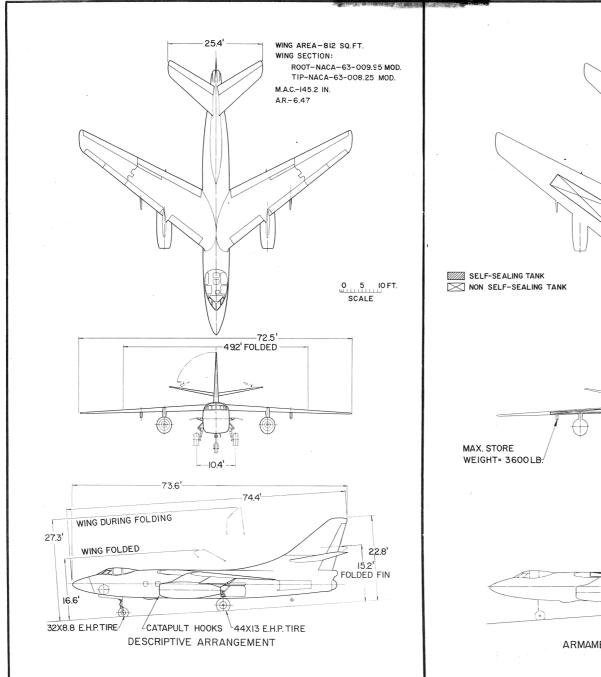
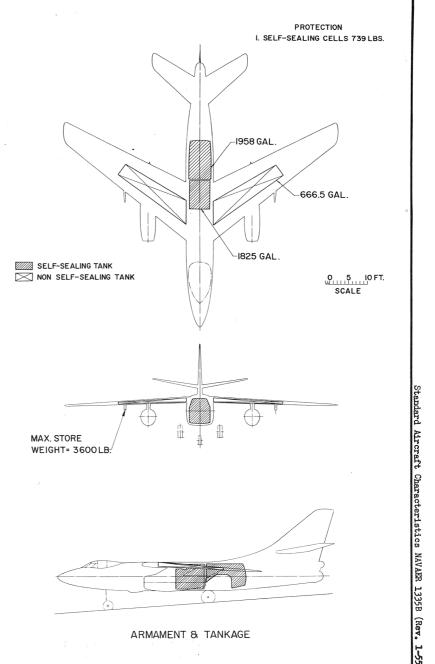


STANDARD AIRCRAFT CHARACTERISTICS A3D-2T "SKY WARRIOR"

DOUGLAS

DECLASSIFIED





LF.

3.4

2.44

2.60

2.72

ASC 155)

ASC 155)

46000 (without

POWER PLANT

No. & Model-----(2) J57-P-10 MFR.----Pratt & Whitney Type ----- Turbojet Compr----- Dual Rotor Axial Flow Lenght-----158 in. Diameter____41 in. No. & Type Assist_12-5KS4500Jato Tail Pipe No. 22LE---Constant Exit Area

RATINGS

SEA LEVEL STATIC THRUST RPM N₂** LB. Maximum 10500 6450 9900 9900 Military 10500 6150 Normal 9000 5900 9650

*N1 Speed of Low Pressure Compressor **N2 Speed of High Pressure Compressor Eng. Spec. No. N-1700-A

ORDNANCE

WING PYLONS

- 3343 Lb. T-65 Training Shapes
- 750 Lb. Aero 8A Practice Bomb Dispensers Carrying 4-MK 89 Bombs Each

No Provision for Internal Carriage of Stores

MISSION AND DESCRIPTION

The primary mission of the A3D-2T airplane is that of a trainer for bombardiers and navigators. The airplane is designed to operate from land bases and from CVA-19 and CVA-59 class carriers.

The airplane has a conventional swept wing and swept tail arrangement. Two turbo-jet engines are pylon-mounted in underwing nacelles. Provisions are made for a crew of eight men, consisting of a pilot, an instructor and six bomb-

The basic structure is nearly identical to the A3D-2Q. The wing has a cambered leading edge and full-span slats. One bomb rack is located under each wing to carry practice bomb dispensers. The rear compartment floor is adaptable

DEVELOPMENT

First Flight ----- May 1959 Service Use (Estimated)-----October 1959

ardier-navigator trainees.

to carrying cargo or transport type seats.

GAL. NO. TANKS LOCATION *1825 Fwd. Fuselage *1958 Aft Fuselare 666.5 ea. Wing * Self Sealing Total Fuel Capacity 5116 Gals. Fuel Grade ----- IP-5 Fuel Spec -----MIL-F-5624 OIL GAL. NO. TANKS LOCATION 11 2 Integral with Engine

WEIGHTS

LBS.

39933

40579

55942

59690

78000

73000

70000

LOADING

Empty

Basic

Design

Combat

Max T.O.(Land)

Max T.O. (Cat.)

Catapult Design

Max. Landing (Land) 56000

Max. Landing (Carrier) 49000 (with

FUEL AND OIL

DIMENSIONS

WING AREA
LENGTH74.4 ft. HEIGHT22.8 ft. TREAD10.4 ft.

*All aerodynamic coefficients are based on the original wing area of 779 square feet and MAC of 140.14 inches

ELECTRONICS

Oil Spec.-----MIL-L-7808

Radio Rec. (VOR) ---- AN/ARN-14-E Radio Compass ------AN/ARN-6 Tacan-----AN/ARN-21 Direction Finder, (UHF) -- AN/ARA-25 Radio Altimeter -----AN/APN-22 IFF Transponder -----AN/APX-6B Coder Group -----AN/APA-89

(without video coder)

Trans-Receiver, (UHF)-AN/ARC-27A Communication Radio --- AN/ARC-38 Radio Receiver (UHF)----- AN/ARR-40 Radar Bomb Director-----ASB-1A Antenna Coupler------CU-509()/AR







PERFORMANCE SUMMARY						
TAKE-OFF LOADING CONDITION	(1) HIGH ALTITUDE ATTACK 2 AERO 8A DISPENSERS	(3) HIGH ALTITUDE ATTACK 2 AERO 8A DISPENSER	(5) HIGH ALTITUDE ATTACK WITH TWO T-65 TRAINING STORES			
TAKE-OFF WEIGHT 1b	. 70,000	78,000	78,000			
Fuel (JP-5)	25,781	33,788	28,595			
Fayload 1b		1500	6686			
Wing loading lb./sq.ft	86.2	96.1	96.1			
Stall speeu - power-off kn	. 119.7	126.4	126.4			
Take-off run at S.L calm (A) ft	4600	5900	5900			
Take-off To Clear 50 ft Calm (A) ft	4920	6300	6300			
Take-off to clear 50 ft JATO (B) ft	2000	2850	2850			
Max. speed/altitude (C) kn./ft	. 531/S.L.	531/S.L.	533/S.L.			
Rate of climb at S.L. (C) fpm		4200	4230			
Time: S.L. to 20,000 ft. (C) min	5.4	6.2	6.2			
Time: S.L. to 30,000 ft. (C) min	9.2	10.7	10.7			
Service ceiling (100 fpm) (C) ft		39,400	39,400			
Combat range n.mi		2815	2295			
Average cruising speed kn	430	430	430			
Cruising altitude(s) ft	• 37600/4500	35600/44900	35600/42600			
Combat radius n.mi	1100	1405 (D)	1190			
Average cruising speed km	. 430	430	430			
Mission Time	5.2	6.6	5.1			
Low Altitude Attack (Radius/Mission Time)n.mi./h	2		1270/6.0			
COMBAT LOADING CONDITION	(2) 60% FUEL WITH STORES	(4) 60% FUEL WITH STORES	(6) 60% FUEL WITH STORES			
COMBAT WEIGHT 1b	59,688	64,485	66,562			
Engine power	MILITARY	MILITARY	MILITARY			
Fuel		20,273	17.157		,	
Combat speed/combat altitude kn./ft		467/41,000	467/40,500			
Rate of climb/combat altitude fpm/ft		700/41,000	700/40,500			
Combat ceiling (500 fpm) ft		41,700	44,900			
Rate of climb at S.L. fpm		5200	5080			
Max. speed at S.L. kn		531	533			
Max. speed/altitude kn./ft	505/35,000	502/35,000	506/35,000			
LANDING WEIGHT	. 46,288	46,703	45,366			
Fuel	40,000	2943	2647	1		
Stall speed - power-off km	~/~=	97.6	96.1			
Stall speed - with approach power kn		96.0	94•5			

NOTES

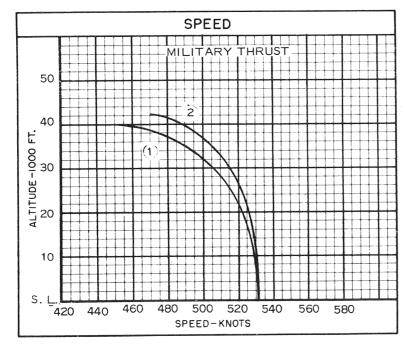
PERFORMANCE BASIS: Flight Test of A3D-1, wind tunnel tests of A3D-2 (cambered wing) and calculations

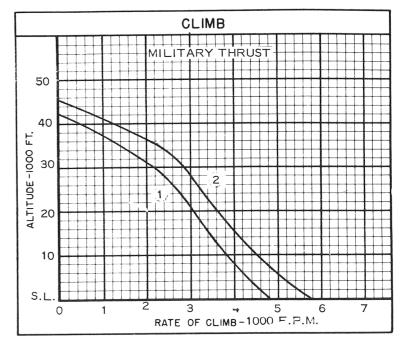
RANGE AND RADIUS are based on engine spec. fuel consumption increased by 5%

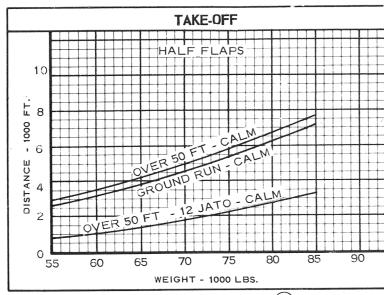
SFOTTING: A total of 27 airplanes can be accommodated in a landing spot on the flight and hangar decks of a CVA-19 class angled deck carrier

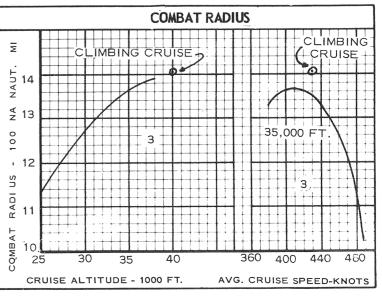
- (A) HALF FLAPS
- (B) HALF FLAPS 12 JATO UNITS INCREASE TAKE-OFF WEIGHTS SHOWN BY 2712 LBS.
- (C) MILITARY THRUST
- (D) WITH ONE REFUELING ON THE OUTBOUND LEG, THE HIGH ALTITUDE ATTACK RADIUS IS INCREASED TO 2036 N.MI. (WITH A MISSION TIME OF 9.54 HOURS)

NAVAER-1335D (Rev.









LOADING CONDITION COLUMN NUMBER

A3D-2T

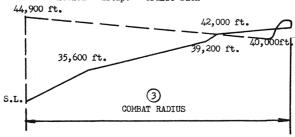


NOTES

HIGH ALTITUDE ATTACK

WARM-UP, TAKE-OFF AND ACCELERATE: 5 minutes at normal thrust at sea level CLIMB: On course to optimum cruise altitude with military thrust CRUISE-OUT: At altitudes and speeds for maximum range CLIMB: With maximum thrust on course to cruise ceiling BOMB RUN: 15 minutes at normal thrust at combat altitude DROP BOMBS: EVASIVE ACTION: 2 minutes with normal thrust at combat altitude (no distance gained) ESCAPE: 8 minutes with normal thrust (assume escape ends at optimum cruise back altitude) CRUISE-BACK: At altitudes and speeds for maximum range RESERVE: 20 minutes at speed for maximum endurance at sea level plus 5% of initial fuel load

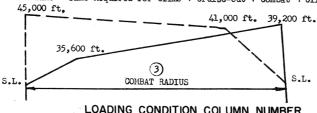
Combat Radius = Climb + Cruise-Out + Climb + Run-In = Escape + Cruise-Back Mission Time = Time Required for Climb + Cruise-Out + Climb + Bomb Rum + Eyasive Action + Escape + Cruise-Back



LOW ALTITUDE ATTACK AND GROUND SUPPORT

WARM-UP, TAKE-OFF AND ACCELERATE: 5 minutes at normal thrust at sea level CLIMB: On course to optimum cruise altitude with military thrust CRUISE-OUT: At altitudes and speeds for maximum range DECEND TO SEA LEVEL: No fuel consumed, no distance gained DROP BOMBS: COMBAT: 5 minutes at military thrust at sea level (no distance gained) CLIMB: On course to optimum cruise altitude with military thrust CRUISE BACK: At altitudes and speeds for maximum range RESERVE: 20 minutes at speed for maximum endurance at sea level plus 5% of initial fuel load

> Combat Radius = Climb + Cruise-Out = Climb + Cruise-Back Mission Time = Time Required for Climb + Cruise-Out + Combat + Climb + Cruise-Back



LOADING CONDITION COLUMN NUMBER