

STANDARD AIRCRAFT CHARACTERISTICS

A4D-2 "SKYHAWK"

DOUGLAS

standard Aircraft Characteristics NAVAER 1555A (Hi

DECLASSIFIED SERVICE PROTECTION: I.PILOT WING AREA 260.0 SQ. FT FLAK PROTECTION & BULLET-INTEGRAL 29 LBS. WING SECTION: RESISTANT GLASS PROVISIONS 125 LBS. ROOT NACA 0008-I.I-25:0875(5X230) SELF-SEALING TANK 2. SELF-SEALING CELL 107 LBS. TIP NACA 0005-.825-50-.0787(5X230) NON-SELF-SEALING TANK M.A.C. 129.64 IN. ASPECT RATIO 2.91 568 GAL. IN-FLIGHT FUELING PROBE \ 232' GAL. (ALTERNATE) O 5 IOFT. O '5 IOFT. -DROP TANK OR STORES UP TO 2150 LB. ON EACH RACK STORES UP TO 3575 LB. EACH ON & 2-20MM GUNS 100 RDS./GUN -.064 SKIN INCREASED TO .I FOR FLAK PROTECTION 24X5.5 E.H.P. TIRE CATAPULT FTG. 18 X5.5 E.H.P. TIRE DESCRIPTIVE ARRANGEMENT ARMAMENT & TANKAGE A4D-2 15 JUNE 1959

11/

POWER PLANT

NO. & MODEL····(1) J	65-W-16A or -4B
TYPE	AXIAL FLOW
DIAMETER	31 in.

RATINGS

MILITARY NORMAL

LBS 7000 RPM 8300

SEA LEVEL STATIC

SPEC. WAD N890-B

ORDNANCE

Tr.	IS	П	٠ ٨	CI

ombs	1-Mk.81 Mod.1(250 1b.) 1-Mk.82 Mod.1 (500 1b.)
	I-MK-82 Mod.1 (300 ID.)
	1-Mk.83 Mod. 2 or 3 (1000 lb)
	(1000 10)
	1-Mk-84 Mod.1 (2000 lb.)

1-1050 lb. Stores 1-1480 1b. 1-1660 lb. 1-2025 lb.

1-3500 lb. 1-In-flight Refueling Store

1-Aero 14B Spray Tank

1-Mk.79 Mod.0(1000 lbs.) Fire Bomb or 1-150 gal.DAC Fuel Tank

Missiles 1-ASM-N-7 Bullpup

Pyrotechnics 1-Aero 5A Flare Dispenser

1-pkg (7) 2.75" Aero 3A 1-pkg.(19) 2.75" Aero X7A 1-pkg. (4) 5.00" Aero X10A

Prac. Bombs 1-Aero 5A Prac. bomb cont.

1-150 gal. (DAC)(2 fins) 1-300 gal. (DAC)(no fins) Drop Tanks

(Continued on NOTE page)

MISSION AND DESCRIPTION

The A4D-2 airplane is a light-weight, carrier based, jet attack airplane whose primary mission is the destruction of enemy ground and surface targets. The airplane is also capable of inflight refueling as a tanker or a receiver.

The arrangement is conventional with all-metal semi-monocogue structure and three span low aspect ratio wing. Landing gear, flaps and speed-brakes are hydraulically operated. An electrically operated, fully adjustable stabilizer is used to trim throughout the normal flight range. The aileron, elevator, and rudder systems are hydraulic-power operated. Manual control is provided for emergencies.

This airplane does not have folding wings. The A4D-2 differs from the A4D-1 in that A4D-2 has the following:

- 1. Receiver and tanker in-flight refueling capabilities
- 2. Basic weight increase of 735 Lbs.
- 3. Installation of a"tadpole" rudder.

DEVELOPMENT

First	Flight	 	 March	1956
Servi	e Use	 • • • • •	 July	1957

DIMENSIONS

ING	
AREA	
SPAN	
MAC	10' -9.6"
SWEEPBACK (4 chord).	33.2°
ENGTH	39" -4.8"
HEIGHT	••••••151 -0"
READ	71 -9.8"
	. ,,,,

WEIGHTS

LOADINGS	LBS	L.F.
EMPTY BASIC DESIGN COMBAT MAX. T.O.(FIELD) (CAT.) MAX. LAND (FIELD) (ARREST)	8965 9570 12504 16014 20,000 22,500 16,000 12,000	7.0 5.5 4.2 3.8 5.5 7.0

ALL WEIGHTS ARE ACTUAL

FUEL AND OIL

NO. TANKS	TOL. GALS.	LOCATION
1 1 2	568 2 32 300 or 600	Wing Fuselage Wing Pylons
FUEL GRADE.	(APPT.TCARTE)	JP-4 or -5

OIL

CAPACITY (GALS)......3.2 SPEC (APPLICABLE).....MIL-L-7808

ELECTRONICS

١	Electronics ControlAN/ASQ-17
	Integrated Package consisting of: UHF CommunicationAN/ARC-27A
	IFFAN/APX-6B
	Coder
	Direction FinderAN/ARA-25
	Automatic Dead
	Reckoning ComputerAN/ASN-19
	TACAN
	•

External Stores

Marker-Beacon Rec.....AN/ARN-12 VOR Rec......AN/ARN-14E





		PERFORM	ANCE SUMMAI	RY		
TAKE-OFF LOADING CONDITION		(1) SEA LEVEL STORE DELIVERY 1-1050 LB. STORE 2-150 GAL.EXT.TANKS	③ SEA LEVEL STORE DELIVERY 1-2025 LB STORE 2-300 GAL.EXT.TANKS	(5) 15,000 FT. STORE DELIVERY 3-BULLPUP MISSILES	TLOW ALT.ATTACK 3-ZUNI ROCKET PACKAGES	MODIFIED HIGH ALTITUDE ATTACK 2-CORVUS MISSILES 1-300 GAL.EXT.TAN
TAKE-OFF WEIGHT	1b.	18,324	21,469	17,203	16,947	20,928
Fuel internal/external (JP-5)	1b./1b.	5440/2040	5440/4080	5440/NONE	5440/NONE	5440/2040
Fayload	1b.	1050	2025	1650	1284	3500
	./sq.ft.	70.5	82.6	66.2	65.2	80.5
Stall speeu - power-off	kn.	*122	132	118	117	130
Take-off run at S.L calm (A)	ft.	3160	4750	2660	2570	4500
Take-off run at S.L. 25 kn. wind (A)	ft.	2250	3500	1900	1870	3250
Take-off to clear 50 ft calm (A)	ft.	4070	6250	3500	3400	5800
Max. speed/altitude (A)	kn./ft.	542/7000	530/11,000	529/17,000	535/6,000	527/10,000
Rate of climb at S.L. (A)	fpm.	6550	5150	6400	7200	5300
Time: S.L. to 20,000 ft. (A)	min.	3.9	5.0	3.9	3.5	4.9
Time: S.L. to 30,000 ft. (A)	min.	7.0	9.8	7.1	6.3	9.4
Service ceiling (100 fpm) (A)	ft.	38,100	34,100	38,400	39,700	34,700
Combat range	n.mi.	1310	1520	715	800	1030
Average cruising speed	kn.	405	400	395	400	400
Cruising altitude(s)	ft.		29,700-38,900	35,100-39,500	34,700-39,500	30,300-36,600
Combat radius/mission time (B)	n.mi/hr.	395/2.1	565/2.8	325/1.8	305/1.7	555/2.8
Average cruising speed	kn.	405	400	400	400	400
IFR-Buddy refuelrad./mission time n	.mi./hr.	800/4.0	955/4.8			925/4.8
IFR(30,000') fuel transf./dist. out	lb/n.mi.	3620/525	4050/490			4000/490
COMBAT LOADING CONDITION		②TANKS OFF STORE RETAINED	(4) TANKS OFF STORE RELEASED	6 missiles retained	8 ROCKET PACKAGES RETAINED	10 TANK OFF MISSILES RETAINE
COMBAT WEIGHT	1b.	16,014	14,964	15,027	14,771	18,705
Engine power		MILITARY	MILITARY	MILITARY	MILITARY	MILITARY
Fuel	1b.	FULL INTERNAL	FULL INTERNAL	60% INTERNAL	60% INTERNAL	FULL INTERNAL
Combat speed/combat altitude	kn./ft.	562/.85/S.L.		530/.85/15,000	536/.81/S. L.	506/.86/30.000
Rate of climb/combat altitude	fpm/ft.	8200/S.L.	9100/S. L.	5100/15,000	8350/S.L.	2000/30,000
Combat ceiling (500 fpm)	ft.	40,400	42,100	40,200	41.600	36,500
Rate of climb at S.L.	fpm.	8200	9100	7500	_8400	6350
Max. speed at S.L.	kn/M	562/.85	574/.87	512/.77	536/-81	538/.81
Max. speed/altitude k	n./M/ft	562/.85/2000	574/.87/S.L.	530/.85/16,000	536/.83/5000	538/.83/4000
Max. speed at 35,000 ft.	kn./M	514/.89	523/.91	500/.87	500/.87	488/.85
LANDING WEIGHT	1b.	10,690	10,794	11.181	11.005	
Fuel	1b.	1166	1270	1068	1073	10,939
Stall speed - power-off/appr. pwr.	kn./kn.	91.5/87.3	91.9/87.7	93.6/89.3	92.8/88.6	92.5/88.3
Distgrnd.run/over 50 ft. obstacle	ft./ft.	2550/3265	2580/3295	2690/3405	2640/3355	2620/3335

NOTES

PERFORMANCE BASIS: CONTRACTOR AND NATC Flight Test Results
RANGE AND/OR RADIUS are based on NATC Flight test fuel consumption data.

- (A) Military Rated Thrust
- (B) For effect of JP-4 Fuel on Combat Radius and Mission Time. See Notes page.
- All configurations include wing pylons and in-flight refueling probe; loadings 5, 6, 7 and 8 include guns and ammunition.

MISSION TIME: Any time where fuel is used and distance gained plus combat and refuel allowance times.

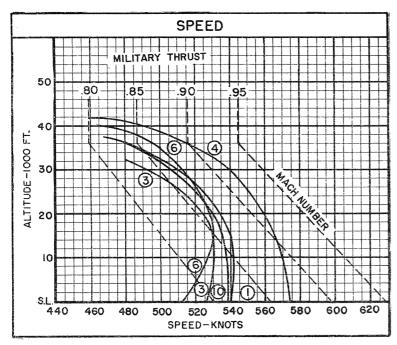
SPOTTING: A total of 106 aircraft can be accommodated in a landing spot on the flight and hanger decks of a CVA-19 class angled-deck carrier.

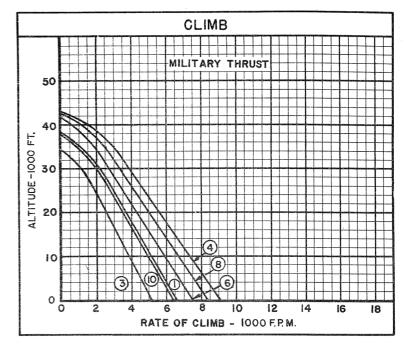
16

VAER-1335D (Rev. 1-5

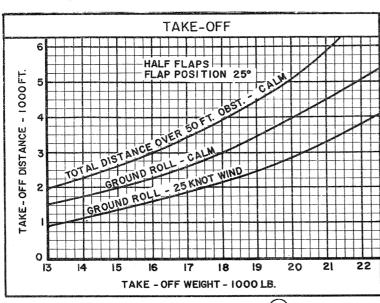


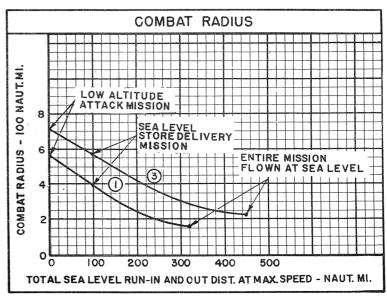






DECLASSIFIED





LOADING CONDITION COLUMN NUMBER

15 JUNE 1959

1

NAVAER 1335E





NOTES

LOADING (All data based on JP-4 fuel)	TAKE-OFF	15,000 FT. STORE	DELIVERY	SEA LEVEL STORE :	DELIVERY
	WEIGHT	Combat Radius	Mission Time	Combat Radius	Mission Time
1-2025 lb. store plus 2-300 gal. ext tanks	16,963 lb.	710 n.mi.	3.7 hr.	525 n.mi.	2.6 hr.
3-Bullpup Missiles		305 n.mi.	1.7 hr.	110 n.mi.	0.6 hr.
2-Corvus Missiles plus 1-300 gal.ext.tank		490 n.mi.	2.5 hr.	310 n.mi.	1.6 hr.

SEA LEVEL STORE DELIVERY

START ENGINE, TAKE-OFF AND ACCELERATE: Fuel for 5 minutes with normal power at sea level.

CLIME OUT: At maximum rate of climb with military power on course to optimum cruise altitude. (RUISE OUT: At maximum range speeds at optimum cruise altitude. (Drop any external tanks when empty.) DESCEND: To sea level (no fuel consumed - no distance covered.)

RUN-IN: At S.L. for 50 N.Mi. at maximum speed with military power. Drop bombs, fire rockets.

military power. Drop bombs, fire rockets.

COMBAT: At sea level for 5 minutes with military power. No distance made good.

RUN-OUT: At sea level for 50 nautical miles at maximum speed with military power.

CLIMB BACK: At maximum rate of climb with military power on course to optimum cruise altitude.

CRUISE BACK: At maximum range airspeeds at optimum cruise altitude.

DESCEND: To sea level (no fuel consumed - no distance covered.)

RESERVE: Fuel allowance; 5% of initial fuel plus 20 minutes at speed for maximum endurance at sea level.

15,000 FT. STORE DELIVERY

START ENGINE, TAKE-OFF AND ACCELERATE: Fuel for 5 minutes with normal power at sea level. CLIMB OUT: At maximum rate of climb with military power on course to optimum cruise altitude. CRUISE OUT: At maximum range speeds at optimum cruise altitude. (Drop any external tanks when empty) DESCEND: To 15,000 ft. (no fuel consumed - no distance covered) drop bombs, fire rockets. COMBAT: At 15,000 ft. for 5 minutes with military power. No distance made good. CLIMB BACK: At maximum rate of climb with military power on course to optimum cruise altitude. CRUISE BACK: At maximum range airspeeds at optimum cruise altitude. DESCEND: To sea level (no fuel consumed - no distance covered.) RESERVE: Fuel allowance; 5% of initial fuel plus 20 minutes at speed for maximum endurance at S. L.

MODIFIED HIGH ALTITUDE ATTACK

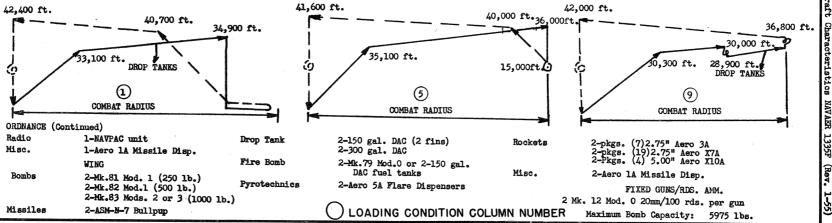
START ENGINE, TAKE-OFF AND ACCELERATE: Fuel for 5 minutes with normal power at sea level. CLIMB OUT: At maximum rate of climb with military power on course to optimum cruise altitude. CRUISE OUT: At maximum range airspeeds at optimum cruise altitude. IN-FLIGHT REFUELING: At 30,000 ft. Fuel allowances for hook-up and flight contingencies = 5 minutes at maximum endurance speeds (no distance made good.) Note: Refueling point limited to return of aircraft to base with normal reserve if refueling contact is not made. CRUISE OUT: Continue to cruise at maximum range airspeeds at optimum cruise altitude. (Drop any external tanks when empty.) DESCEND: To 30,000 ft. (no fuel consumed - no distance covered.) Launch missiles. COMBAT: At 30,000 ft. for 5 minutes with military power. No distance made good. CLIMB BACK: At maximum rate of climb with military power on course to optimum cruise altitude.

CRUISE BACK: At maximum range airspeeds at optimum cruise altitude.

DESCEND: To sea level (no fuel consumed - no

distance covered.)

RESERVE: Fuel allow.: 5% of initial fuel plus 20 min. at speed for maximum endurance at S.L.



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18