--------|
| T.O. | 2,700 | 2,900 | S.L. |
| MIL. | 2,700 | 2,900 | 3,700 |
| | 2,100 | 2,600 | 14,500 |
| NORM. | 2,300 | 2,600 | 6,200 |
| | 1,900 | 2,600 | 17,000 |

Spec. No.....N836-D

ELECTRONICS

UHF Trans.-Rec.....AN/ARC-27A
 MHF Trans.-Rec.....AN/ARC-2
 Radio Altimeter.....AN/APN-22
 Marker Beacon.....AN/ARN-12
 IFF.....AN/APX-6
 IFF Coder.....AN/APA-89
 LF ADF.....AN/ARN-6
 UHF ADF.....AN/ARA-25
 Interphone.....AN/AIC-4
 Radar Search.....AN/APS-31C
 LAB Radar Bombsight.....AN/APA-16
 LAB R-R Adapter.....MX-476/APA-16
 Sonobuoy Rec.....AN/ARR-26
 Searchlight.....AN/AVQ-2A
 ECM Rec.....AN/APR-9B
 ECM DF.....AN/APA-69A
 ECM Rec.....AN/APR-13

Provisions

VHF Trans.-Rec.....AN/ARC-1
 Bomb Director.....MK-3 MOD-5

MISSION AND DESCRIPTION

The AD-5Q is a dual purpose airplane, capable of two distinct missions; one, that of a radar reconnaissance airplane, detecting enemy radar installations and, secondly, that of a radar countermeasures airplane that jams enemy radar during an attack mission by a group of bombers.

The AD-5Q is a kit-modification to the AD-5N airplane. Crew consists of four: a pilot and navigator in the cockpit and two ECM operators in a rear compartment. The airplane is designed to operate from all classes of aircraft carriers or land bases.

The airplane is conventional in design and structure incorporating a single reciprocating engine, folding wings, conventional landing gear and catapult and arrested landing equipment. Provisions are incorporated for the carrying of fuel tanks and various stores required for the missions on the bomb racks, and for installation of 4-20mm guns in the inner wings.

DEVELOPMENT

First Flight.....October 1956
 Service Use.....July 1957

DIMENSIONS

WING
 AREA.....400.3 sq.ft.
 SPAN.....50 ft.
 MAC.....8.4 ft.
 LENGTH.....40.0 ft.
 HEIGHT.....15.8 ft.
 TREAD.....13.9 ft.
 PROP. GRD. CLEARANCE.....6 in.

WEIGHTS

| LOADINGS | LBS. | L.F. |
|----------------------|-------------|------|
| EMPTY..... | 12,097..... | |
| BASIC..... | 15,932..... | |
| DESIGN..... | 17,000..... | 6.4 |
| COMBAT..... | 19,395..... | 5.6 |
| MAX.T.O.(FIELD)..... | 25,000..... | |
| (CAT.)..... | 25,000..... | |
| MAX.LDG.(FIELD)..... | 21,000..... | |
| (ARREST)..... | 17,500..... | |

ALL WEIGHTS ARE CALCULATED

FUEL AND OIL

| GALS. | NO. TANKS | LOCATION |
|--|------------|-----------|
| 380* | 1..... | Fuselage |
| 150 or 300..... | 1..... | Ctr. Drop |
| 150 or 300..... | 2..... | Wing Drop |
| Fuel Grade..... | 115/145 | |
| Fuel Spec..... | MIL-F-5572 | |
| *Self Sealing Tank | | |
| Max. useable fuel 980 gal. (limited by oil cap.) | | |

OIL

CAPACITY.....39 gals.
 SPEC.....AN-O-8
 GRADE.....1120

ORDNANCE

Does not normally carry ordnance.

Provisions for a total of 12 Aero 14 bomb racks on outer wings and 4-20mm wing guns with 200 rounds of ammunition each.

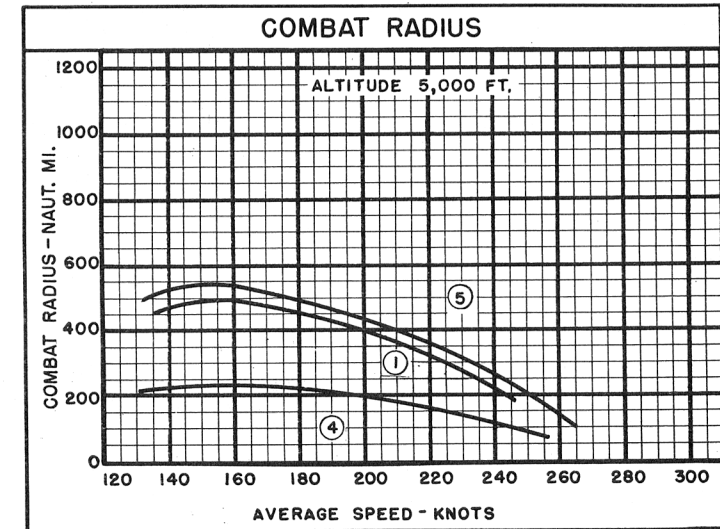
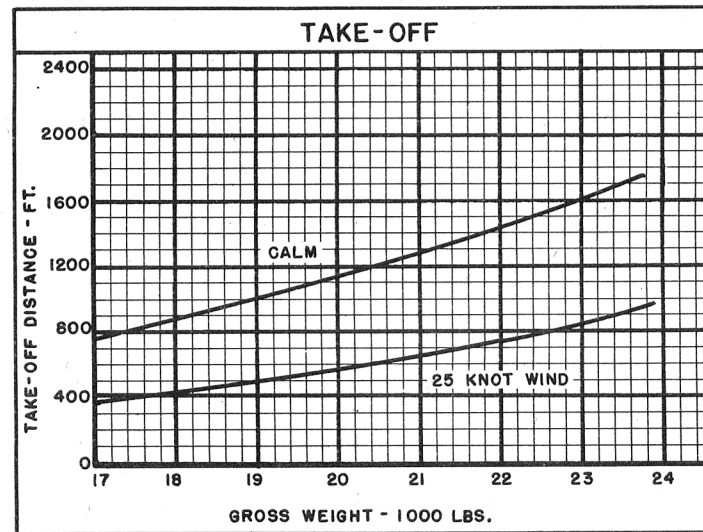
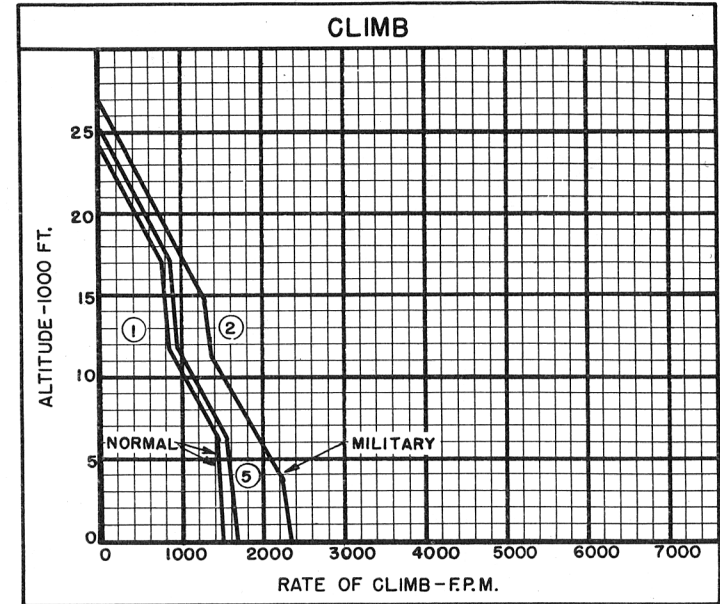
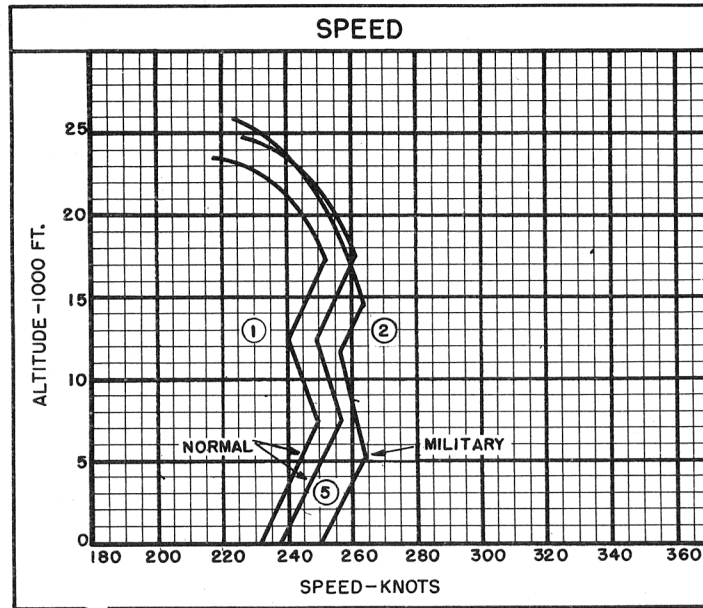
PERFORMANCE SUMMARY

| TAKE-OFF LOADING CONDITION | (1) PATHFINDER 1-300 gal. Aero 1A Fuel Tank (B) and (F) | (3) PATHFINDER 2-300 gal. Aero 1A Fuel Tanks (B) and (G) | (4) RECONNAISSANCE (C) | (5) RECONNAISSANCE 1-300 gal. Aero 1A Fuel Tank (C) | (6) RECONNAISSANCE 1-300 gal. Aero 1A 2-150 gal. Aero 1A Fuel Tanks (C) | |
|-----------------------------------|--|---|---------------------------|--|--|------------|
| TAKE-OFF WEIGHT | lb. | 21,351 | 23,142 | 18,598 | 20,590 | 22,596 |
| Fuel | lb. | 4,080 | 5,880 | 2,280 | 4,080 | 5,880 |
| Fayload | lb. | None | None | None | None | None |
| Wing loading | lb./sq.ft. | 53.4 | 57.9 | 46.5 | 51.5 | 56.5 |
| Stall speed - power-off | kn. | 91.5 | 95.2 | 85.4 | 89.8 | 94.1 |
| Take-off run at S.L. - calm | ft. | 1,320 | 1,620 | 930 | 1,200 | 1,530 |
| Take-off run at S.L. 25 kn. wind | ft. | 690 | 875 | 455 | 615 | 800 |
| Take-off to clear 50 ft. - calm | ft. | 2,240 | 2,820 | 1,590 | 2,040 | 2,620 |
| Max. speed/altitude | (A) kn./ft. | 252/17,200 | 247/17,200 | 268/17,400 | 261/17,300 | 252/17,200 |
| Rate of climb at S.L. | (A) fpm. | 1,510 | 1,270 | 2,010 | 1,660 | 1,370 |
| Time: S.L. to 10,000 ft. | (A) min. | 7.1 | 8.7 | 5.3 | 6.4 | 8.1 |
| Time: S.L. to 20,000 ft. | (A) min. | 18.3 | 24.0 | 12.8 | 16.2 | 21.4 |
| Service ceiling (100 fpm) | (A) ft. | 23,100 | 21,700 | 27,500 | 24,500 | 22,300 |
| Combat range | n.mi. | 1,180 | 1,685 | 680 | 1,285 | 1,800 |
| Average cruising speed | kn. | 165 | 165 | 165 | 165 | 165 |
| Cruising altitude(s) | ft. | 5,000 | 5,000 | 5,000 | 5,000 | 5,000 |
| Combat radius | n.mi. | 485 | 480 | 225 | 530 (D) | 795 (E) |
| Average cruising speed | kn. | 165 | 165 | 165 | 165 | 165 |
| Mission time | hrs. | 6.2 | 6.0 | 3.0 | 6.7 | 9.8 |
| COMBAT LOADING CONDITION | (2) CLEAN Full Internal Fuel | | | | | |
| COMBAT WEIGHT | lb. | 19,359 | | | | |
| Engine power | | Military | | | | |
| Fuel | lb. | 2,280 | | | | |
| Combat speed/altitude | kn./ft. | 251/Sea Level | | | | |
| Rate of climb/altitude | fpm/ft. | 2,320/Sea Level | | | | |
| Combat ceiling (500 fpm) | ft. | 22,000 | | | | |
| Rate of climb at S.L. | fpm. | 2,320 | | | | |
| Max. speed at S.L. | kn. | 251 | | | | |
| Max. speed/altitude | kn./ft. | 264/5,100 | | | | |
| LANDING WEIGHT | lb. | 17,416 | | | | |
| Fuel | lb. | 337 | | | | |
| Stall speed - power-off | kn. | 82.6 | | | | |
| Stall speed - with approach power | kn. | 77.6 | | | | |

NOTES

- (A) Normal rated power.
- (B) Pathfinder configuration includes following external antennae: AS-776/APA-69 radome, AT-321/APQ-33 blade, AS-694/ALQ-2, ARN-6, APX-6, and ARN-21.
- (C) Reconnaissance configuration includes following external antennae: AS-776/APA-69 radome, AS-435/APA-69 radome, AS-694/ALQ-2, APR-13 blade, ARN-C, APX-6 and ARN-21.
- (D) The combat radius with the extendable APA-69 is 513 n.mi. if the radome is extended for two hours and the 300-gallon fuel tank is retained.

Continued on NOTES Page



○ LOADING CONDITION COLUMN NUMBER

NOTES

Continued from PERFORMANCE SUMMARY Page

(E) The combat radius shown is based on retaining the external fuel tanks during the combat period. If the tanks and 1539 lbs. of fuel are dropped prior to the combat period, the combat radius is 526 n.mi. and the mission time is 6.6 hours.

(F) Includes two AN/ALT-2 Stores, MX-900A and AN/APS-31P.

(G) Includes two AN/ALT-2 Stores, AN/APS-31P.

PERFORMANCE BASIS: Performance is calculated and is based on estimates and contractor's flight tests of models AD-4B, AD-5 and AD-6.

COMBAT RADIUS AND RANGE is based on fuel consumption data from AD-4B, AD-5 and AD-6 flight tests and is increased 5%.

All loadings include centerline and inner wing bomb racks, 12 Aero 14 racks and no guns.

SPOTTING: A total of 83 airplanes can be accommodated in a landing spot on the flight and hangar decks of a CVA-19 class angled deck carrier.

LOW ALTITUDE ATTACK AND GROUND SUPPORT BOMBER MISSION-COMBAT RADIUS PROBLEM

WARM-UP, TAXI, TAKE-OFF: 10 minutes at normal power.

CLIMB: On course to 5,000 feet with normal power.

CRUISE-OUT: At 5,000 feet at velocity for long range. (Drop external fuel tank when empty)

DESCEND: To sea level. (No fuel used - no distance gained)

DROP BOMBS, FIRE ROCKETS.

COMBAT: 15 minutes at sea level (5 minutes at military power and 10 minutes at normal power).

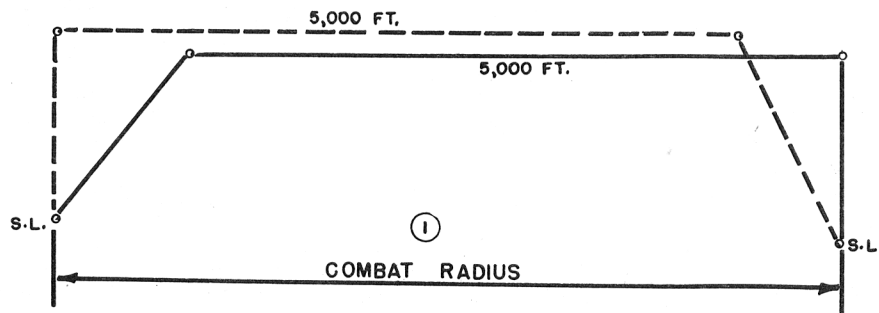
CLIMB: On course to 5,000 feet with normal power.

CRUISE-BACK: At 5,000 feet at velocity for long range.

RESERVE: 20 minutes at velocity for long range at sea level plus 5% of initial fuel load.

$$\text{COMBAT RADIUS} = \text{CLIMB} \neq \text{CRUISE-OUT} = \text{CLIMB} \neq \text{CRUISE-BACK}$$

$$\text{MISSION TIME} = \text{TIME REQUIRED FOR CLIMB} \neq \text{CRUISE-OUT} \neq \text{COMBAT} \neq \text{CLIMB} \neq \text{CRUISE-BACK}$$



○ LOADING CONDITION COLUMN NUMBER