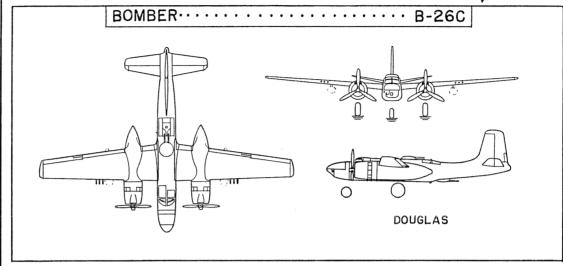
Characteristics Summary



Wing area 541.0 sq ft

Length 51.3 ft

Span 70.0 ft

Height 18.5 ft

AVAILABILITY Number available			PI	ROC	UR	E M	E	N	T
			Number to be delivered in fiscal years						
ACTIVE	RESERVE	TOTAL							

STATUS

- 1. Differs from B-26B "LEFT" project by the addition of lower turret and change from allpurpose attack nose to a bombardier nose.
- 2. First acceptance: May 1944
- 3. Production completed: July 1945
- 4. Navy designation: JD-1

POWER PLANT

(2) R-2800-79 Pratt-Whitney ENGINE RATINGS

BHP - RPM - ALT T.O: 2000 - 2700 -

2000 - 2700 - 1500 Mil:

1600 - 2700 - 13,500 1600 - 2400 - 5700 Nor:

1450 - 2400 - 13,000

FEATURES

Crew:

Cabin Heating and Cooling Periscopic Sighting for Tur-

ret

Propeller Anti-icing

Bombing-Navigation Radar

Loran

Max Fuel Cap:

1910 gal

ARMAMENT

Turrets:

Guns:

12x.50 cal. 4600 rds

Ammunition: Max Bomb Load:

4×1000 lb (int.)

4x500 lb (ext.)

Torpedoes (alternate)

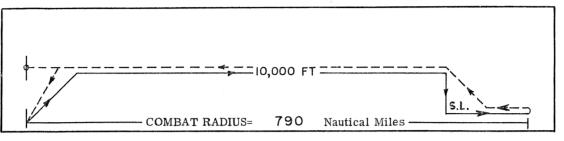
2×2000 lb

ROCKETS

(14) 5" HVAR

Ry Ly July 1

Characteristics Summary Basic Mission ··· ·· · · B-260



PERFORMANCE									
COMBAT RADIUS	COMBAT RANGE	COMBAT SPEED							
790 naut. mi	1510 naut. mi	315 knots at S.L. ft alt, max power							
with 4000 lb payload	with 4000 lb payload	MAXIMUM SPEED							
at 198 knots avg.	at 200 knots avg.	323 knots at							
in 8.15 hours.	in 7.67 hours.	10,000 ft alt, max power							
CLIMB	CEILING	TAKE-OFF							
sea level, take-off weight normal power	20450 ft 100 fpm, take-off weight normal power	ground run 3390ft — ft no assist assisted							
2800 fpm sea level, combat weight maximum power	23400 ft 500 fpm, combat weight maximum power	over 50 ft height 4180 ft ft assisted							
L O A D	WEIGHTS	STALLING SPEED							
Bombs: 4000 lb Ammunition: 4600 rds/.50 cal	Empty 22,690 lb Combat 29,500 lb	95 knots flaps down, take-off weight							
Fuel: 1235 gal protected 75 % droppable 25 % external 25 %	Take - off 37,740 lb	TIME TO CLIMB							

N 0 E S

1. PERFORMANCE BASIS:

- (a) Flight test(b) Fuel density: 6.0 lb/gal
- (c) In computing Radius and Range, specific fuel consumptions have been increased 5% to allow for variation of fuel flow in service aircraft.

 (d) Performance shown above for max power is based on war emergency power (wet) of
- 2370 BHP @ 2700 RPM @ S.L.
- 2. REVISION BASIS: To show engine ratings and changes in "Status", "Features" and "Notes" block.