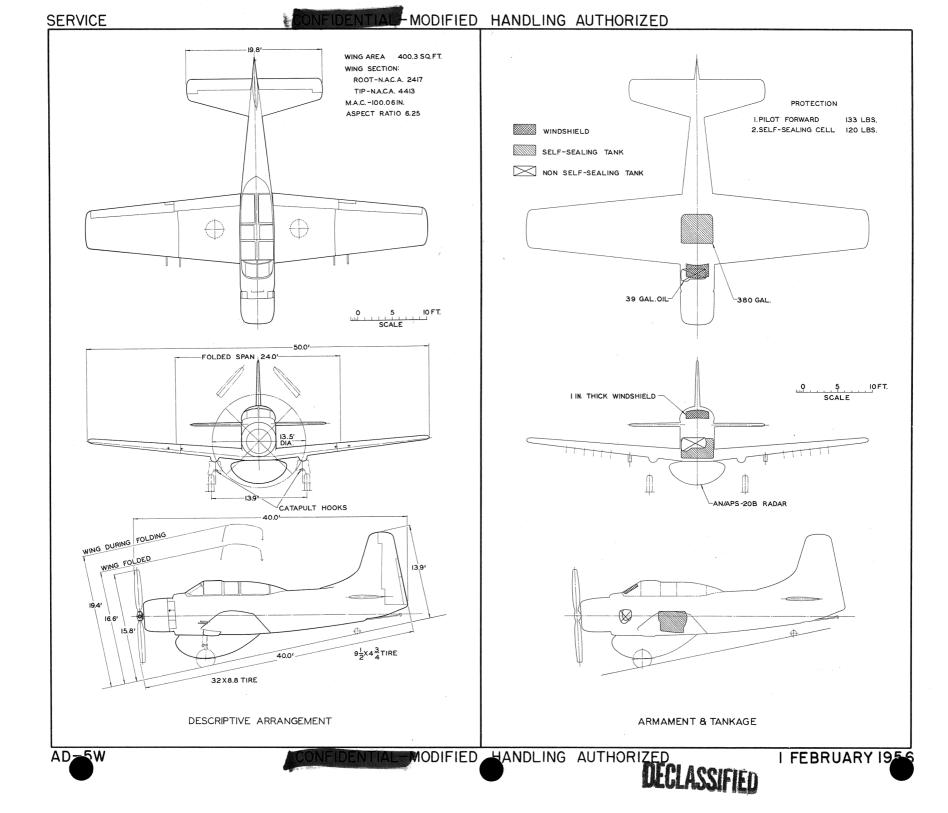


STANDARD AIRCRAFT CHARACTERISTICS AD-5W SKYRAIDER

DOUGLAS AIRCRAFT COMPANY, INC., EL SEGUNDO DIVISION

DECLASSIFIED

COPY NO. 20



POWER PLANT

No. & Model (1)R-3350 - 26 - WA
Mfr	Wright Aero
Engine Spec. No	N-836-D
Supchg Single	Stage Two Speed
Reduction Gear Rat	io4375:1
Prop. Mfr	Aero Products
Prop. Des	A-642-G-805
Prop. Blade Des	M20A2-162-0
No.Blades/Prop. Dia	a4/13 ' 6"

RATINGS

BHP	@RPM	@Alt.
2700	2900	S.L.
2700	2900	3700
2100	2600	14500
2300	2600	6200
1900	2600	17000
	2700 2700 2100 2300	2700 2900 2700 2900 2100 2600 2300 2600

ORDNANCE

Although no ordnance is carried on search missions, the AD-5W airplane is equipped with innerwing bomb racks identical to those on other AD-5 models and can carry the same stores. There is no centerline bomb station on the AD-5W. All provisions including controls are available in the outer wing for the installation of six Aero 14 racks per wing, and in the wing fold for four 20mm guns with 200 rounds of ammunition per gun.

MISSION AND DESCRIPTION

The principal mission of the AD-5W is that of airborne early warning and anti-submarine search from carrier or land bases. The AD-5W is a development of the proven AD-4 Series and incorporates improvements in equipment, arrangement, and performance and stability. Installation provisions for all armor and armament carried on other AD-5 models are retained.

The side-by-side seating arrangement of pilot and assistant pilot-navigator operator facilitates all-weather operation. The unified cockpit arrangement with the radar operator aft of the pilot provides interchange of crew positions and maintenance of electronic equipment in flight. AN/APS-20B radar is installed for search operations. Appropriate radar and communications relay equipment also are installed.

DEVELOPMENT

Contract approval dates:

NOa(s) 52-247, 19 December 1951, 64 airplanes

NOa(s) 52-979, 24 April 1952, 84 airplanes

NOa(s) 54-324, L.I., 16 October 1953, 50 airplanes.

Production status:

In production. Delivery of last airplane schedule March 1956.

DIMENSIONS

Span .		50.0	ft.
Length	1	40.0	ft.
Height	·	15.8	ft.
Tread		13.9	ft.
Prop.	Gnd.	Clr6.25	in.
Turn.	Radi	us32	ft.

WEIGHTS Loadings Lbs. L.F. Empty (A) 12,092 Basic 14,802 Flt. Design 17,000 6.4 Combat 17,125 6.4 Max. T.O. 25,000 Arr. Landing 17,500

FUEL AND OIL

<u>Gal</u> .	No.Tanks	Location	
380*	1	Fuselage	
150 or 30	00 2	WingDrop	
Fuel		115/145	
Fuel	Spec	MIL-F-5572	
*Self Sealing Tank			
Max. Us	sable Fuel 980	Gals.	

OIL

(limited by oil capacity)

Capacity	39 Gals.
Spec	AN-0-8
Grade	1120

ELECTRONICS

UHFTrans-Rec.&Relay	AN/ARC-27A
Radio Altimeter	AN/APN - 22
Marker Beacon	AN/ARN-12
IFF	AN/APX-6
IFF Coder	AN/APA-89
IFF	AN/APX-7
LF ADF	AN/ARN-6
UHF ADF	AN/ARA <i>-</i> 25
Interphone	AN/AIC-4
Radar Relay	AN/ART - 28
Radar Search	AN/APS-20
Radar GPI	AN/APA-57
Provisions	
VHF Relay	AN/ARC - 28
MHF Trans-Rec	AN/ARC - 2

	PERFORMANCE SUMMARY				
TAKE-OFF LOADING CONDITION	(1) SEARCH 1500 ft. cruise altitude	(1) SEARCH 7000 ft. cruise altitude	(5) SEARCH 2-150 gal. fuel tanks (AerolA)	(6) SEARCH 2-300 gal. fuel tanks (AerolA)	
TAKE-OFF WEIGHT 1b			20,041	22,047	
Fuel lb	2200		4080	5880	
Payload (AN/APS-20) lb					
Wing loading lb./sq.ft			50.1	55.1	
Stall speed - power-off km	0100		92.7	97.2	
Take-off run at S.L calm ft			1360	1760	
Take-off run at S.L 25 kn. wind ft			735	980	
Take-off to clear 50 ft calm ft			2385	3150	
Max. speed/altitude (A) kn./ft			257/17,300	250/17,200	
Rate of climb at S.L. (A) fp			1640	1330	
Time: S.L. to 10,000 ft. (A) min			6.5	8.3	
Time: S.L. to 20,000 ft. (A) min			17.1	24.2	
Service ceiling (100 fpm) (A) ft			23,800	21,100	
Combat range n.mi		658	1294	1798	
Average cruising speed kn	100	170	160	160	
Cruising altitude(s) ft		7000	1500	1500	
Combat radius (C) n.mi	274	263	518	719	-
Average cruising speed km	. 160	170	160	160	
Total Mission Time hr	4.4	3.9	8.1	11.1	
COMBAT LOADING CONDITION	(2) 60% FUEL	(3) 60% FUEL			
COMBAT WEIGHT 1b	17,125	17,125			
Engine power	MILITARY	NORMAL			
Fuel lb	. 1368	1368			
Combat speed/combat altitude kn./ft	253/1500				
Rate of climb/combat altitude fpm/ft	. 2750/1500				
Combat ceiling (500 fpm) ft					The state of the s
Rate of climb at S.L. fp		2210			
Max. speed at S.L. kn	. 250	237			
Max. speed/altitude kn./ft	. 266/14,900	264/17,300			
LANDING WEIGHT 1b					
Fuel lb	232			4	
Stall speed - power-off kn					
Stall speed - with approach power kn	77.1				

NOTES

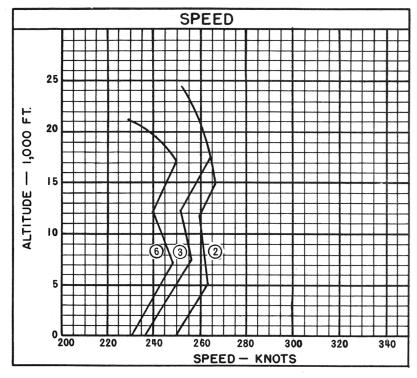
- (A) Normal Rated Power.
- B) Performance Basis: Performance is calculated and based on flight tests of the AD-5W.

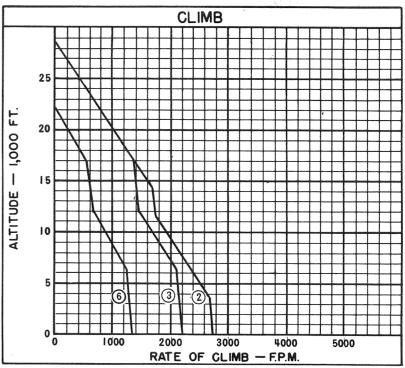
Combat range and radius are based on fuel consumption data from AD-4B, AD-5, and AD-6 flight tests and are increased 5%.

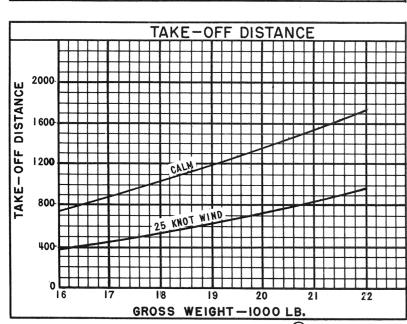
- (C) Combat Radius is 40% of Combat Range.
- (D) All loadings include inner wing racks.
- (E) This revision prepared to incorporate results of AD-5W flight tests.
- (F) Spotting: 20 aircraft can be spotted in an area 96 ft. wide and 200 ft. long.

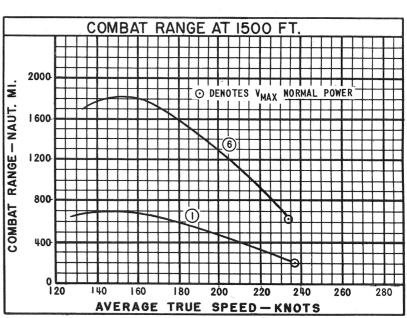
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NOTES

COMBAT RANGE

Warm-Up, Taxi, Take-Off

Fuel for 10 min. at S.L. normal power.

Climb

At maximum rate of climb with normal power to 1500 ft.

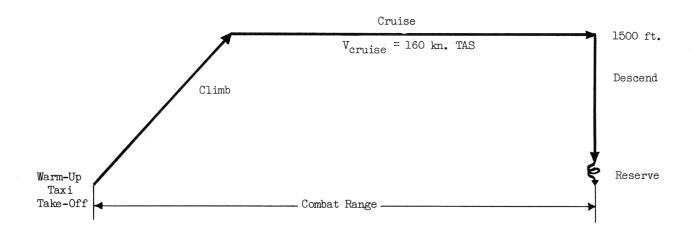
Cruise

At 1500 ft. at speed for maximum range operation, tanks dropped when empty.

Reserve

Fuel allowance: 5 percent of initial fuel plus 20 min. at speed for long range at sea level.

Combat Range = Climb + Cruise

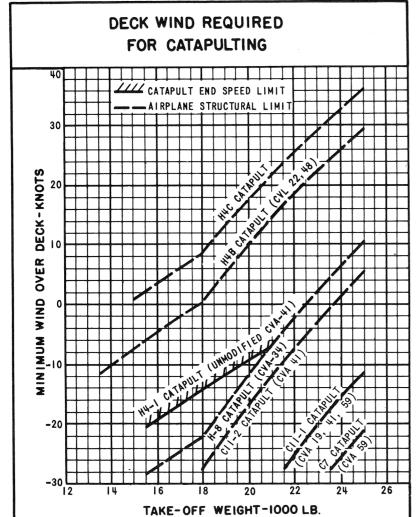


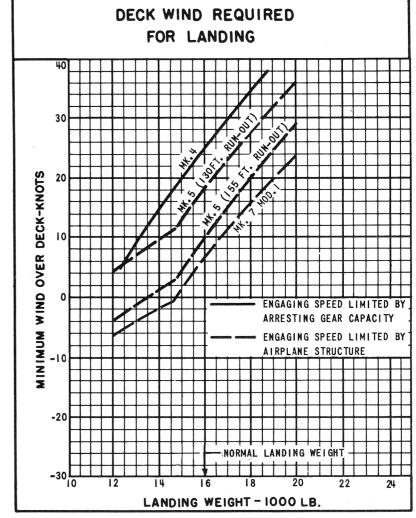
COMBAT RADIUS

Combat radius is 40 percent of combat range at 1500 ft. altitude.

DEGLASSIFIED

CARRIER SUITABILITY





NOTE:

The arrested landing approach speed equals 1.15 times the power-off stalling speed.

