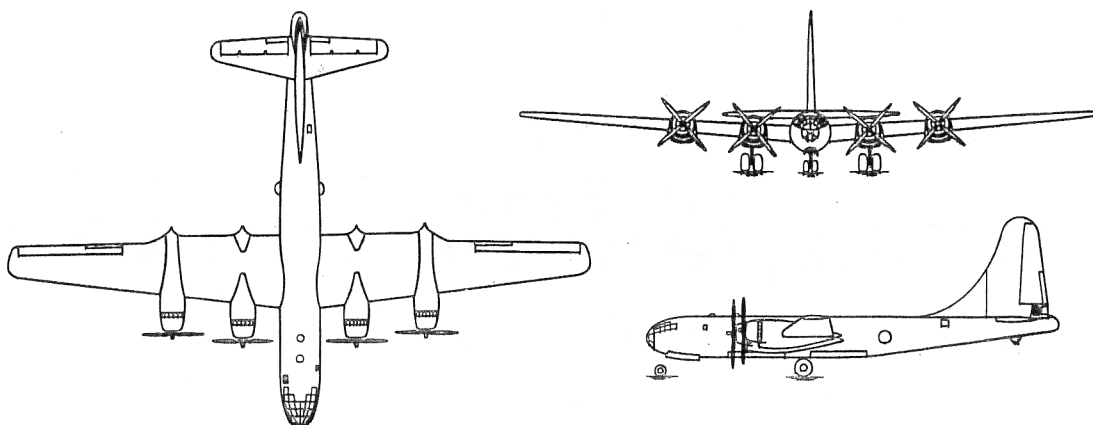


Characteristics Summary

TRAINER TB-50D



"SUPERFORTRESS"

BOEING

Wing Area 1720 sq ft Length 99.0 ft
 Span 141.2 ft Height 32.7 ft

AVAILABILITY			PROCUREMENT			
Number available			Number to be delivered in fiscal years			
ACTIVE	RESERVE	TOTAL				

STATUS

1. The TB-50D was developed from the B-50D airplane.
2. Contract Approval: Apr 50
3. First Flight: Feb 51
4. First Acceptance: Jan 51
5. Production Completed: Jun 51

Navy Equivalent: None

Mfr's Model: 345-9-24

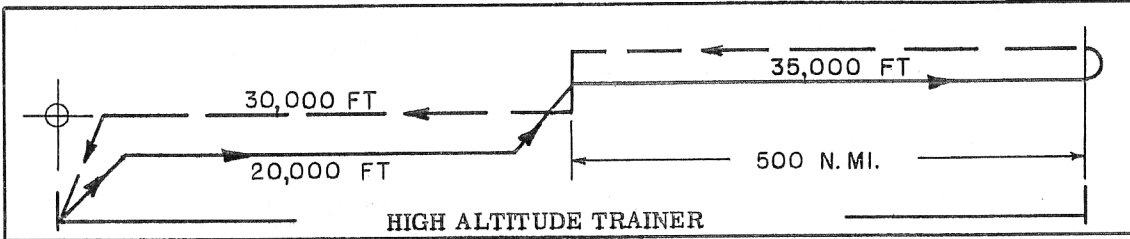
POWER PLANT
(4) R-4360-35 Pratt & Whitney
ENGINE RATINGS
BHP -RPM-ALT-MIN
T.O: *3500-2700- SL -5 3250-2700- SL -5
Mil: *3500-2700-Turbo-30 3250-2700-Turbo-30
Nor: 2650-2550-Turbo-Cont
* Wet

FEATURES
Crew: 13 includes
Navigator Trainees plus Instructor 3
Radar & K-3 Trainees plus Instructor 5
K-3 Bombing-Navigational System
Cabin Pressurization
Thermal Anti-icing
Folding Fin and Rudder
Reversible Pitch Props
Max Fuel Capacity: *8055gal
*Includes Nacelle Skate Tanks

ARMAMENT
Turrets: None
Guns: None
BOMBS
Maximum
Size (lb) Load (lb)
War II
Box Fin .1000.... 20x500

There is a duplicate for this date.

Characteristics Summary Basic Mission TB-50D



PERFORMANCE		
COMBAT RADIUS	FERRY RANGE	S P E E D
1749 naut. mi with 10,000 lb payload at 273 knots avg. in 13.1 hours.	4373 naut. mi with 8055 gal fuel at 215 knots avg. in 20.4 hours at 135,646 lb T.O. wt.	COMBAT 357 knots at 35,000 ft alt, max power MAX 363 knots at 31,000 ft alt, max power BASIC 353 knots at 25,000 ft alt, max power
C L I M B	C E I L I N G	T A K E - O F F
1040 fpm sea level, take-off weight normal power	35,200 ft 100 fpm, take-off weight normal power	ground run 3940 ft no assist — ft assisted
2300 fpm sea level, combat weight maximum power	38,500 ft 500 fpm, combat weight maximum power	over 50 ft height 4850 ft no assist — ft assisted
L O A D	W E I G H T S	S T A L L I N G S P E E D
Bombs: 10,000 lb Fuel: 8055 gal protected 100 % droppable 0 % external 0 %	Empty..... 79,104 lb Combat... 106,777 lb Take - off 145,646 lb limited by space	99.0 knots power-off, landing config- uration, take-off weight
		T I M E T O C L I M B
		—————

N O T E S

1. Performance Basis:
 - (a) Data Source: Calculations based on flight test of B-50D aircraft with configuration adjustments.
2. Revision Basis: Data coordinated this date.