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REPORT NO. TSEAL-6-AI

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AAF AIR TECHNICAL SERVICE COMMAND

TACTICAL PLANNING

Characteristics & Performance Chart

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PUBLISHED BY CHIEF, ENGINEERING DIVISION, AIR TECHNICAL SERVICE COMMAND, WRIGHT FIELD, OHIO.

Attention
FUTURE ISSUES OF THIS CHART
WILL BE ON A QUARTERLY BASIS.
THE NEXT ISSUE WILL BE JAN. 1946.

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SEPTEMBER-1945

BOMBER

CARGO

FIGHTER

GLIDER

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3-13
1745

REPORT NO. TSEAL-6-A1

(FORMERLY)

AAF AIR TECHNICAL SERVICE COMMAND

TACTICAL PLANNING

Characteristics & Performance Chart

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BOMBERS

CARGO

FIGHTERS

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MISC.

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IS CHART
RLY BASIS.
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PAGE 2
AS OF:
30 AUGUST 1945

The following changes have been made since the July issue.



Based on new performance data received from from the Wright Field Flight Test Division, and from various contractors, present performance data on the P-47 and the P-51 with rockets, the A-26, C-46, C-47B and C-54E is changed as follows: P-47 - Effect of rockets on take-off, climb, speed and range is shown on page 35.

P-51 - Effect of rockets on speed, radius of action and take-off distance and change in P-51H War Emergency speeds and rates of climb with AN-F-28 fuel shown on page 36.

A-26 - Added rocket data and revised range figures to show increase over those previously shown. This is located on page 23.

C-46 - Increased range data previously shown by adding one more 100-gallon removable fuselage tank. High speed and rate of climb reworked to show slightly lower figures based on a fighter combat weight. May be found on page 25.

C-47 - Added military power rate of climb figures. Decreased ranges shown are due to higher fuel flows than those on which previous performance was based. This information is on page 27.

C-54 - Show increased performance in high speed and rate of climb at altitude. This information is on page 29.

UC-78 - Revised range data. May be found on page 43.

Performance data on the following airplanes was introduced for the first time in this publication and may be found in the Miscellaneous Section: OA-10, L-14, R-5 and R-6.



More and more emphasis is being placed on the B-29 which justifies more emphasis on its performance in this publication. Two technical representatives from ATSC have been sent to XX Air Force in Guam to act as advisors on cruise control and to funnel B-29 performance data to this organization for dissemination. In this issue the B-29 sample mission is presented based on conditions of temperature and humidity encountered in the Marianas. On Okinawa, where summers are hotter and more humid than in the Marianas, performance would be somewhat less than that shown. The Okinawa winter is colder and dryer than in the Marianas; therefore, increased performance may be expected over that shown. The general performance data on the B-29 has been reworked based on the information available in the latest Technical Order. The B-29 Sample Mission is located on page 11 and the general performance on page 13.

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Your

Request for this publication should be addressed to Director, ATSC, Wright Field, Dayton, Ohio.
Attn: TSEAL-6J Flight Data Branch.

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FOREWORD

This chart is a supplement to the Semi-Annual Chart of Engineering Characteristics and Performance, TSEAL-6-A2 "Production". This issue covers typical models now in theater operation and is kept up to date as new models are received by the A.A.F.

All data includes service allowances based on theater experience. Individual airplanes may vary appreciably due to age and service changes. Data printed in red are preliminary and subject to revision after flight check. Data printed in black have been derived from information obtained in flight but are not necessarily actual flight test results. For detailed planning see T.O. listed as references.

WARNING

THESE CHARTS CONTAIN CONSERVATIVE AVERAGES FOR TACTICAL PLANNING AND ARE NOT SUITABLE FOR AERODYNAMIC ANALYSIS.

Performance is based on the requirements of Army-Navy Aeronautical specification AN-H-8a; "Handbooks, Pilot's Flight Operating Instructions" which sets forth specific allowances for practical service operation. Take off and landing distances are 125% of optimum at 3000 ft. on hard surface, no wind, standard temperature. Weights are "Basic" weight plus crew, oil, full ammunition, fuel and bombs, cargo, passengers or troops as applicable. For bombers and cargo, "War Maximum" weights are based on limiting applied positive maneuver factor of 2.00G unless otherwise specified. All speeds shown are true air speeds. Range and endurance are based on the following assumptions:

- (a) Allowance for warm up, taxi, run up, take off, and landing, (equal to ten minutes max. cont. at S.L.).
- (b) Allowances for fuel consumed in climb. Distance and time to climb are included in range and endurance.
- (c) Allowance for carrying bombs and droppable tanks entire flight.
- (d) Allowance for 10% net ideal range and endurance for miscellaneous differences in airplanes, equipment, pilot technique, atmospheric conditions other than wind, unusable fuel, weight, and similar variables. (For example, range varies day and night due to fuel expansion prior to take off.)

THIS REPORT SUPERSEDES PREVIOUS TSEAL-6-A1 PUBLICATIONS; SUPERSEDED PUBLICATIONS, IF NO LONGER OFFICIALLY REQUIRED, WILL BE DESTROYED IN THE MANNER PRESCRIBED IN A.R.380-5.

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DEFINITIONS

CHARACTERISTICS	PERFORMANCE
SUPERCHARGER: "TURBO" indicates exhaust driven turbine with single speed integral diffuser.	BOMBS: Maximum rack capacity for standard sizes.
BHP/ALT: Brake horsepower per engine at critical altitude with ram.	CARGO: Cargo, passenger or troop capacities.
T: Maximum power for take off.	Max. load: One size only on each bomb rack (in some cases a reduction in fuel load is required)
M: Military power for combat at altitude shown, (usually limited to 15 min. duration)	Practical total cargo based on gross weight. Values are at least 125% of optimum.
WE: War emergency power for combat at altitude shown, (limited to 5 min. duration only)	TAKE OFF & LANDING: (At 3000' runway alt.)
C: Maximum power for continuous operation.	Gross weight: (To clear 50')
Note: Maximum cruise power is the maximum power for unlimited operation with lean mixture.	Ground run: Distances representative of minimum runway requirements.
SIZE: Approximate dimensions for storage planning.	Landing speed: Distances representative of minimum runway requirements.
Length: Does not include protruding guns.	High speed & climb: Values are for clean new airplanes.
Height: Maximum in three point position.	Military power: For limited periods of combat operation.
Wing area: Standard aerodynamic surface area.	War emergency power: (Performance with War Emergency ratings are shown when applicable).
Tread: Center to center main wheels: (outboard if dual).	Time to climb: Time to climb from sea level based on maximum continuous power unless otherwise noted.
WEIGHTS: Approximate averages for planning purposes.	RANGE & ENDURANCE: See discussion in "FOREWORD", page 2
B: Includes all equipment, that has a fixed location and is actually present in the airplane; air frame; power plant; and accessories; trapped fuel oil; full hydraulic; cooling and anti-icing fluid systems and reservoirs; armor plate, ordnance (less ammunition & bombs); chemical, navigation, oxygen, pyrotechnics, and radio equipment.	Take off weight: Brief list of typical gross weights.
BASIC WEIGHT:	Bombs, etc: Typical loadings combined with various tank capacities. No consideration is given to loadings requiring partially filled tanks.
C: Combat weight is based on latest confirmed information from theaters and usually includes full built-in fuel, full ammunition, and an arbitrary bomb or cargo load.	Note: Take off weights with corresponding load and fuel quantity applies to all values to the right on the same line.
W: War maximum weight for special missions, limited by structural considerations, based on a positive maneuver factor of 2.0 unless otherwise noted. Wing tanks must be full.	Max. cont. power: High speed cruising with rich mixture. (For emergency cruising only)
M: Recommended max. landing as established in Tech. Order 01-1B-44. (10 March 1945).	Max. cruise power: Maximum continuous operation with lean mixture setting.
FUEL & OIL Standard fuel tankage and capacities as well as max. oil.	Long range: Practical maximum range for planning purposes under all conditions set forth.
ARMAMENT: Resume' of information in the "ARMAMENT & BOMB INSTALLATIONS CHART" TSEAL-6-A5.	RED PRINT: Preliminary estimates or calculated data.
RADIO Resume' of typical models that may be installed.	

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BOEING (SEATTLE) "BO"

MODEL & BLOCK-NO.	ARMAMENT			BOMBS			FUEL U.S. GAL. (MAXIMUM)
	NO GUNS & CALIBER	RDS. PER. GUN	LOCATION & TYPE	INTERNAL	EXTERNAL	MAX. LOAD LB.	
				NO & SIZE	NO & SIZE		
B-17F-BO (-1 thru -27)	1-.30 2-.50 2-.50 2-.50	500 a 300 400 500 565	NOSE...FLEX. SIDE WAIST-FL. UPPER TURRET LOW BALL TRR. TAIL...FLEX.MT.	2-2000 6-1600 6-1000 12- 500 16- 250 24- 100	NONE	3600	2550
B-17F-BO (-30 thru -50)	NONE 1) 2) 2) 2)	NONE 300 b 300 400 500 565	NOSE...FLEX. RADIO COMPART. SIDE WAIST-FL. UPPER TURRET LOW BALL TRR. TAIL...FLEX.MT.	2-2000 6-1600 6-1000 12- 500 16- 250 24- 100	2-4000 2-2000 2-1600 2-1000	6-1600 2-4000x 17600	AS ABOVE
B-17F-BO -55 thru 1st.30(-85)	2) 1) 2) 2) 2)	300 300 400 500 565	NOSE...FLEX. RADIO COMPART. SIDE WAIST-FL. UPPER TURRET LOW BALL TRR. TAIL...FLEX.MT.	A S	A B O V E		2550 (-55 thru -75) 3630 -50 & SUB.
B-17F-BO Last 70(-85) thru -130	3) 1) 2) 2) 2)	300 300 400 500 565	NOSE...FLEX. RADIO COMPART. SIDE WAIST-FL. UPPER TURRET LOW BALL TRR. TAIL...FLEX.MT.	AS ABOVE EXCEPT THAT EXTERNAL BOMB RACKS DELETED AFTER 15th.(-95)			3630
B-17G-BO (-1 thru -110)	2) 1) 2) 2) 2)	Not 610 365 300 c 600 d 400 500 565 e	CHEEK...FLEX. CHIN TURRET RADIO COMPART. SIDE WAIST-FL. UPPER TURRET LOW BALL TRR. TAIL...FLEX.MT.	2-2000 6-1600 6-1000 12- 500 16- 250 24- 100	MOUNT LUGS & CONTROLS RETAINED	9600	AS ABOVE

NOTES: a NOSE GUN DELETED LAST 35(-27). d EARLY MODELS CARRY 300 RDS./GUN.
b RADIO COMPART. GUN ADDED AFTER 16th(-30). e CHEYENNE TAIL MOUNT ON LAST 20(-90)
c RADIO COMP. GUN DELETED LAST 68(-105 & -110). AND SUBSEQUENT. x EXTERNAL

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DOUGLAS (LONG BEACH) "DL"

MODEL & BLOCK-NO.	ARMAMENT			BOMBS			FUEL U.S. GAL. (MAXIMUM)
	NO GUNS & CALIBER	RDS. PER. GUN	LOCATION & TYPE	INTERNAL	EXTERNAL	MAX. LOAD LB.	
				NO & SIZE	NO & SIZE		
B-17F-DL -1 thru 1st.13(-10)	1-.30 2-.50 2-.50 2-.50	500 a 300 400 500 565	NOSE...FLEX. SIDE WAIST-FL. UPPER TURRET LOW BALL TRR. TAIL...FLEX.MT.	2-2000 6-1600 6-1000 12- 500 16- 250 24- 100	NONE	9600	2550
B-17F-DL Last 12(-10) thru -20	2) 1) 2) 2) 2)	300 300 400 500 565	NOSE...FLEX. RADIO COMPART. SIDE WAIST-FL. UPPER TURRET LOW BALL TRR. TAIL...FLEX.MT.	AS ABOVE	PROV. FOR EXTERNAL RACKS ON -20 & SUB.	AS ABOVE	AS ABOVE
B-17F-DL -25 thru 1st.21(-35)	A S	A B O V E	AS ABOVE	PROV. FOR EXTERNAL RACKS	AS ABOVE	AS ABOVE	3630
B-17F-DL Last 19(-35) thru -65	3) 1) 2) 2) 2)	300 300 400 500 565	NOSE...FLEX. RADIO COMPART. SIDE WAIST-FL. UPPER TURRET LOW BALL TRR. TAIL...FLEX.MT.	A S	A B O V E	AS ABOVE	AS ABOVE
B-17G-DL (-1 thru -95)	2) 1) 2) 2) 2)	Not 610 365 300 c 600 d 400 500 565 e	CHEEK...FLEX. CHIN TURRET RADIO COMPART. SIDE WAIST-FL. UPPER TURRET LOW BALL TRR. TAIL...FLEX.MT.	AS ABOVE EXCEPT ONLY MOUNT- ING LUGS & CONTROLS RETAINED FOR EXTERNAL BOMBS ON (-80) & SUBSEQUENT MODELS.		AS ABOVE	AS ABOVE

NOTES: a NO NOSE GUN ON 1st.13(-10). d EARLY MODELS CARRY 300 RDS./GUN.
b NOT CARRIED ON -1 & -5. e CHEYENNE TAIL MOUNT ON (-50) & SUBSQT.
c RADIO COMPART. GUN DELETED (-75) & SUBSQT. x EXTERNAL

VEGA (BURBANK) "VE"

MODEL & BLOCK-NO.	ARMAMENT			BOMBS			FUEL U.S. GAL. (MAXIMUM)
	NO GUNS & CALIBER	RDS. PER. GUN	LOCATION & TYPE	INTERNAL	EXTERNAL	MAX. LOAD LB.	
				NO & SIZE	NO & SIZE		
B-17F-VE (-1 thru -10)	1-.30 CAL. (300 RDS.) IN 1st. 12 AIRPLANES; NOSE GUN DELETED ON NEXT 28 THRU (-10). NO RADIO COMPARTMENT GUN UNTIL LAST 14(-15). OTHER ARMAMENT AS B-17F.		NOSE GUN	2-2000 6-1600 6-1000 12- 500 16- 250 24- 100	NONE	9600	2550
B-17F-VE (-15 & -20)	2) 1) 2) 2) 2)	300 300 400 500 565	NOSE...FLEX. RADIO COMPART. SIDE WAIST-FL. UPPER TURRET LOW BALL TRR. TAIL...FLEX.MT.	AS ABOVE	EXTERNAL RACKS ON Last 31 (-20) & SUB.	AS ABOVE	AS ABOVE
B-17F-VE (-25 & -30)	A S	A B O V E	AS ABOVE	b 2-4000 2-2000 2-1600 2-1000		6-1600 2-4000x 17600	2550 (B-17F-25) 3630 (B-17F-30)
B-17F-VE (-35 thru -50)	3) 1) 2) 2) 2)	300 300 400 500 565	NOSE...FLEX. RADIO COMPART. SIDE WAIST-FL. UPPER TURRET LOW BALL TRR. TAIL...FLEX.MT.	AS ABOVE EXCEPT ONLY MOUNT- ING LUGS & CONTROLS RETAINED FOR EXTERNAL BOMB RACKS ON (-35) & SUBSEQUENT MODELS.			3630
B-17G-VE (-1 thru -110)	2) 1) 2) 2) 2)	Not 610 365 300 e 600 d 400 500 565 c	CHEEK...FLEX. CHIN TURRET RADIO COMPART. SIDE WAIST-FL. UPPER TURRET LOW BALL TRR. TAIL...FLEX.MT.	A S	A B O V E		

NOTES: a NOT ON (-1) & (-5). d EARLY MODELS CARRIED 300 RDS./GUN.
b EXTERNAL RACKS DELETED LAST 4(-30); e RADIO GUN STOWED (-35) thru 1st. 16(-35);
LUGS & CONTROLS RETAINED. c RADIO GUN DELETED LAST 24(-85) & SUBSQT. x EXTERNAL
c CHEYENNE TAIL MOUNT ON LAST 11(-55) & SUBSQT.

GENERAL NOTE:

THE 6x1600 LB. ARMOR PIERCING BOMBS
MAY BE CARRIED, BUT BOMB RAIL LOAD
FACTORS AND CLEARANCES ARE REDUCED
AND MANEUVERING OF AIRPLANE MUST BE
LIMITED ACCORDINGLY.

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Page 3A
As of:
11 August 1945.

SECURITY CLASSIFICATION NOTICE

Due to a very poor printing job a number of security classification designations are incorrectly shown on page 3, the classification page. The following is submitted for your information and is the correct security classification for the material presented in the 1 August 1945 issue of Report No. TSEA-6-A1:

1. B-24 Radio Restricted.
2. B-29 Fuel, Guns, Bombs (number and size internal) Restricted.
B-29 Bombs (Maximum load), Radio Confidential.
3. B-32 Combat Crew, Fuel, Guns, Bombs (No. & size inter), Radio Restricted.
B-32 Bombs (Maximum load) Confidential.
4. B-25 Rocket Inst., Radio Restricted.
5. A-20 Rocket Inst., Restricted.
6. A-26 Bombs Restricted.
7. All Cargo Cargo Information Restricted.
8. P-47 Rocket Inst. Restricted.
9. P-51 Rockets Restricted.
10. P-61 Engine (Supercharger), Bombs (Maximum load) Restricted.
11. P-80 Size, Restricted.

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TACTICAL PLANNING CHARACTERISTICS & PERFORMANCE CHART

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MODEL & BLOCK-NO.	ENGINE & PROP.		SIZE		WEIGHT		COMBAT CREW	FUEL & OIL		ARMAMENT			BOMBS		RADIO	REMARKS & REFERENCE		
	NUMBER	ENGINE MFR. MODEL SUPERCHARGER PROP. MFR. TYPE	B.H.P./ALT.		S-SPAN L-LENGTH H-HEIGHT T-TREAD W-WING AREA	B-BASIC C-COMBAT W-WAR MAX. M-MAX LAND		NOTES	FUEL TANKAGE		NO. AND SIZE			INTERNAL	EXTERNAL		MAX. LOAD	TYPICAL MODELS CARRIED
			T-TAKE OFF W-WAR EMERG. M-MILITARY C-CONTINUOUS	T-TAKE OFF W-WAR EMERG. M-MILITARY C-CONTINUOUS					TYPE OR LOCATION ALSO (MAX. OIL)	NO. AND CAPACITY ALSO (MAX. FUEL)	NO. GUNS AND SIZE	RDS. PER GUN	LOCATION AND TYPE					
B-17F (LATEST MODEL)	4	WRIGHT R-1820-97 TURBO HAMILTON STD. 11"7" DIAL-3 BL. F.F., HYDROMATIC	T 1200/3 L W 1380/25000a M 1200/27000 C 1000/30000	S 103'9" L 74'9" H 19'1" T 21'2" W 1420 Sq. ft.	B 39000 C 55000 W 72000* M 50000	ABSOLUTELY ESSENTIAL THAT ALL THESE WEIGHTS & THAT MAXIMUM VIEW BE KEPT TO A MINIMUM.	MAIN - WINGS 5-1730 WING TIPS 2 x 540 BOMB BAY 2 x 410 (148)	31 21 11 21 21 21	300 300 300 400 500 555	NOSE - FLEX. SIDE WAIST-FLEX. RADIO COMPART. UPPER TURRET LOW BALL TURRET TAIL-FLEX. MOUNT	2-2000 6-1900 6-1000 12-500 18-250 24-100	2-4000 2-2000 2-1800 2-1000	6-1600 2-4000 17500	SEE NOTE (A)	REFER TO PAGE 4 FOR "FOREWORD." REFER TO PAGE 3 FOR SECURITY CLASSIFICATION. REFER TO PAGE 5 FOR DEFINITIONS. REFER TO T.O. LISTED FOR DETAILED PLANNING.			
B-17G (LATEST MODEL)	4	AS ABOVE	T 1200/3 L W 1380/28700 a M 1200/32700 b C 1000/35200 b	S 103'9" L 74'9" H 19'1" T 21'2" W 1420 Sq. ft.	AS ABOVE	AS ABOVE EXCEPT PROVISIONS ONLY, FOR THE BOMB BAY FUEL TANKS.	AS ABOVE	21 21 21 21 21 21	315 Tot. 385 400 400 500 565	CHEEK - FLEX. CHIM TURRET SIDE WAIST-FLEX. UPPER TURRET LOW BALL TURRET CHEYENNE TAIL MT.	AS ABOVE	MOUNTING LUGS & CONTROLS RETAINED FOR EXTERNAL RACKS.	9600	AS ABOVE	B-17F TECH. ORDER 01-20EF-1 (REV. 10 FEBRUARY 1945) B-17G TECH. ORDER 01-20EG-1 (REV. 25 FEBRUARY 1945) (a) WAR EMERGENCY POWER PERMITTED ONLY WHEN CARBURETORS MODIFIED TO PARTS LIST NO. 395563-3. (b) EARLIER B-17G'S HAD CRITICAL ALTITUDE SHOWN FOR B-17F MODELS. (c) EARLY MODELS HAVE 74'9" LENGTH. (d) SEE PAGE OPPOSITE FOR DETAILED ARMAMENT BREAKDOWN.			

MODEL & BLOCK-NO.	TAKE OFF & LANDING HARD SURFACE-NO WIND				HIGH SPEED & CLIMB CLEAN AIRPLANE AT NORMAL COMBAT WEIGHT								RANGE AND ENDURANCE STATUTE AIR MILES-NO WIND- NO ALLOWANCE FOR RESERVE										REMARKS				
	GROSS WEIGHT	TO CLEAR 50'		LANDING SPEED	GROUND RUN		STD. ALT.	WAR EMERG.		MIL. POWER		MAX. CONTIN. POWER		LOADING			MAX. CONTINUOUS POWER		MAX. CRUISE POWER		LONG RANGE						
		T.O. DIST.	LAND DIST.		T.O. DIST.	LAND DIST.		HIGH SPEED	RATE OF CLIMB	HIGH SPEED	RATE OF CLIMB	HIGH SPEED	RATE OF CLIMB	TIME TO CLIMB	TAKE OFF WEIGHT	BOMBS CARG. BASE	TOTAL FUEL	at 10000 FT.	at 25000 FT.	at 10000 FT.	at 25000 FT.	at 10000 FT.		at 10000 FT.			
B-17F	40000 45000 50000 55000 60000 65000	- - 3200 4000 4800 5300	3000 3500 4000 4500 5000 5500	90 95 100 105 110 115	1200 1500 1900 2400 3700 4900	30000 25000 20000 15000 10000 5000	310 314 303 288 278 265	550 1025 1150 1250 1350 1425	298 299 288 274 263 252	400 725 850 950 1050 1125	280 299 286 255 245 236	225 36.2 25.7 17.3 10.2 4.0	52.0 37.0 25.7 17.3 10.2 4.0	65500 59500 53500 47500 41500 35500	NONE NONE 4000 8000	3630 2810 2810 2810 2810 2810	1400 1400 1400 1400 1400 1400	5.1 5.1 5.1 5.1 5.1 5.1	1550 1550 1550 1550 1550 1550	6.2 6.2 6.2 6.2 6.2 6.2	2050 2050 2050 2050 2050 2050	10.1 10.1 10.1 10.1 10.1 10.1	2150 2150 2150 2150 2150 2150	9.9 9.9 9.9 9.9 9.9 9.9	2800 2700 2600 2500 2400 2300	17.4 17.4 17.2 17.2 17.2 17.2	(a) FIRST 3 HOURS OF FLIGHT AT 20000 FT. ALTITUDE. (b) FIRST 4 HOURS OF FLIGHT AT 20000 FT. ALTITUDE. (c) MAX. FERRY RANGE MAY BE RESTRICTED BY INSUFFICIENT OIL IF ENGINE IS NOT IN BETTER THAN AVERAGE CONDITION; OIL CONSUMPTION VARIES FROM 3 QT. TO 9 GAL. PER HR. PER ENGINE DEPENDING ON TIME SINCE OVERHAUL.
B-17G	AS ABOVE	AS ABOVE	AS ABOVE	AS ABOVE	AS ABOVE	30000 25000 20000 15000 10000 5000	310 302 291 280 269 258	550 1050 1200 1300 1400	295 287 277 267 256 246	400 750 850 950 1000 1100	250 255 245 234 223 210	54.0 37.0 26.0 16.0 7.0	65500 59500 53500 47500 41500 35500	NONE NONE 4000 8000	3630 2810 2810 2810 2810 2810	1400 1400 1400 1400 1400 1400	5.3 5.3 5.3 5.3 5.3 5.3	1500 1440 1550 1650 1800 1800	6.2 6.2 6.2 6.2 6.2 6.2	2000 1950 1800 1800 1800 1800	10.0 10.0 9.8 9.8 9.8 9.8	2250 2150 2050 2050 2050 2050	8.5 8.7 8.7 8.7 8.7 8.7	2500 2350 2250 2250 2250 2250	17.10 14.4 13.0 12.5	(d) MAX. FERRY RANGE MAY BE RESTRICTED BY INSUFFICIENT OIL IF ENGINE IS NOT IN BETTER THAN AVERAGE CONDITION; OIL CONSUMPTION VARIES FROM 3 QT. TO 9 GAL. PER HR. PER ENGINE DEPENDING ON TIME SINCE OVERHAUL.	

NOTES: (A) EARLIER MODEL... POSSIBLE THEATER USE ONLY... RED FIGURES ARE PRELIMINARY: SUBJECT TO REVISION AFTER FLIGHT CHECK.



B-17

"FORTRESS"

MANUFACTURED BY BOEING - DOUGLAS - VEGA

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CONSOLIDATED (SAN DIEGO) "CO"

MODEL & BLOCK-NO.	ARMAMENT			BOMBS			FUEL U.S. GAL. (MAXIMUM)
	NO GUNS & CALIBER	RDS. PER GUN	LOCATION & TYPE	INTERNAL	EXTERNAL	MAX. LOAD LB.	
				NO & SIZE	NO & SIZE		
B-24D-CO D thru 1st. 94(-15)	1) 2) .50 2)	200 400 600	NOSE.....FLEX. UPPER TURRET TAIL TURRET	4-2000 8-1000 12- 500 12- 250 20- 100	NONE	8000	2364
B-24D-CO Last 36(-15) thru -20	1) 2) .50 2)	200 400 600	NOSE.....FLEX. UPPER TURRET TUNNEL...FLEX. TAIL TURRET	A S	A B O V E		2364 Last 36(-15) 3614 -20 & sub.
B-24D-CO -25 thru 1st. 26(-110)	3) 2) .50 1) 2)	100 250 400 100 600	NOSE.....FLEX. SIDE WAIST-FL. UPPER TURRET TUNNEL...FLEX. TAIL TURRET	4-2000 8-1600 8-1000 12- 500 12- 250 20- 100	NONE	12800	3614
B-24D-CO Last 9(-110) thru -170	3) 2) .50 2) 2)	100 250 400 508 600	NOSE.....FLEX. SIDE WAIST-FL. UPPER TURRET LOW BALL TR. TAIL TURRET	A S	A B O V E		AS ABOVE
B-24L-CO (-1 thru -210) B-24M-CO (-1 thru -20) B-24N-CO (-1 thru -50)	2) 2) .50 2) 2)	600 250 ^b 400 508 600	NOSE TURRET SIDE WAIST-FL. UPPER TURRET LOW BALL TR. See "a"	A S	A B O V E		AS ABOVE

NOTES: a B-24J's HAVE TAIL TURRET; B-24L's HAVE HAND HELD GUNS IN TAIL &
 * NOT ON B-24D-25 B-24M's HAVE LIGHT WEIGHT TAIL TURRET.
 b B-24N-20 HAS PROVISIONS FOR ONLY 250 RDS. PER GUN AT SIDE WAIST POSITION.

CONSOLIDATED (PT. WORTH) "CP"

MODEL & BLOCK-NO.	ARMAMENT			BOMBS			FUEL U.S. GAL. (MAXIMUM)
	NO GUNS & CALIBER	RDS. PER GUN	LOCATION & TYPE	INTERNAL	EXTERNAL	MAX. LOAD LB.	
				NO & SIZE	NO & SIZE		
B-24D-CF (-1 & -5)	3) 2) 2) .50 1) 2)	100 250 400 100 600	NOSE.....FLEX. SIDE WAIST-FL. UPPER TURRET TUNNEL...FLEX. TAIL TURRET	4-2000 8-1000 12- 500 12- 250 20- 100	NONE	8000	3614
B-24D-CF (-10 thru -20)	A S	A B O V E				12800	AS ABOVE
B-24J-CF (-1 thru -105) B-24H-CF (-1 thru -15)	2) 2) .50 2) 2)	600 250 400 508 600	NOSE TURRET SIDE WAIST-FL. UPPER TURRET LOW BALL TR. TAIL TURRET	A S	A B O V E (SEE a)		AS ABOVE
B-24H-CF (-20 thru -30)	2) 2) .50 2) 2)	600 500 ^b 400 508 600	NOSE TURRET S. WAIST-EG MT. UPPER TURRET LOW BALL TR. TAIL TURRET	A S	A B O V E		AS ABOVE
B-24J-CF (-101)	A S	A B O V E					E

NOTES: a B-10 SHACKLES FOR 8-1600 LB. BOMBS START ON 2101st. FORD KNOCK-DOWN SET.
 b B-24H-20 HAS PROVISIONS FOR ONLY 250 RDS. PER GUN AT SIDE WAIST POSITION.

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DOUGLAS (Tulsa) "DT" & FORD (Willow Run) "FO"

MODEL & BLOCK-NO.	ARMAMENT			BOMBS			FUEL U.S. GAL. (MAXIMUM)
	NO GUNS & CALIBER	RDS. PER GUN	LOCATION & TYPE	INTERNAL	EXTERNAL	MAX. LOAD LB.	
				NO & SIZE	NO & SIZE		
B-24E-DT & FO (-1 thru -15)	2) 2) .50 2) 2)	600 250 400 508 600	NOSE TURRET SIDE WAIST-FL. UPPER TURRET LOW BALL TR. TAIL TURRET	4-2000 8-1600 8-1000 12- 500 12- 250 20- 100	NONE	12800	3614
B-24E-DT & FO (-20 thru -30)	2) 2) .50 2) 2)	600 500 ^b 400 508 600	NOSE TURRET S. WAIST-EG MT. UPPER TURRET LOW BALL TR. TAIL TURRET	A S	A B O V E		
B-24E-DT (-1 thru -10) B-24E-FO (-1 thru -20)	B-24E-DT & FO AIRPLANES RECEIVE NEW B-24J DESIGNATIONS WHEN C-1 AUTOMATIC PILOT & M-9 BOMB SIGHT REPLACE A-5 AUTOMATIC PILOT & S-1 BOMB SIGHT.						
B-24L-FO (-1 thru -20) B-24M-FO (-1 thru -30)	2) 2) .50 2) 2)	600 500 400 508 600	NOSE TURRET S. WAIST-EG MT. UPPER TURRET LOW BALL TR. TAIL-HAND HELD	4-2000 8-1600 8-1000 12- 500 12- 250 20- 100	NONE	12800	3614

NOTES: a B-10 SHACKLES FOR 8-1600 A.P. BOMBS START ON 2101st. FORD KNOCK-DOWN SET.
 b B-24M-20's HAVE PROVISIONS FOR ONLY 250 ROUNDS PER GUN.

NORTH AMERICAN (DALLAS) "NT"

MODEL & BLOCK-NO.	ARMAMENT			BOMBS			FUEL U.S. GAL. (MAXIMUM)
	NO GUNS & CALIBER	RDS. PER GUN	LOCATION & TYPE	INTERNAL	EXTERNAL	MAX. LOAD LB.	
				NO & SIZE	NO & SIZE		
B-24G-NT	3) 2) 2) .50 2) 2)	230 TOT. 250 400 508 600	NOSE.....FLEX. SIDE WAIST-FL. UPPER TURRET LOW BALL TR. TAIL TURRET	4-2000 8-1000 12- 500 12- 250 20- 100	NONE	8000	3614
B-24G-NT (-1 thru -5)	2) 2) .50 2) 2)	600 250 400 508 600	NOSE TURRET SIDE WAIST-FL. UPPER TURRET LOW BALL TR. TAIL TURRET	A S	A B O V E		AS ABOVE
B-24G-NT (-10 thru -15)	A S	A B O V E				12800	AS ABOVE
B-24J-NT (-1 & -5)	A S	A B O V E					E

NOTES: a LOWER BALL TURRET ON 5th. & SUBSEQUENT; NO TUNNEL GUN ON 1st. 4. ARTICLES.
 B-24G-20-NT AIRPLANES ARE DESIGNATED B-24J-1, ETC. WHEN
 C-1 AUTOMATIC PILOT & M-9 BOMB SIGHT REPLACE A-5 AUTO-
 MATIC PILOT & S-1 BOMB SIGHT.

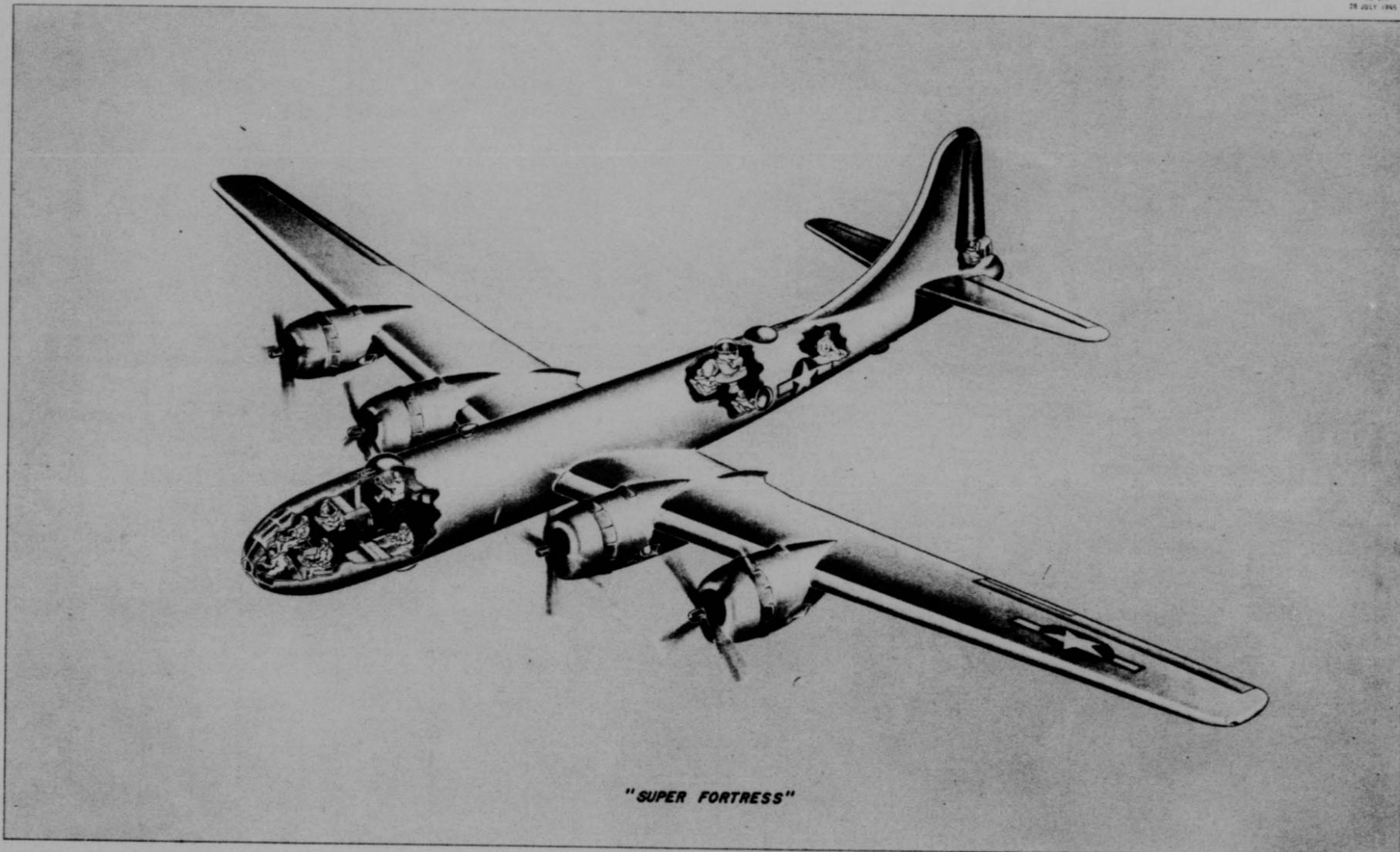
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SAMPLE B-29 MISSION



TOTAL STAT. MILES	0	33	1653	1693	1893	2285	3436	3586
TOTAL NAUT. MILES	0	29	1435	1470	1640	1980	2990	3110
TOTAL TIME		0.2	7.6	7.8	8.5	10.2	16.1	16.9
TOTAL FUEL	170	351	3350	3552	4144	4657	6197	6397
CAL. AIR SPEED	190	202-198	190	240-250	184-180	180-176	178	0
COWL FLAP ANGLE	W.O.-7°	1°	2½°	1½°	0°	0°	0°	0°
WEIGHT	137400	136300	118300	117100	97500	94500	85200	84000

Note:

- FUEL LISTED IN FIRST COLUMN IS ALLOWANCE FOR WARM UP TAKE-OFF AND CLIMB BASED ON 10 MINUTES OPERATION AT MAXIMUM CONTINUOUS POWER AT SEA LEVEL.
- NO ALLOWANCE HAS BEEN MADE IN RANGE FOR FORMATION FLIGHT, DIFFERENCES IN AIRPLANES, PILOT TECHNIQUE, WIND ETC.
- THE ABOVE DATA IS FOR COMBAT B-29 WITH TURRETS, BLISTERS, RADAR DOME, SHORT CHORD COWL FLAPS, LATEST ENGINE BAFFEEES AND IS BASED ON MARIANAS DATA. THE B-29 IS GREATLY AFFECTED BY TEMPERATURE CHANGES. AS AN EXAMPLE AN INCREASE OF 10° C IN TEMPERATURE HAS THE SAME EFFECT ON RATE OF CLIMB AS ADDING 5000 LB. TO SHIP WEIGHT.

GROSS WT. - 138400 LB.
 FUEL - 6988 GAL.
 BOMBS - 16000 LB.
 RANGE OF 3586 MILES
 (WITH 591 GAL. RESERVE)

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BOEING (ATLANTA) "BA"

MODEL & BLOCK NO.	ARMAMENT			FUEL		
	NO GUNS & CALIBER	RDS. PER. GUN	LOCATION & TYPE	TANK TYPE & LOCATION	NO. TANKS & CAPACITY	MAX. CAPACITY (U. S. GALS)
B-29-BA -1	1-20MM. 2) .50 2) 2)	100 1000* 1000* 1000* 1000*	TAIL, NR, TYPE B UPPER POSE TER. UPPER REAR TER. LOWER POSE TER. LOWER REAR TER. TAIL POWER MT.	OUTRD. WINGS IMRD. WINGS BOMB BAY	11-2726 8-2912 1x 640	8198
B-29-BA -5 thru 1st.9 (-15)	A S	A B O V E		OUTRD. WINGS IMRD. WINGS CENTER WING BOMB BAY	11-2726 8-2912 1-1350 1x 640	9516
B-29-BA 1st.11(-15) thru (-20)	1-20MM. 4) .50 2) 2) 2)	100 875* 1000* 1000* 1000* 1000*	TAIL, NR, TYPE B UPPER POSE TER. UPPER REAR TER. LOWER POSE TER. LOWER REAR TER. TAIL POWER MT.	A S	A B O V E	
B-29-BA -25 thru 23 (-48)	4) 2) .50 2) 2)	875* 1000* 1000* 1000* 1000*	UPPER POSE TER. UPPER REAR TER. LOWER POSE TER. LOWER REAR TER. TAIL POWER MT.	A S	A B O V E	
B-29-BA -30 thru -60 (SEE **)	2-.50	1000*	POWER TAIL MT.	A S	A B O V E	

NOTES: * FILLER STRIPS ARE ADDED IN AMMUNITION CANS TO REDUCE ROUNDS TO 800 PER GUN.
** 2 CAL. 50 ON -30, -35, 1st. 47 (-40)
3 CAL. 50 ON LAST 15 (-40) THRU -66

BOEING (BOSTON) "BF"

MODEL & BLOCK NO.	ARMAMENT			FUEL		
	NO GUNS & CALIBER	RDS. PER. GUN	LOCATION & TYPE	TANK TYPE & LOCATION	NO. TANKS & CAPACITY	MAX. CAPACITY (U. S. GALS)
B-29-BF -1	1-20MM. 2) .50 2) 2)	100 1000* 1000* 1000* 1000*	TAIL, NR, TYPE B UPPER POSE TER. UPPER REAR TER. LOWER POSE TER. LOWER REAR TER. TAIL POWER MT.	OUTRD. WINGS IMRD. WINGS BOMB BAY	11-2726 8-2912 1x 640	8198
B-29-BF -5 thru 1st.11 (-10)	A S	A B O V E		OUTRD. WINGS IMRD. WINGS CENTER WING BOMB BAY	11-2726 8-2912 3-1100 1x 640	9298
B-29-BF 1st.9 (-10) thru (-15)	2) 2) .50 2) 2)	1000* 1000* 1000* 1000* 1000*	UPPER POSE TER. UPPER REAR TER. LOWER POSE TER. LOWER REAR TER. TAIL POWER MT.	A S	A B O V E	
B-29-BF -20 & -40	4) 2) .50 2) 2)	875* 1000* 1000* 1000* 1000*	UPPER POSE TER. UPPER REAR TER. LOWER POSE TER. LOWER REAR TER. POWER TAIL MT.	A S	A B O V E	

NOTES: * FILLER STRIPS ARE ADDED IN AMMUNITION CANS TO REDUCE ROUNDS TO 800 PER GUN.

TRAPPED UNAVAILABLE FUEL The following table is based on transfer of fuel from center section and bomb bay tanks with the body center line at 0° and wings level.

FLIGHT ATTITUDE WINGS LEVEL OF BODY	TOTAL TRAPPED FUEL (GAL.)
4° UP	87
2° UP	81
0°	93
2° DOWN	121
4° DOWN	163
6° DOWN	207
8° DOWN	249
15° DOWN	531
20° DOWN	667

NOTE

It is probable that a 20° nose down attitude will be reached while making a normal approach, however, if conditions warrant, an approach may be made as flat as 6° to 8° nose down.

BOEING (WICHITA) "BW"

MODEL & BLOCK NO.	ARMAMENT			FUEL		
	NO GUNS & CALIBER	RDS. PER. GUN	LOCATION & TYPE	TANK TYPE & LOCATION	NO. TANKS & CAPACITY	MAX. CAPACITY (U. S. GALS)
B-29-BW -1 thru -30	1-20MM. 2) .50 2) 2)	100 1000* 1000* 1000* 1000*	TAIL, NR, TYPE B UPPER POSE TER. UPPER REAR TER. LOWER POSE TER. LOWER REAR TER. TAIL POWER MT.	OUTRD. WINGS IMRD. WINGS BOMB BAY	11-2726 8-2912 1x 640	8198
B-29-BW -25 thru 1st.25 (-40)	A S	A B O V E		OUTRD. WINGS IMRD. WINGS CENTER WING BOMB BAY	11-2726 8-2912 1-1350 1x 640	9516
B-29-BW LAST 75(-40) thru 1st.80(-50)	1-20MM. 4) .50 2) 2) 2)	100 875* 1000* 1000* 1000* 1000*	TAIL, NR, TYPE B UPPER POSE TER. UPPER REAR TER. LOWER POSE TER. LOWER REAR TER. TAIL POWER MT.	A S	A B O V E	
B-29-BW Last 20(-50) thru (-80)	4) 2) .50 2) 2)	875* 1000* 1000* 1000* 1000*	UPPER POSE TER. UPPER REAR TER. LOWER POSE TER. LOWER REAR TER. TAIL POWER MT.	A S	A B O V E	

NOTES: * FILLER STRIPS ARE ADDED IN AMMUNITION CANS TO REDUCE ROUNDS TO 800 PER GUN.

MARTIN (OMAHA) "M"

MODEL & BLOCK NO.	ARMAMENT			FUEL		
	NO GUNS & CALIBER	RDS. PER. GUN	LOCATION & TYPE	TANK TYPE & LOCATION	NO. TANKS & CAPACITY	MAX. CAPACITY (U. S. GALS)
B-29-M -1 thru 1st.9 (-20)	1-20MM. 4) .50 2) 2)	100 875* 1000* 1000* 1000*	TAIL, NR, TYPE B UPPER POSE TER. UPPER REAR TER. LOWER POSE TER. LOWER REAR TER. TAIL POWER MT.	OUTRD. WINGS IMRD. WINGS CENTER WING BOMB BAY	11-2726 8-2912 1-1350 1x 640	9516
B-29-M Last 19(-20) thru (-40)	4) 2) .50 2) 2)	875* 1000* 1000* 1000* 1000*	UPPER POSE TER. UPPER REAR TER. LOWER POSE TER. LOWER REAR TER. TAIL POWER MT.	A S	A B O V E	

NOTES: * FILLER STRIPS ARE ADDED IN AMMUNITION CANS TO REDUCE ROUNDS TO 800 PER GUN.

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TACTICAL PLANNING CHARACTERISTICS & PERFORMANCE CHART																		
MODEL & BLOCK-NO.	ENGINE & PROP.		SIZE		WEIGHT		COMBAT CREW	FUEL & OIL		ARMAMENT			BOMBS		RADIO	REMARKS & REFERENCE		
	NUMBER	ENGINE MFR. MODEL SUPERCHARGER PROP. MFR. TYPE	B.H.P./ALT.		S-SPAN L-LENGTH H-HEIGHT T-TREAD W-WING AREA	B-BASIC C-COMBAT W-WAR MAX. M-MAX LAND		NOTES	FUEL TANKAGE		NO. ROCKETS-SIZE-INSTALLATION			NO. AND SIZE		MAX. LOAD	TYPICAL MODELS CARRIED	REFER TO PAGE 4 FOR "FOREWORD" REFER TO PAGE 3 FOR SECURITY CLASSIFICATION REFER TO PAGE 5 FOR DEFINITIONS. REFER TO T.O. LISTED FOR DETAILED PLANNING.
			T-TAKE OFF W-WAR EMERG. M-MILITARY C-CONTINUOUS	T-TAKE OFF W-WAR EMERG. M-MILITARY C-CONTINUOUS					TYPE OR LOCATION ALSO (MAX. OIL)	NO. AND CAPACITY ALSO (MAX. FUEL)	NO. GUNS AND SIZE	RDS PER GUN	LOCATION AND TYPE	INTERNAL	EXTERNAL			
B-29 & A (LATEST MODEL)	1	WEIGHT R-3350-23, -234, -41, -57 or -58 TURBO HAMILTON STD. 157" DI. & -4 BL. F.F., HYDRASTIC	T 2320/11.1 W 2430/32000 M 2310/33500 C 2120/36200	S 141' 3" L 99' H 29' 7" T 31' 5" W 1736 sq. ft.	B 75000 C 137000 W 152000* M 120000	WEIGHT INCLUDES NOT EXCEEDS LIMITS. SEE NOTE (A) FOR CHARACTERISTICS.	11 12	OUTSIDE-WINGS 14-2728 14-2912 21-50 21-50 21-50 21-50	40 20 20 20 20 20	300 300 300 300 300 300	UPPER FURT. TAIL 8-2000 LOWER FURT. TAIL 12-1400 LOWER REAR TAIL 12-1000 TAIL POWER MT.	14-4000 8-2000 12-1400 12-1000 40-300	20000	SEE NOTE (A)	B-29, A & B TECH. ORDER 01-20EJA-1 (REV. 11 JULY 1945)			
B-29B (LATEST MODEL)	4	AS ABOVE	AS ABOVE	AS ABOVE	B 72700 C 137000 W 152000* M 120000		10	AS ABOVE	2-50	300	TAIL POWER MT.	AS ABOVE	AS ABOVE	(a) WAR EMERGENCY RATING WITH 50.5" H.G. AND 28000 RPM. (b) OUTBOARD TREAD 31" 4"; INBOARD TREAD 25" 5" (c) SEE PAGE OPPOSITE FOR ARMAMENT BREAKDOWN. AN/APG-15B USED WITH TAIL POWER MOUNT ON B-29B TO GIVE RADAR GUN SIGHTING FOR USE AGAINST NIGHT FIGHTERS.				

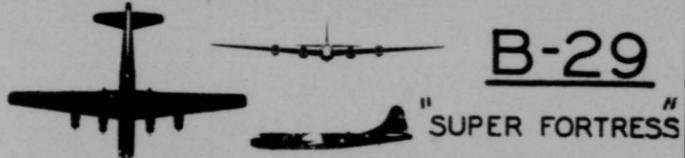
MODEL & BLOCK-NO.	TAKE OFF & LANDING HARD SURFACE-NO WIND				HIGH SPEED & CLIMB CLEAN AIRPLANE AT NORMAL COMBAT WEIGHT						RANGE AND ENDURANCE STATUTE AIR MILES-NO WIND- NO ALLOWANCE FOR RESERVE										REMARKS					
	TO CLEAR 50'		GROUND RUN		WEIGHT	STD. ALT.	WAR EMERG.		MIL. POWER		MAX. CONTIN. POWER		LOADING			BOMBING ALTITUDE				FERRY						
	T.O. DIST.	LAND DIST.	T.O. DIST.	LAND DIST.			HIGH SPEED	RATE OF CLIMB	HIGH SPEED	RATE OF CLIMB	HIGH SPEED	RATE OF CLIMB	TIME TO CLIMB	TAKE OFF WEIGHT	BOMBS CARRIED	TOTAL FUEL	at 5000 FT.		at 15000 FT.			at 25000 FT.		at 30000 FT.		at 10000 FT.
B-29 & A	80000	-	-	-	30000	355	350	353	300	351	150	72.0	141000	NONE	9548	NO BOMBS	NO BOMBS	NO BOMBS	NO BOMBS	NO BOMBS	NO BOMBS	NO BOMBS	NO BOMBS	5500	25.2	(A) CLIMB DATA IS FOR 120000 LB.
B-29B	95000	4600	110	3100	20000	346	825	340	575	330	425	38.0	138500	7000	8258	4550	22.2	4575	20.8	4500	20.0	4400	19.4	4775	21.8	* B-29A MODELS CARRY ONLY 1100 GALL. FUEL IN CENTER WING TANKS DUE TO DIFFERENT INTERNAL WING STRUCTURE. SEE PAGE OPPOSITE FOR FUEL TANKAGE OF INDIVIDUAL BLOCKS.

RANGE AND ENDURANCE NOTICE
 IN ORDER TO PRESENT A MORE TACTICALLY CORRECT RANGE SUMMARY THE FOLLOWING WAS USED AS A BASIS FOR THE B-29 RANGE AND ENDURANCE FIGURES SHOWN ABOVE:
 (1) FERRY AT 10,000' - CLIMB TO 10,000' FROM TAKE-OFF, CRUISE ENTIRE DISTANCE AT 10,000'. OTHER CONDITIONS GIVEN IN "FOREWORD" AND "DEFINITIONS" EXCEPT RANGE CONTAINS DISTANCE COVERED IN DESCENT.
 (2) 5,000' BOMBING ALTITUDE - CLIMB TO 5,000' FROM TAKE-OFF, CRUISE AT 5,000' TO POINT 100 MILES FROM TARGET, 100 MILES AT MAXIMUM CONTINUOUS POWER BEFORE AND AFTER DROPPING BOMBS AT TARGET, CRUISE RETURN IS AT 5,000. OTHER CONDITIONS ARE AS GIVEN IN "FOREWORD" AND "DEFINITIONS" EXCEPT RANGE CONTAINS DISTANCE COVERED IN DESCENT. TARGET DISTANCE IS EQUAL TO 1/2 THE GIVEN RANGE.
 (3) 15,000', 25,000' AND 30,000' BOMBING ALTITUDES - CLIMB TO 5,000 FROM TAKE-OFF, CRUISE AT 5,000 TO POINT WHERE CLIMB TO BOMBING ALTITUDE (15,000', 25,000' OR 30,000') WILL END 100 MILES FROM TARGET, 100 MILES AT MAXIMUM CONTINUOUS POWER BEFORE AND AFTER DROPPING BOMBS AT TARGET, DESCEND TO 15,000' FOR CRUISE RETURN. OTHER CONDITIONS ARE AS GIVEN IN "FOREWORD" AND "DEFINITIONS" EXCEPT RANGE CONTAINS DISTANCE COVERED IN DESCENT. TARGET DISTANCE IS EQUAL TO 1/2 THE GIVEN RANGE.
 BASIC DATA FOR THE RANGE AND ENDURANCE FIGURES IS FROM T.O. 01-20EJA-1 WHICH WILL BE PUBLISHED SEPTEMBER 1945.

NOTES:

B-29 & A	B-29B	NAME
AN/ARR-5	AN/ARR-5	GUIDE PATH
AN/ARR-10	AN/ARR-10	SEARCHLIGHT
AN/ARR-11	AN/ARR-11	SEARCHLIGHT
AN/ARR-12	AN/ARR-12	SEARCHLIGHT
AN/ARR-13	AN/ARR-13	SEARCHLIGHT
AN/ARR-14	AN/ARR-14	SEARCHLIGHT
AN/ARR-15	AN/ARR-15	SEARCHLIGHT
AN/ARR-16	AN/ARR-16	SEARCHLIGHT
AN/ARR-17	AN/ARR-17	SEARCHLIGHT
AN/ARR-18	AN/ARR-18	SEARCHLIGHT
AN/ARR-19	AN/ARR-19	SEARCHLIGHT
AN/ARR-20	AN/ARR-20	SEARCHLIGHT
AN/ARR-21	AN/ARR-21	SEARCHLIGHT
AN/ARR-22	AN/ARR-22	SEARCHLIGHT
AN/ARR-23	AN/ARR-23	SEARCHLIGHT
AN/ARR-24	AN/ARR-24	SEARCHLIGHT
AN/ARR-25	AN/ARR-25	SEARCHLIGHT
AN/ARR-26	AN/ARR-26	SEARCHLIGHT
AN/ARR-27	AN/ARR-27	SEARCHLIGHT
AN/ARR-28	AN/ARR-28	SEARCHLIGHT
AN/ARR-29	AN/ARR-29	SEARCHLIGHT
AN/ARR-30	AN/ARR-30	SEARCHLIGHT
AN/ARR-31	AN/ARR-31	SEARCHLIGHT
AN/ARR-32	AN/ARR-32	SEARCHLIGHT
AN/ARR-33	AN/ARR-33	SEARCHLIGHT
AN/ARR-34	AN/ARR-34	SEARCHLIGHT
AN/ARR-35	AN/ARR-35	SEARCHLIGHT
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AN/ARR-42	AN/ARR-42	SEARCHLIGHT
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AN/ARR-44	AN/ARR-44	SEARCHLIGHT
AN/ARR-45	AN/ARR-45	SEARCHLIGHT
AN/ARR-46	AN/ARR-46	SEARCHLIGHT
AN/ARR-47	AN/ARR-47	SEARCHLIGHT
AN/ARR-48	AN/ARR-48	SEARCHLIGHT
AN/ARR-49	AN/ARR-49	SEARCHLIGHT
AN/ARR-50	AN/ARR-50	SEARCHLIGHT

RED FIGURES ARE PRELIMINARY; SUBJECT TO REVISION AFTER FLIGHT CHECK.



B-29
"SUPER FORTRESS"

MANUFACTURED BY JELL (ATLANTA) T-24; HOPKINS (EVANSTON) T-24; HOPKINS (EVANSTON) T-24; HOPKINS (EVANSTON) T-24; HOPKINS (EVANSTON) T-24

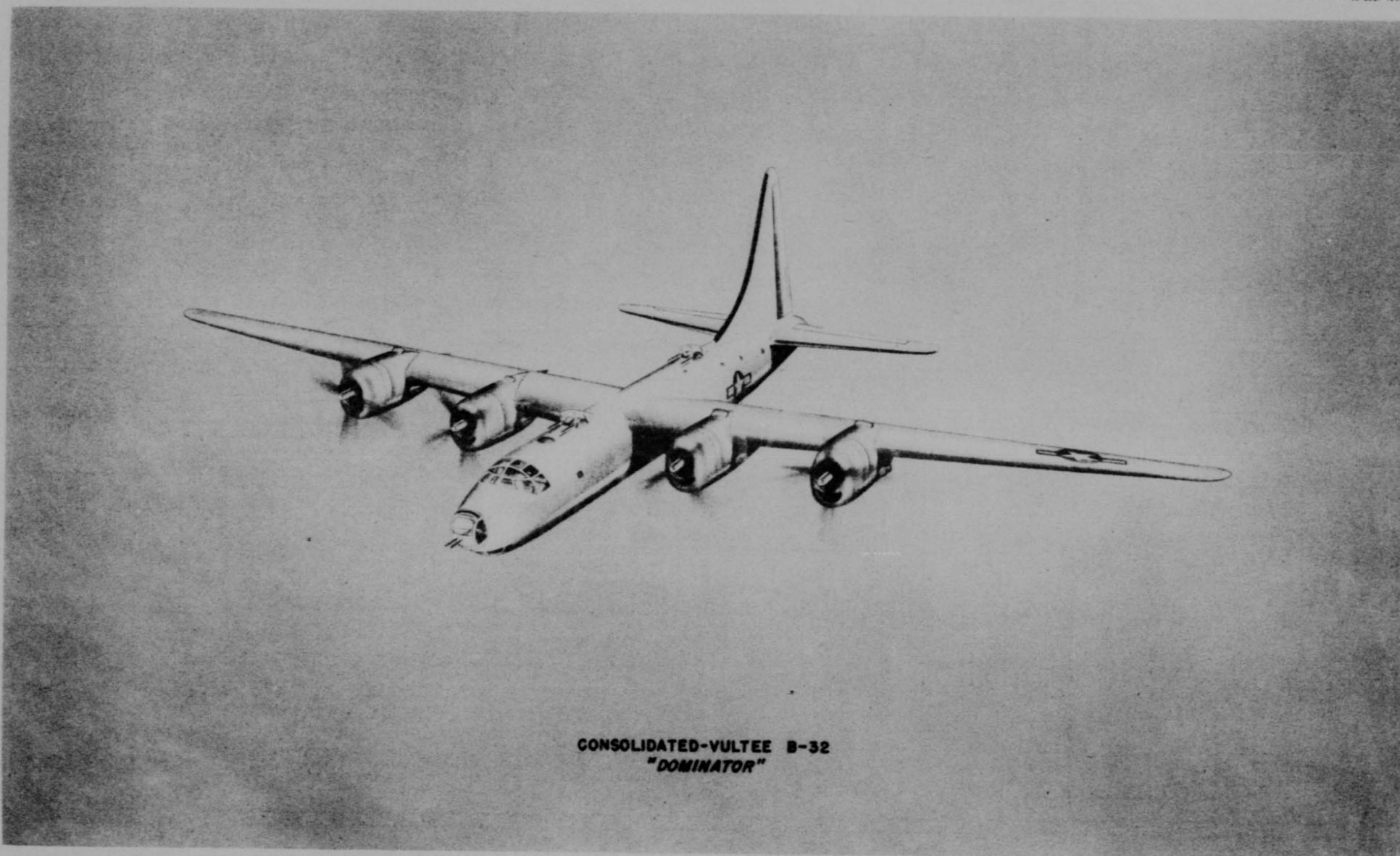
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PAGE 14
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CONSOLIDATED-VULTEE B-32
"DOMINATOR"

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TACTICAL PLANNING CHARACTERISTICS & PERFORMANCE CHART

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MODEL & BLOCK-NO.	ENGINE & PROP.		SIZE		WEIGHT		COMBAT CREW	FUEL & OIL		ARMAMENT			BOMBS			RADIO	REMARKS & REFERENCE	
	NUMBER	ENGINE MFR MODEL SUPERCHARGER PROP MFR TYPE	B.H.P./ALT.		S-SPAN L-LENGTH H-HEIGHT T-TREAD W-WING AREA	B-BASIC C-COMBAT W-WAR MAX M-MAX LAND		NOTES	FUEL TANKAGE		ROCKETS SIZE INSTALLATION			NO. AND SIZE				TYPICAL MODELS CARRIED
			T-TAKE OFF W-WAR EMERG M-MILITARY C-CONTINUOUS	T-TAKE OFF W-WAR EMERG M-MILITARY C-CONTINUOUS					TYPE OR LOCATION ALSO (MAX OIL)	NO AND CAPACITY ALSO (MAX FUEL)	NO GUNS AND SIZE	RDS PER GUN	LOCATION AND TYPE	INTERNAL	EXTERNAL	MAX. LOAD		
B-32-CF -1 thru -35		WRIGHT 5-2870-23 TURBO C18T155 187701A-4 EL. CONST. SPD. ELECTRIC	T 3200 S L W NONE M 3300/34000 C 3300/32500	S 143'0" L 42'1" H 33'0" T 26'0" W 1472 Sq.	B 62300 C 108000 W 114000 M -	NOTES	12	INBOARD-WINGS MIDBOARD-WINGS OUTER BAY * (302)	5-3400 4-2010 2-780 (8950)	25 25 25 25	345 100 400 150 1000	ROSE TURRET FRONT TOP TURRET REAR TOP TURRET LOW. HALL TURRET TAIL TURRET	4-9000 4-8700 8-1500 12-1000 W-500	20000	SEE NOTE (d)	B-32 TECH. ORDER 01-507-1 (5 MARCH 1945) (a) RELATING WITH B-32-25 CF, PROVISIONS FOR INSTALLATION OF 2 BOMB BAY TANKS: 4 BOMB BAY TANKS CAN BE CARRIED IF SPECIAL INSTALLATION IS MADE. (b) NORMAL ROUNDS LISTED: SPACE PROVISIONS FOR 600 RDS. PER GUN.		

MODEL & BLOCK-NO.	TAKE OFF & LANDING HARD SURFACE-NO WIND					HIGH SPEED & CLIMB CLEAN AIRPLANE AT NORMAL COMBAT WEIGHT										RANGE AND ENDURANCE STATUTE AIR MILES-NO WIND- NO ALLOWANCE FOR RESERVE										REMARKS		
	GROSS WEIGHT LB	TO CLEAR 50'		LANDING SPEED MPH	GROUND RUN		WEIGHT LB	STD. ALT. FT	WAR EMERG.		MIL. POWER		MAX. CONTIN. POWER		LOADING			MAX. CONTINUOUS POWER				MAX. CRUISE POWER					LONG RANGE	
		T.O. DIST. FT	LAND DIST. FT		T.O. DIST. FT	LAND DIST. FT			HIGH SPEED MPH	RATE OF CLIMB FT/MIN	HIGH SPEED MPH	RATE OF CLIMB FT/MIN	HIGH SPEED MPH	RATE OF CLIMB FT/MIN	TIME TO CLIMB MIN	TAKE OFF WEIGHT LB	BOMBS CARRIED	TOTAL FUEL U.S. GAL	at 10,000 FT.		at 25,000 FT.		at 10,000 FT.		at 25,000 FT.		at 10,000 FT.	
B-32-CF -1 thru -35	75000 80000 90000 100000 110000 120000	- 3900 4500 5700 7300 9300	2700 4100 4600 5600 - -	55 80 106 118 - -	2200 2500 2900 3200 -	100000 20000 30000 15000 10000 5000	30000 25000 20000 15000 10000 5000	30000 25000 20000 15000 10000 5000	357 324 312 296 281	250 700 850 950 1050	311 300 289 279 259	400 500 650 750 850	38.0 27.5 19.0 12.0 8.0	111500 119500 109000 121000	NONE 8000 8000 20000	8960 8960 5460 5460	- 1700 1350 1300	- 5.8 4.5 4.5	- - 1400 ABOVE PRAC. CEIL.	- - 4.8 4.8	2950 107.5 2100 2050	17.5 9.7 8.2 8.2	2700 2700 2000 ABOVE PRAC. CEIL.	9.7 7.8 7.8 ABOVE PRAC. CEIL.	3800 3300 2900 2400	21.0 18.0 16.0 13.0	FLIGHT TEST IN PROGRESS	

NOTES: (a) SCB-274N COMMAND SET
SCB-855 TFI
SCB-158B EMERGENCY TRANSMITTER
SCB-779 STARDOME INTERCOM
SC-224 BLIND LANDING
SA-140W-3 MAJOR BEACON
SC-151 RADIO FILTER
AN-140W-7 RADIO COMPASS
AN-170W-1 LOW ALT. RADIO ALTIMETER
AN-131W-58 LAD
AN-140W-15 ATE
AN-170W-4 LORAN
AN-140W-7 STATIC DISCHARGER
AN-112W-2 INTERPHONE
AN-112W-2 PROX. INDICATOR

RED FIGURES ARE PRELIMINARY: SUBJECT TO REVISION AFTER FLIGHT CHECK

* MAX. WEIGHT HAS BEEN TENTATIVELY RESTRICTED TO 110000 LB. PENDING COMPLETION OF STATIC TESTS.

MANUFACTURED BY CONSOLIDATED - VULTEE (SAN DIEGO) *COT & CONSOLIDATED - VULTEE (FT. WORTH) *CF*



B-32
"DOMINATOR"

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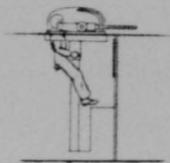
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AS OF:
25 JAN. 1945

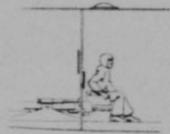
B-25 MITCHELL BOMBER



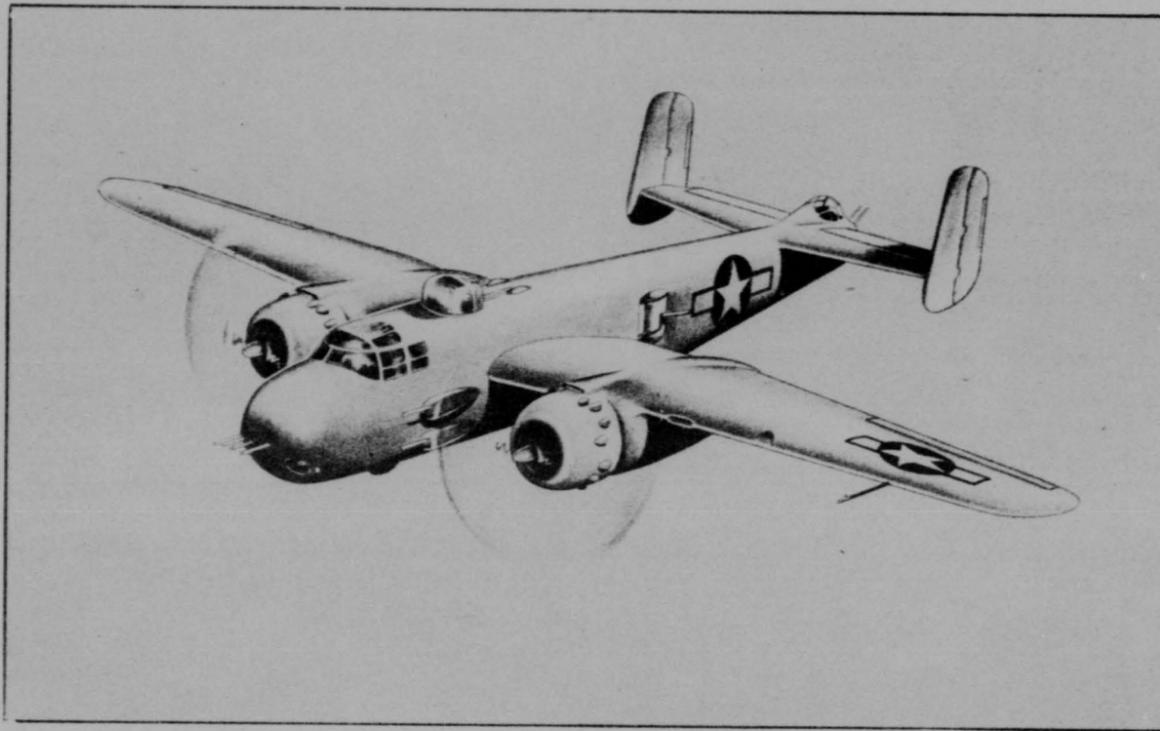
TAIL GUNNER
B-25H&J



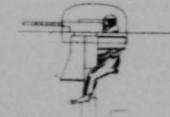
UPPER TURRET (REAR)
B-25C, D, & G



CANNONEER
B-25G&H



B-25H



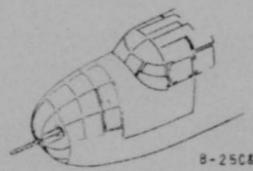
UPPER TURRET (FRONT)
B-25H&J



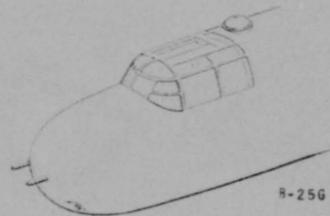
BOTTOM TURRET
B-25C, D, & G



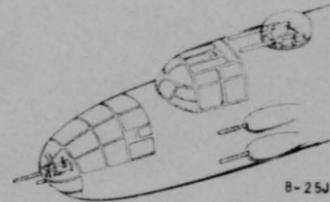
WAIST GUNNER
B-25H&J



B-25C&D



B-25G



B-25J

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TACTICAL PLANNING CHARACTERISTICS & PERFORMANCE CHART

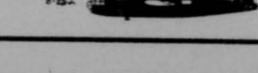
MODEL & BLOCK-NO.	ENGINE & PROP.		SIZE		WEIGHT		COMBAT CREW	FUEL & OIL		ARMAMENT			BOMBS		RADIO	REMARKS & REFERENCE		
	NUMBER	ENGINE MFR. MODEL SUPERCHARGER PROP. MFR. TYPE	B.H.P./ALT.	S-SPAN L-LENGTH H-HEIGHT W-WING AREA	B-BASIC C-COMBAT W-WAR MAX. M-MAX.LAND.	NOTES		TYPE OR LOCATION ALSO (MAX OIL)	NO AND CAPACITY ALSO (MAX FUEL)	NO. GUNS AND SIZE	RDS. PER GUN	LOCATION AND TYPE	NO. AND SIZE		MAX. LOAD		TYPICAL MODELS CARRIED (SEE "A")	
													INTERNAL	EXTERNAL				
B-25C-NA -1 thru -25	2	R-2600-13 or -29 2 SPD. SUPERCH. HAMILTON STD. 12" 7" DIA. -3 BL. F.F., HYDROMATIC	T W M C	1700 / S L	S 67' 6"	B 22000	5	MAIN - WINGS 2 x 184	1)	300	NOSE - FLEX.	1-2000	2-1600	2-1600	SCR-211 SCR-269G SCR-274H SCR-287A SCR-522 SCR-535 SCR-578 SCR-595 SCR-695 AN/ARN-7 RC-32 RC-36 RC-43B RC-103 RC-196 RC-198	REFER TO PAGE 4 FOR "FOREWORD." REFER TO PAGE 3 FOR SECURITY CLASSIFICATION. REFER TO PAGE 5 FOR DEFINITIONS. REFER TO T.O. LISTED FOR DETAILED PLANNING. B-25C & D TECH. ORDER 01-80GB-1. (REV. 30 AUG. 1944) B-25G TECH. ORDER 01-80GC-1. (REV. 10 DEC. 1944) B-25H TECH. ORDER 01-80GD-1. (REV. 10 FEB. 1945) B-25J TECH. ORDER 01-80GE-1. (REV. 25 MARCH 1945) (a) LENGTH B-25C is 53 FT. (b) B-25C-25 & D-20 THRU -35 HAVE SAME FUEL CAPACITY AS LISTED FOR B-25C. ONLY B-25D AIRPLANES CARRY 1 x 125 GAL. SIDE WAIST TANK. (c) B-25C-1 & D-1 CARRY ONLY 1-.30 CAL. FLEX. NOSE GUN OF 600 ROUNDS IN PLACE OF 2 NOSE GUNS. (d) ARMOR PIERCING BOMBS ON SOME MODELS ONLY. (e) BOMB BAY FUEL CONSISTS OF EITHER 1 x 215 GAL. SELF SEALING REMOVABLE TANK & 1 x 325 GAL. METAL DROPPABLE TANK OR 1M LIED, 1 x 585 GAL. DROPPABLE METAL TANK. (f) SOME B-25H-1 MODELS CARRY ONLY 2 NOSE PACKAGE GUNS. (g) THE 2000 LB. STATION IS DELETED ON 131st. B-25H-5 AND SUBSEQUENT, ALSO 1st. 150 B-25J-1's. T TORPEDO X EXTERNAL ★ ★ (8) 5" AR OR HYAR INSTALLED ON SOME SERVICE MODELS.		
B-25D-NC -1 thru -35				1850 / S L (-13) 1800 / S L (-29)	L 52' 8"	C 33500		AUX. - WINGS 6 - 304	1)	300	NOSE - FIXED	2-1500					1-2000 T	2-1600
B-25G-NA -1 thru -10				AS ABOVE	AS ABOVE	H 15' 9"		C 33500*	2)	400	UPPER TURRET	3-1000					6-325	8-250
B-25H-NA -1 thru -10				AS ABOVE	AS ABOVE	T 19' 4"		W 35000*	1 x 125	350	LOWER TURRET	6-500					8-250	8-100
B-25J-NC -1 thru -35				AS ABOVE	AS ABOVE	W 610 Sq. ft.		M 32200	1 x 585 (1884)	★★	(SOME MODELS)	8-250					8-100	12-100

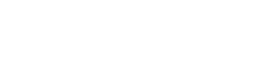
MODEL & BLOCK-NO.	TAKE OFF & LANDING HARD SURFACE-NO WIND					HIGH SPEED & CLIMB CLEAN AIRPLANE AT NORMAL COMBAT WEIGHT							RANGE AND ENDURANCE STATUTE AIR MILES-NO WIND- NO ALLOWANCE FOR RESERVE										REMARKS		
	GROSS WEIGHT LB.	TO CLEAR 50'		GROUND RUN		STD. ALT. FT.	WAR EMERG.			MIL. POWER				LOADING		MAX. CONTINUOUS POWER		MAX. CRUISE POWER		LONG RANGE					
		T.O. DIST. FT.	LAND DIST. FT.	LANDING SPEED MPH	T.O. DIST. FT.		LAND DIST. FT.	HIGH SPEED MPH	RATE OF CLIMB FT/MIN	HIGH SPEED MPH	RATE OF CLIMB FT/MIN	HIGH SPEED MPH	RATE OF CLIMB FT/MIN	TIME TO CLIMB MIN	TAKE OFF WEIGHT LB.	BOMBS CARRIED LB.	TOTAL FUEL U.S. GAL.	RANGE MILES	ENDUR. HOURS	RANGE MILES	ENDUR. HOURS	RANGE MILES		ENDUR. HOURS	RANGE MILES
B-25C-NA -1 thru -25	24000	3300	105	1500	1700	20000	265	285	260	280	31.7	34000	NONE	1584	1375	5.2	1450 ^z	5.7 ^z	2250	10.3	2175 ^z	10.0 ^z	2650	12.6	
B-25D-NC -1 thru -35	27000	2900	3500	110	1300	15000	284	285	270	280	31.7	34000	NONE	1559	1275	4.7	1300 ^z	5.2 ^z	2200	9.4	1950 ^z	9.0 ^z	2500	11.8	
B-25G-NA -1 thru -10	30000	4000	3800	112	3200	10000	282	275	274	830	9.5	33200	3200	974	775	2.9	800	3.3	1350	5.7	1175	5.3	1525	7.3	
B-25H-NA -1 thru -10	32000	4800	-	-	3900	5000	282	1225	274	1150	4.3	32200	2000 T	974	750	3.0	ABOVE PRAC- TICAL CEILING	1250	5.5	ABOVE PRAC- TICAL CEILING	1225	6.1	1350	7.1	
B-25J-NC -1 thru -35	35000	6000	-	-	4800	5 L	284	1375	257	1200	0.0	35000*	5200	974	725	2.9	1100	5.0	1100	5.0	ABOVE PRAC- TICAL CEILING	1225	6.1	1350	7.1

NOTES: (A) SCR-211 FREQUENCY METER
SCR-269G RADIO COMPASS
SCR-274H COMMAND SET
SCR-287A LIAISON SET
SCR-522 COMMAND SET
SCR-535 SET
SCR-578 EMERGENCY TRANSMITTER
SCR-595 SET
AN/ARN-7 RADIO COMPASS
RC-32 FILTER EQUIPMENT
RC-36 INTERPHONE AMPLIFIER
RC-43B MARKER BEACON
RC-103 LOCALIZER
RC-196 CONTACTOR UNIT
RC-198 FILTER EQUIPMENT

RED FIGURES ARE PRELIMINARY: SUBJECT TO REVISION AFTER FLIGHT CHECK.







B-25 "MITCHELL"

MANUFACTURED BY NORTH AMERICAN (INGLEWOOD) "NA" & (KANSAS CITY) "NC"

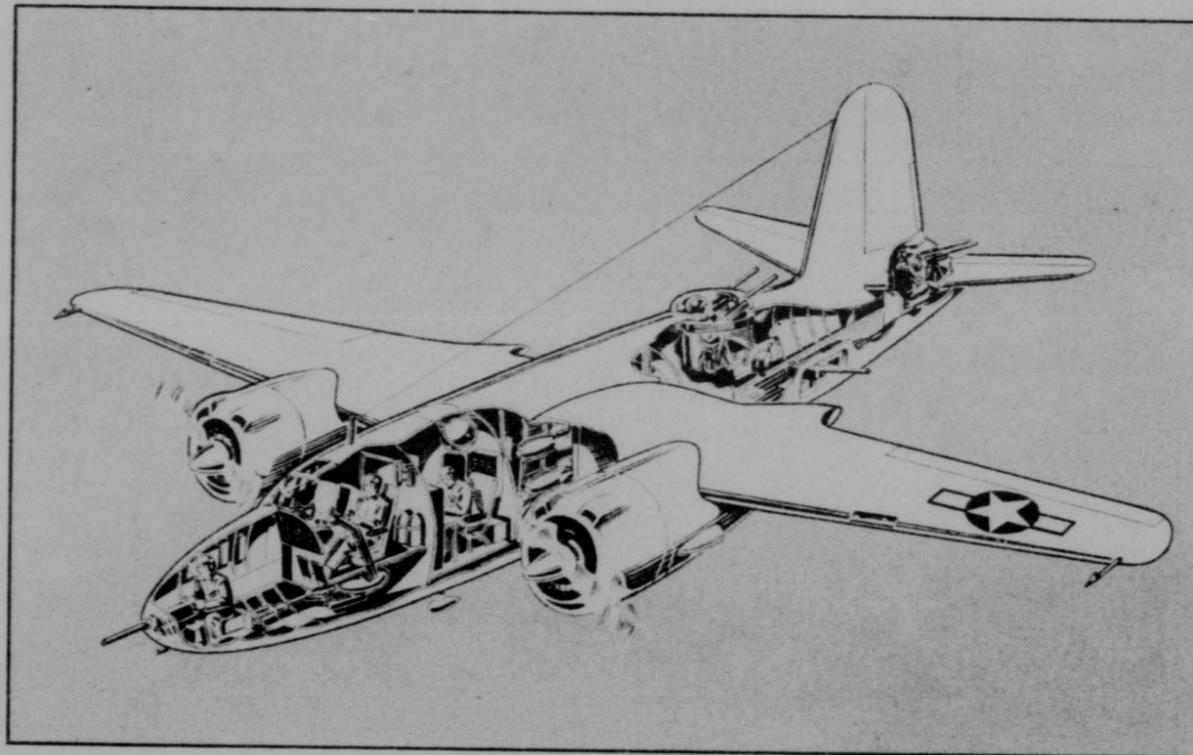
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25 JAN. 1945

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B-26 "MARAUDER"

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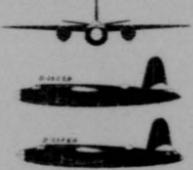
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TACTICAL PLANNING CHARACTERISTICS & PERFORMANCE CHART																		
MODEL & BLOCK-NO.	ENGINE & PROP.		SIZE		WEIGHT		COMBAT CREW	FUEL & OIL		ARMAMENT			BOMBS		RADIO	REMARKS & REFERENCE		
	NUMBER	ENGINE MFR. MODEL SUPERCHARGER PROP. MFR. TYPE	B.H.P./ALT.		S-SPAN L-LENGTH H-HEIGHT T-TREAD W-WING AREA	B-BASIC C-COMBAT W-WAR MAX. M-MAX. LAND.		M NOTES	FUEL TANKAGE		NO. GUNS AND SIZE			NO. AND SIZE		MAX. LOAD	TYPICAL MODELS CARRIED (SEE A-1)	REFER TO PAGE 4 FOR "FOREWORD" REFER TO PAGE 3 FOR SECURITY CLASSIFICATION. REFER TO PAGE 5 FOR DEFINITIONS. REFER TO T.O. LISTED FOR DETAILED PLANNING.
			T-TAKE OFF W-WAR EMERG M-MILITARY C-CONTINUOUS	W-WAR EMERG M-MILITARY C-CONTINUOUS					TYPE OR LOCATION ALSO (MAX. OIL)	NO. AND CAPACITY ALSO (MAX. FUEL)	NO. GUNS AND SIZE	RDS. PER GUN	LOCATION AND TYPE	INTERNAL	EXTERNAL			
B-26B-MA -10 thru -20	2	PRATT-WHITNEY R-2600-43 2 SPD. SUPERCHARGER CURTISS 12' 6" DIA. - 4 BL. P.F., ELECTRIC	T 1920/ S L W NONE M 1920/ 3200 C 1490/ 14300 1550/ 7100 1960/ 14600	S 71' 0" L 58' 3" H 21' 6" T 22' 0" W 658 Sq. ft.	B 25300 C 37000 W 38200* M 34000	FLIGHT CHARACTERISTICS CRITICAL ABOVE 38200 LB.	6	MAIN - WINGS AUX. - WINGS BOMB BAY (FOR A REAR) (83)	2 x 360 2 x 121 4 x 250 (1962)	1) 200 2) 270 2) 240 2) 250 2) 200 2) 400 2) 200	NOSE - FIXED NOSE - FLEX. SIDE MAIST - FLEX. U. FUSEL. - PACKAGE L. FUSEL. - PACKAGE UPPER TURRET *TAIL-POWER MOUNT	2-2000 2-1600 4-1000 8- 500 16- 250 30- 100	1-2000T TORPEDO	2-1600 1-2000T 5200	SCR-211 SCR-269G SCR-274 SCR-287A SCR-522 SCR-535 SCR-578 SCR-595 SCR-695 RC-36 RC-439 RC-103 RC-198 RC-32	B-26B & C TECH. ORDER 01-35EB-1. (REV. 30 APRIL 1945) B-26F & G TECH. ORDER 01-35EC-1. (REV. 15 OCT. 1944)		
B-26C-MO -5 thru -25	2	AS ABOVE	AS ABOVE	AS ABOVE	AS ABOVE		6	MAIN - WINGS AUX. - WINGS BOMB BAY (FOR ONLY) (83)	2 x 360 2 x 121 2 x 250 (1462)	AS ABOVE EXCEPT FIXED NOSE GUN DELETED ON LATEST MODELS	2-2000 2-1600 4-1000 6- 500 10- 250 20- 100	AS ABOVE	4000			(a) EARLY MODELS HAVE FLEXIBLE TAIL MOUNT WITH 1500 HOURS PER GUN INSTEAD OF POWER MOUNT. (b) EARLY MODELS HAVE SAME NUMBER OF BOMB STATIONS AS LISTED ABOVE; DECREASE IN BOMB STATIONS AS NOTED DUE TO DELETION OF 2 REAR BOMB BAYS. (c) FUEL CAPACITY 20 GAL. PER TANK GREATER DUE TO INCREASE OF 3 1/2" ANGLE INCIDENCE OF WING.		
B-26F-MA -1, -2 & -5	2	AS ABOVE	AS ABOVE	AS ABOVE	AS ABOVE		6	MAIN - WINGS AUX. - 1 & 25 BOMB BAY (FOR ONLY) (83)	2 x 380 2 x 121 2 x 250 (1502)	300 250 275 225 400 900	NOSE - FLEX. SIDE MAIST - FLEX. U. FUSEL. - PACKAGE L. FUSEL. - PACKAGE UPPER TURRET TAIL-POWER MOUNT	2-2000 2-1600 4-1000 6- 500 10- 250 20- 100	NONE	4000				

MODEL & BLOCK-NO.	TAKE OFF & LANDING HARD SURFACE-NO WIND					HIGH SPEED & CLIMB CLEAN AIRPLANE AT NORMAL COMBAT WEIGHT							RANGE AND ENDURANCE STATUTE AIR MILES-NO WIND-NO ALLOWANCE FOR RESERVE										REMARKS		
	GROSS WEIGHT LB.	TO CLEAR 50'		GROUND RUN		WEIGHT LB.	STD. ALT. FT.	WAR EMERG.		MIL. POWER		MAX. CONTIN. POWER			LOADING			MAX. CONTINUOUS POWER		MAX. CRUISE POWER		LONG RANGE			
		T.O. DIST. FT.	LAND DIST. FT.	LANDING SPEED MPH	T.O. DIST. FT.			LAND DIST. FT.	HIGH SPEED MPH	RATE OF CLIMB FT. MIN.	HIGH SPEED MPH	RATE OF CLIMB FT. MIN.	HIGH SPEED MPH	RATE OF CLIMB FT. MIN.	TIME TO CLIMB MIN.	TAKE OFF WEIGHT LB.	BOMBS PARS.	TOTAL FUEL U.S. GAL.	at 10000FT. RANGE ENOUR.	at 20000FT. RANGE ENOUR.	at 10000FT. RANGE ENOUR.	at 20000FT. RANGE ENOUR.		at 10000FT. RANGE ENOUR.	at 10000FT. RANGE ENOUR.
B-26B-MA -10 thru -20	28000	-	3400	135	-	1900	20000	-	-	-	-	-	38200*	NONE	1962	1425	5.2	1600	7.8	2375	10.6	2050	9.1	2600	13.3
B-26C-MO -5 thru -25	32000	3000	3900	136	1900	2000	15000	282	500	278	490	23.0	37800	NONE	1462	1900	3.8	1075	4.9	1725	7.8	1425	6.5	1825	9.2
B-26B-MA -25 thru -55	34000	3500	4200	138	2300	2200	10000	NONE	-	-	-	-	38200*	1500	1462	1000	3.8	1075	4.9	1700	7.8	1400	6.5	1800	9.2
B-26C-MO -30 & -45	36000	4000	-	-	2700	-	5000	285	1000	269	900	5.0	36300	2000	962	700	2.5	675	3.0	1100	4.9	800	4.0	1200	6.2
B-26F-MA -1, -2 & -5	38200	4400	-	-	3000	-	37000	270	1200	254	875	0.0	36100	4000	962	650	2.5	675	3.1	1050	4.9	800	4.1	1075	5.4
B-26C-MO -30 & -45	AS ABOVE	AS ABOVE	AS ABOVE	AS ABOVE	AS ABOVE	AS ABOVE	AS ABOVE	AS ABOVE	AS ABOVE	AS ABOVE	AS ABOVE	AS ABOVE	37800	NONE	1462	1000	3.8	1075	7.8	1725	7.8	1425	6.5	1825	9.2
B-26C-MO -30 & -45	36300	2000	-	-	-	-	5000	962	700	2.5	675	4.9	1100	4.9	800	4.0	1200	6.2			800	4.1	1075	5.4	
B-26F-MA -1, -2 & -5	28000	-	-	122	-	-	20000	255 ^d	-	258	-	-	38000	NONE	1502	1050	3.9	1175	5.1	1775	8.0	1525	6.7	1875	9.6
B-26C-MO -30 & -45	32000	2500	-	124	1800	-	15000	274	-	274	350	31.6	38000	NONE	1502	1050	3.9	1175	5.1	1775	8.0	1525	6.7	1875	9.6
B-26C-MO -30 & -45	34000	3000	-	125	2200	-	10000	277	-	259	830	8.1	38200*	2000	1002	725	2.6	700	3.1	1190	5.1	900	4.2	1250	6.5
B-26G-MA -1 thru -25	36000	3500	-	-	2500	-	9000	281	1000	242	600	0.0	38200*	4000	1002	675	2.6	700	3.2	1100	5.1	900	4.3	1125	5.7
B-26G-MA -1 thru -25	38200	4200	-	-	2800	-	5000	281	1000	242	600	0.0	38200*	4000	1002	675	2.6	700	3.2	1100	5.1	900	4.3	1125	5.7

NOTES: (A) SCR-211 FREQUENCY METER
SCR-269G RADIO COMPASS
SCR-274N COMMAND SET
SCR-287A LIAISON SET
SCR-522 COMMAND SET
SCR-532 SET
SCR-578 FREQUENCY METER SET (142)
SCR-595 SET (142)
SCR-695 SET (142)
RC-32 FILTER EQUIPMENT
RC-38 INTERPHONE AMPLIFIER
RC-439 WARNER BEACON
RC-103 LOCALIZER
RC-140 FILTER EQUIPMENT

RED FIGURES ARE PRELIMINARY: SUBJECT TO REVISION AFTER FLIGHT CHECK

B-26

"MARAUDER"

MANUFACTURED BY MARTIN (BALTIMORE) "M" & (OHIO) "M"

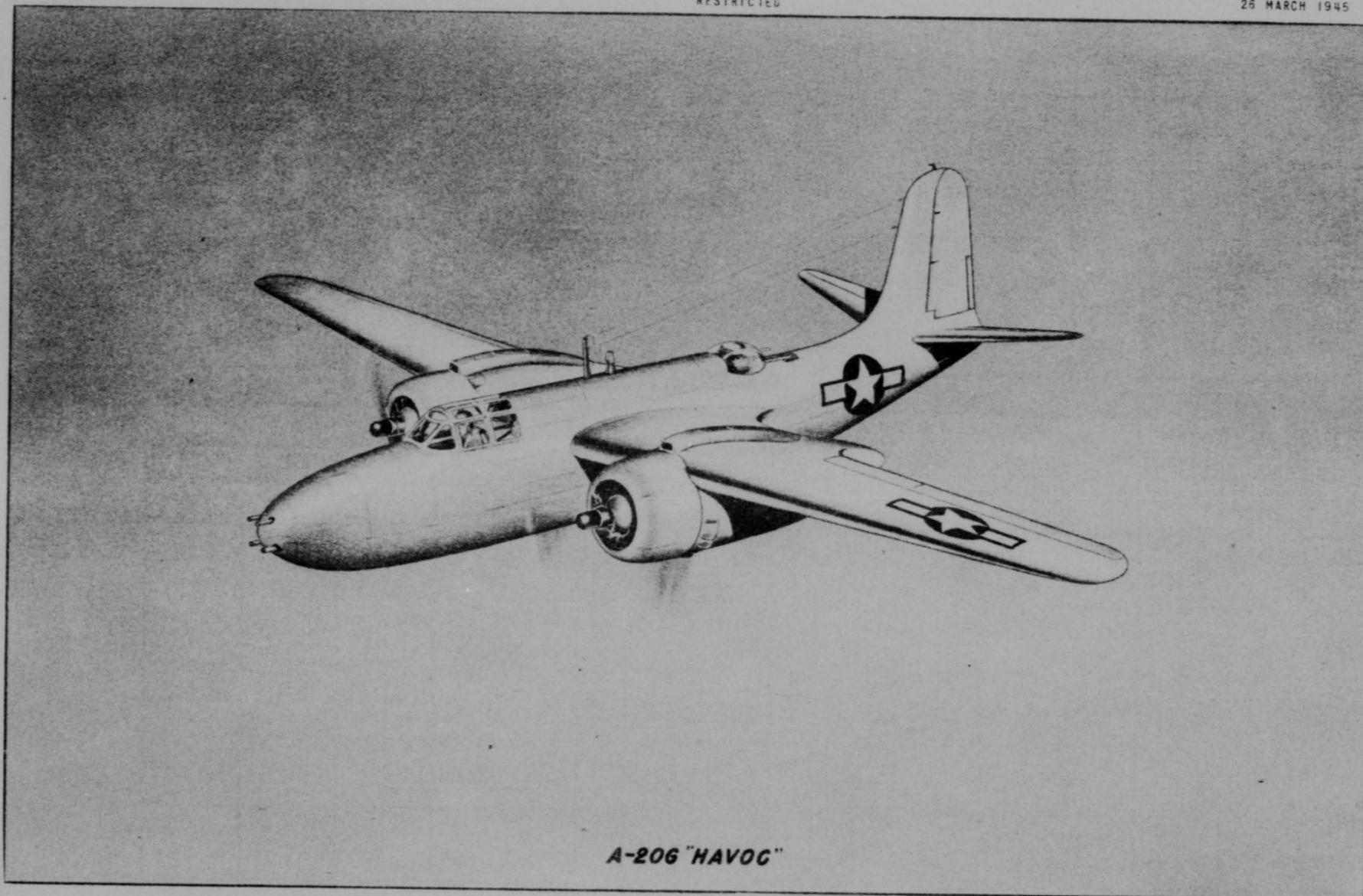
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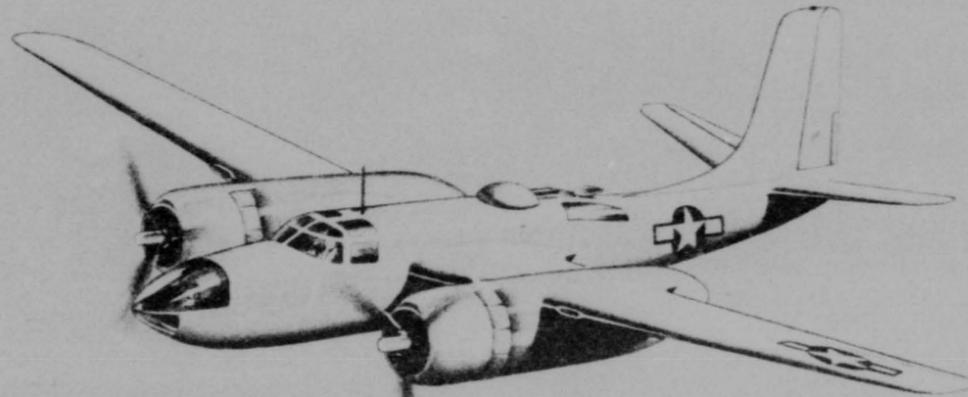
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AS OF:
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A-26C

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TACTICAL PLANNING CHARACTERISTICS & PERFORMANCE CHART																		
MODEL & BLOCK-NO.	ENGINE & PROP.		SIZE		WEIGHT		COMBAT CREW	FUEL & OIL		ARMAMENT			BOMBS			RADIO	REMARKS & REFERENCE	
	NUMBER	ENGINE MFR. MODEL SUPERCHARGER PROP. MFR. TYPE	B.H.P./ALT. T-TAKE OFF W-WAR EMERG. M-MILITARY C-CONTINUOUS	S-SPAN L-LENGTH H-HEIGHT T-TREAD W-WING AREA	B-BASIC C-COMBAT W-WAR MAX. M-MAX.LAND.	* NOTES		FUEL TANKAGE		(NO. ROCKETS-SIZE-INSTALLATION)			NO. AND SIZE			TYPICAL MODELS CARRIED (SEE 4)		
								TYPE OR LOCATION ALSO (MAX. OIL)	NO. AND CAPACITY ALSO (MAX. FUEL)	NO. GUNS AND SIZE	RDS PER GUN	LOCATION AND TYPE	INTERNAL	EXTERNAL	MAX. LOAD			
A-26B-DL -1 & -5	2	PRATT-WHITNEY R-2800-27 SINGLE STD.-2 SPEED HAMILTON STD. 12"7" DIA.-3 BL. F.F., HYDROMATIC	T-2000/ S L 2340/ 5 L W-1800/ 1500 M-1800/ 1500 C-1450/ 14000	S 70' L 50' H 18' 6" T 19' 6" W 540 Sq. Ft.	B 23275 C 35000 W 42300* M 31000	2	MAIN - WINGS 2 x 300 AUX. - WINGS 2 x 100 BOMB BAY(S.S.) 1 x 125 B.L.BAY (HERRY) 1 x 575 (68)	2	14(5") ZERO LENGTH RAIL ROCKETS	400 NOSE 500 UPPER TURRET 500 LOWER TURRET	4-1000 2-2000T 4-1000 4-500 4-250 15-100	4-1000 4-500 4-250 6000	SCR-269 SCR-274 SCR-522 SCR-595 SCR-695	REFER TO PAGE 4 FOR "FOREWORD" REFER TO PAGE 3 FOR SECURITY CLASSIFICATION. REFER TO PAGE 5 FOR DEFINITIONS. REFER TO T.O. LISTED FOR DETAILED PLANNING.				
A-26B-DT -5															A-26B-DL -10 thru -45	A-26B-DT -10 thru -25	A-26B-DL -50 thru -70	A-26D-5-DL (See g)
COLUMN	1	2	3	4	5	6	7	8	9	10	11	12	13	14	15	16	17	18

MODEL & BLOCK-NO.	TAKE OFF & LANDING HARD SURFACE-NO WIND					HIGH SPEED & CLIMB CLEAN AIRPLANE AT NORMAL COMBAT WEIGHT										RANGE AND ENDURANCE STATUTE AIR MILES-NO WIND-NO ALLOWANCE FOR RESERVE										REMARKS		
	GROSS WEIGHT LB	TO CLEAR 50'		LANDING SPEED MPH	GROUND RUN		WEIGHT LB	STD. ALT. FT.	WAR EMERG.		MIL. POWER		MAX. CONTIN. POWER		LOADING		MAX. CONTINUOUS POWER				LONG RANGE							
		T.O. DIST. FT.	LAND DIST. FT.		T.O. DIST. FT.	LAND DIST. FT.			HIGH SPEED MPH	RATE OF CLIMB FT. MIN.	HIGH RATE MPH	RATE OF CLIMB FT. MIN.	HIGH SPEED MPH	RATE OF CLIMB FT. MIN.	HIGH SPEED MPH	RATE OF CLIMB FT. MIN.	TIME TO CLIMB MIN.	TAKE OFF WEIGHT LB	BOMBS CARGO PASS.	TOTAL FUEL U.S. GAL.	at 10000 FT.		at 20000 FT.		at 10000 FT.		at 20000 FT.	
A-26 SERIES	25000	-	3400	88	-	1900	20000 ^a	15000 ^b	361	NO	335	350	330	300	22.5	37200	400E	2035	1450	4.8	1975 ^c	6.5 ^d	2700	9.7	2600 ^e	9.1 ^f	3450	16.1
	29000	4300	4000	195	2600	2200	10000	10000	373	DATA	352	1150	325	1050	6.0	31400	400E	925	850	2.0	900	2.8	1250	4.3	1375	4.7	1625	7.6
	32000	5500	-	-	3400	-	5000	5000	351	-	355	1550	325	1275	3.5	34400	3000	925	850	2.0	850	2.8	1250	4.4	1300	4.9	1525	6.8
	35000	7200	-	-	4400	-	5000	5000	351	-	335	2025	305	1450	0.0	34000	4000	800	550	1.7	725	2.4	1050	3.7	1075	4.1	1275	5.7
	38000	9100	-	-	5600	-	5000	5000	360	-	335	2025	305	1450	0.0	34000	4000	800	550	1.7	725	2.4	1050	3.7	1075	4.1	1275	5.7

NOTES: FIVE OR MORE OF FOLLOWING:

- (A) SCR-269 RADIO COMPASS
- SCR-274 COMMAND SET
- SCR-522 COMMAND SET
- SCR-595 IFF
- SCR-695 IFF
- WH-26 RADIO COMPASS
- RC-32 FILTER EQUIPMENT
- RC-36 INTERPHONE AMP.
- RC-192 MARKER BEACON
- RC-198 FILTER EQUIPMENT
- AN/AIC-2 INTERPHONE
- AN/APN-4 or 9 LORAN
- AN/ARC-3 COMMAND (WH)
- AN/ARN-7 RADIO COMPASS
- AN/ARC-8 LIAISON SET
- AN/ASA-3 STATIC DISCHARGE
- ROM RADIO COUNTER MEASURE
- GEE BRITISH LORAN
- AN/APD-9 JAMMING TRANSMITTER

RED FIGURES ARE PRELIMINARY; SUBJECT TO REVISION AFTER FLIGHT CHECK.



A-26
"INVADER"

MANUFACTURED BY DOUGLAS (LONG BEACH) "DL" & (TULSA) "DT"

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PAGE 24
AS OF:
29 JUNE 1945

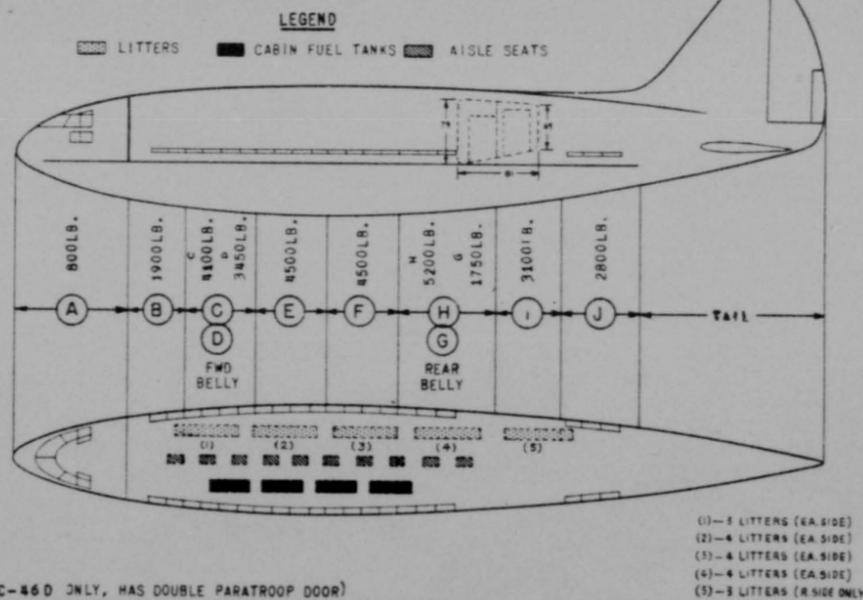
LOADING THE CURTISS "COMMANDO"

ITEM	WEIGHT	STATION
JEEP  	3100 LB.	(C) THRU (H)
105 MM GUN & CAISSON  	GUN 4300LB. CAISSON 4700LB.	(E) THRU (H)
LITTERS  	3 TIER LITTER 50LB. OXYGEN BOTTLE 138LB.	REAR - (B) THRU FORE - (J)
75 MM HOWITZER & CAISSON  	HOWITZER 2100LB. CAISSON 5000LB.	(C) THRU (H)
37 MM GUN & MOTOR CHASSIS  	5500LB.	(C) THRU (H)
75 MM GUN & CAISSON  	GUN 3400LB. CAISSON 5000LB.	(C) THRU (H)
LARGE ENGINE & MOUNT  	4900LB.	(C) THRU (I)

WEIGHT LEGEND

ATTENDANT OR PASSENGER — 200LB.
TROOP AND EQUIPMENT — 240LB.
LITTER (INCL. PATIENT) — 250LB.
PARACAN — 250LB.
PARATROOPER — 260LB.

LOADING STATIONS OF THE C-46



ANY COMBINATION OF CARGO CAN BE CARRIED BUT MUST SATISFY FOUR REQUIREMENTS: (1) Total weight. (2) Space limitations. (3) Center of gravity. (4) Floor loading. (Typical items are illustrated).

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TACTICAL PLANNING CHARACTERISTICS & PERFORMANCE CHART

PAGE 25
AS OF 30 AUGUST 1945

MODEL & BLOCK-NO.	ENGINE & PROP.		SIZE		WEIGHT		COMBAT CREW	FUEL & OIL		CARGO OR PASSENGERS			RADIO	REMARKS & REFERENCE
	NUMBER	ENGINE MFR. MODEL SUPERCHARGER PROP MFR. TYPE	B.H.P./ALT.	S-SPAN L-LENGTH H-HEIGHT T-TREAD W-WING AREA	B-BASIC C-COMBAT W-WAR MAX T-TREAD M-MAX LAND	NOTES		FUEL TANKAGE		STATION CAPACITY MAX LOAD	STATION PLACES MAX LOAD	TYPICAL MODELS CARRIED		
								TYPE OR LOCATION ALSO (MAX OIL)	NO AND CAPACITY ALSO (MAX FUEL)					
C-46A-5 thru -51 (C) C-46A-1, -5, -15, -40 (C) C-46A-1 (CS) C-46D-1 thru -20 (C)	2	PRATT-WHITNEY 3-2400-11 or -26 2 SPEED-SINGLE STG. CURTISS 19' 6" DIA. - 4 BL. F.F., ELECTRIC	T 3000/5 L W NONE M 2200/1500 C 1800/1200 1450/1300	S 108' 0" L 75' 4" H 22' 0" T 20' 0" W 1360 Sq. Ft.	B 31700 C 49600** W 56000 M 49500	PERMISSIBLE ONLY WITH ALL WING TANKS FULL.	WING - FUSE 2 x 294 WING - CENTER 2 x 256 WING - REAR 2 x 171 FUSEL. - TENDON 9 x 100 (50 x 40) (2302)	CAP (M) (M) 10000 FELLY (F) (F) 2700 FELLY (A) (A) 1600	STATION CAPACITY MAX LOAD 10000 2700 1600	STATION PLACES MAX LOAD 50 50 50 ATTENDANTS 4	12000 9000	SCR-211, 289, 274R, 287, 518, 322, 578, 195, 30-36, 43, 103, 199, MN-25, AN/AIC-2, 3, AN/ARA-10, AN/ARC-3, 8, 9, AN/ARN-5, 7, 11, AN/ASX-1, AN/ASA-1, AN/SPW-1, 2, 4, 9	REFER TO PAGE 4 FOR "FOREWORD" REFER TO PAGE 3 FOR SECURITY CLASSIFICATION. REFER TO PAGE 5 FOR DEFINITIONS REFER TO T.O. LISTED FOR DETAILED PLANNING.	
C-46F-CU -1 thru -5	2	PRATT-WHITNEY R-2600-75 2 SPEED-SINGLE STG. HAMILTON STD. 15' 1" DIA. - 3 BL. F.F., HYDROMATIC	T 45 ABOVE W AS ABOVE M AS ABOVE C AS ABOVE	S 45 ABOVE L AS ABOVE H AS ABOVE T AS ABOVE W AS ABOVE	B 31700 C 49600** W 56000 M 49500	AS ABOVE	AS ABOVE	AS ABOVE	AS ABOVE	AS ABOVE	AS ABOVE	AS ABOVE	AS ABOVE	(*) LATER C-46D'S HAVE EXTRA PARATROOP DOOR. ** PROVIDED T.O. 01-25LA-59 DATED 20 JULY 44 IS COMPLIED WITH.

MODEL & BLOCK-NO.	TAKE OFF & LANDING HARD SURFACE - NO WIND					HIGH SPEED & CLIMB CLEAN AIRPLANE AT NORMAL COMBAT WEIGHT										RANGE AND ENDURANCE STATUTE AIR MILES - NO WIND - NO ALLOWANCE FOR RESERVE										REMARKS	
	GROSS WEIGHT LB	TO CLEAR 50'		LANDING SPEED MPH	GROUND RUN		STD ALT. FT	WAR EMERG. HIGH SPEED MPH	MIL POWER RATE OF CLIMB FT/MIN	MAX CONTIN. POWER HIGH SPEED MPH	TIME TO CLIMB MIN	LOADING			MAX CONTIN.		LONG RANGE CRUISE										
		T.O. DIST. FT	LAND DIST. FT		T.O. DIST. FT	LAND DIST. FT						WEIGHT LB	BOMBS CARGO PASS. LB	TOTAL FUEL U.S. GAL	at 10000 FT.	at 5000 FT.	at 10,000 FT.				at 15,000 FT.				at 20,000 FT.		
		RANGE	ENDUR.		RANGE	ENDUR.						RANGE	ENDUR.	RANGE	ENDUR.	RANGE	ENDUR.	RANGE	ENDUR.	RANGE	ENDUR.	RANGE	ENDUR.	RANGE	ENDUR.		
C-46A-5 thru -51 (C) C-46A-1, -5, -15, -40 (C) C-46A-1 (CS) C-46D-1 thru -20 (C)	34000 38000 40000 45000 50000	- 2100 2300 2900 3500	2300 2500 2700 2900 -	82 88 94 96 -	- 1500 1400 1500 1600 -	15000 10000 9000 8000 7000	270 284 282 245	525 700 825 900	246 240 234 220	350 440 525 615	29.2 17.4 7.9 0.0	10000 40000 40000 50000	1400 4000 5000 10000	2302 1402 1040 825	1400 900 650 525	5.4 3.8 2.7 2.7	2975 1800 1300 1225	17.2 10.3 7.3 7.0	1475 1750 1250 1150	15.4 9.3 6.7 6.2	2925 1700 1200 1000	14.5 10.2 7.3 6.5	2475 1500 1030 950	13.8 9.0 6.3 5.6	(b) FIRST 4 HRS. OF FLIGHT AT 15000 FT. ALTITUDE. (c) FIRST 3 HRS. OF FLIGHT AT 10000 FT. ALTITUDE.		
C-46F-CU -1 thru -5	-	AS ABOVE	AS ABOVE	AS ABOVE	AS ABOVE	15000 10000 5000 5 L	278 270 276 252	525 700 850 950	255 243 239 222	325 450 675 650	25.8 16.5 7.6 0.0	10000 40000 40000 50000	1200 3800 5800 5800	2202 1402 1050 825	1425 850 650 525	5.0 3.0 2.7 2.7	3050 1850 1350 1200	18.7 11.8 8.4 7.7	2025 1850 1350 1200	17.8 11.0 7.9 7.0	2910 1825 1200 1150	16.8 10.3 7.3 6.5	2650 1500 1150 1000	14.3 9.0 6.3 5.8			

NOTES: TWELVE OR MORE OF FOLLOWING:

EARLY TYPES	NAME	LATER TYPES
SCR-211	FREQUENCY METER	AN/ARC-7 53
AN/ARC-1	COMM. SET	AN/ARC-7 53
AN/ARC-2	COMM. SET	AN/ARC-7 53
AN/ARC-3	COMM. SET	AN/ARC-7 53
AN/ARC-4	COMM. SET	AN/ARC-7 53
AN/ARC-5	COMM. SET	AN/ARC-7 53
AN/ARC-6	COMM. SET	AN/ARC-7 53
AN/ARC-7	COMM. SET	AN/ARC-7 53
AN/ARC-8	COMM. SET	AN/ARC-7 53
AN/ARC-9	COMM. SET	AN/ARC-7 53
AN/ARC-10	COMM. SET	AN/ARC-7 53
AN/ARC-11	COMM. SET	AN/ARC-7 53
AN/ARC-12	COMM. SET	AN/ARC-7 53
AN/ARC-13	COMM. SET	AN/ARC-7 53
AN/ARC-14	COMM. SET	AN/ARC-7 53
AN/ARC-15	COMM. SET	AN/ARC-7 53
AN/ARC-16	COMM. SET	AN/ARC-7 53
AN/ARC-17	COMM. SET	AN/ARC-7 53
AN/ARC-18	COMM. SET	AN/ARC-7 53
AN/ARC-19	COMM. SET	AN/ARC-7 53
AN/ARC-20	COMM. SET	AN/ARC-7 53
AN/ARC-21	COMM. SET	AN/ARC-7 53
AN/ARC-22	COMM. SET	AN/ARC-7 53
AN/ARC-23	COMM. SET	AN/ARC-7 53
AN/ARC-24	COMM. SET	AN/ARC-7 53
AN/ARC-25	COMM. SET	AN/ARC-7 53
AN/ARC-26	COMM. SET	AN/ARC-7 53
AN/ARC-27	COMM. SET	AN/ARC-7 53
AN/ARC-28	COMM. SET	AN/ARC-7 53
AN/ARC-29	COMM. SET	AN/ARC-7 53
AN/ARC-30	COMM. SET	AN/ARC-7 53
AN/ARC-31	COMM. SET	AN/ARC-7 53
AN/ARC-32	COMM. SET	AN/ARC-7 53
AN/ARC-33	COMM. SET	AN/ARC-7 53
AN/ARC-34	COMM. SET	AN/ARC-7 53
AN/ARC-35	COMM. SET	AN/ARC-7 53
AN/ARC-36	COMM. SET	AN/ARC-7 53
AN/ARC-37	COMM. SET	AN/ARC-7 53
AN/ARC-38	COMM. SET	AN/ARC-7 53
AN/ARC-39	COMM. SET	AN/ARC-7 53
AN/ARC-40	COMM. SET	AN/ARC-7 53
AN/ARC-41	COMM. SET	AN/ARC-7 53
AN/ARC-42	COMM. SET	AN/ARC-7 53
AN/ARC-43	COMM. SET	AN/ARC-7 53
AN/ARC-44	COMM. SET	AN/ARC-7 53
AN/ARC-45	COMM. SET	AN/ARC-7 53
AN/ARC-46	COMM. SET	AN/ARC-7 53
AN/ARC-47	COMM. SET	AN/ARC-7 53
AN/ARC-48	COMM. SET	AN/ARC-7 53

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28 JULY 1945

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LOADING THE DOUGLAS "SKYTRAIN"

ITEM	WEIGHT	STATION
JEEP 	3100 LB.	(D) OR (E)
LITTERS 	4 TIER LITTER 50LB. OXYGEN BOTTLE 138LB.	(D) THRU (F)
75 MM HOWITZER & CAISSON 	HOWITZER 2100LB. CAISSON 5000LB.	(D) THRU (F)
37 MM GUN & MOTOR CHASSIS 	5500LB.	(D) AND (E)
75 MM GUN & CAISSON 	GUN 3400LB. CAISSON 5000LB.	(D) THRU (F)
LARGE ENGINE & MOUNT 	4900LB.	(D) AND (E)

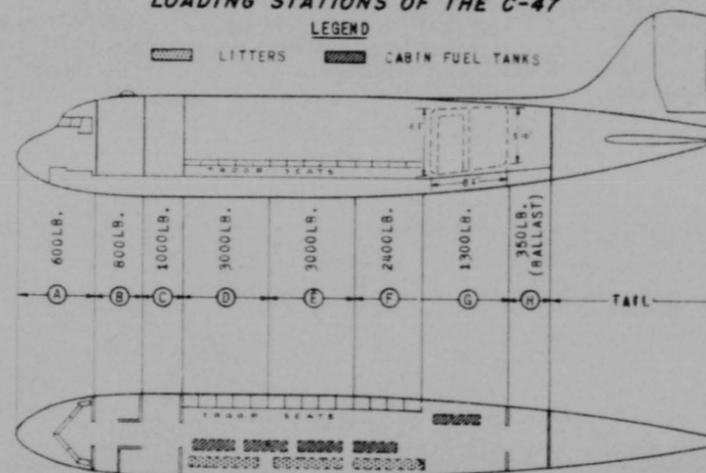
WEIGHT LEGEND

ATTENDANT OR PASSENGER—— 200LB.
 TROOP AND EQUIPMENT—— 240LB.
 LITTER (INCL. PATIENT)—— 250LB.
 PARACAN—— 250LB.
 PARATROOPER—— 260LB.

LOADING STATIONS OF THE C-47

LEGEND

 LITTERS  CABIN FUEL TANKS



(3) 4 TIER LITTERS EA. SIDE
 RH-14) TROOPS
 LH-13)

ANY COMBINATION OF CARGO CAN BE CARRIED BUT MUST SATISFY FOUR REQUIREMENTS: (1) Total weight. (2) Space limitations. (3) Center of gravity. (4) Floor loading. (Typical items are illustrated).

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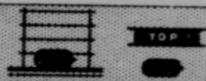
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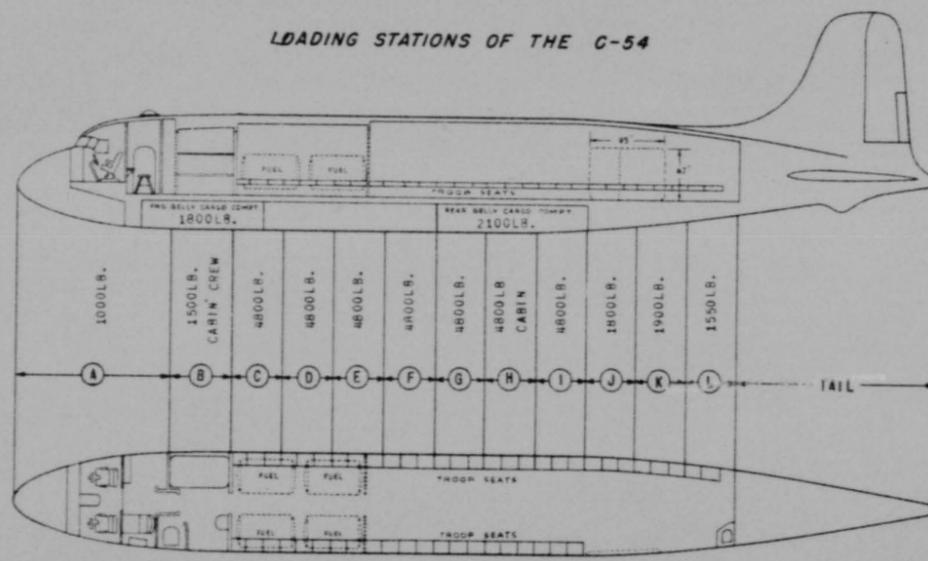
LOADING THE DOUGLAS "SKYMASTER"

ITEM	WEIGHT	STATION
JEEP 	3100 LB.	(C) THRU (L)
105 MM GUN & CAISSON 	GUN 4300 LB. CAISSON 4700 LB.	INTERNAL (C) THRU (L) EXTERNAL (E)
LITTERS 	3 TIER LITTER 50 LB. OXYGEN BOTTLE 138 LB.	(C) THRU (L)
37 MM GUN & MOTOR CHASSIS 	5500 LB.	(C) THRU (L)
75 MM GUN & CAISSON 	GUN 3400 LB. CAISSON 5000 LB.	INTERNAL (C) THRU (L) EXTERNAL (E)
LARGE ENGINE & MOUNT 	4900 LB.	(C) THRU (L)
LIGHT TANK 	15,500 LB.	EXTERNAL ONLY (E)

WEIGHT LEGEND

- ATTENDANT OR PASSENGER — 200 LB.
- TROOP AND EQUIPMENT — 240 LB.
- LITTER (INCL. PATIENT) — 250 LB.
- PARACAN — 250 LB.
- PARATROOPER — 260 LB.

LOADING STATIONS OF THE C-54



WITHOUT FUSELAGE TANKS—36 LITTERS (5 TIERS RH, 4 TIERS LH)
WITH FUSELAGE TANKS—28 LITTERS (4 TIERS RH, 3 TIERS LH)

ANY COMBINATION OF CARGO CAN BE CARRIED BUT MUST SATISFY FOUR REQUIREMENTS: (1) Total weight. (2) Space limitations. (3) Center of gravity. (4) Floor loading. (Typical items are illustrated).

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TACTICAL PLANNING CHARACTERISTICS & PERFORMANCE CHART

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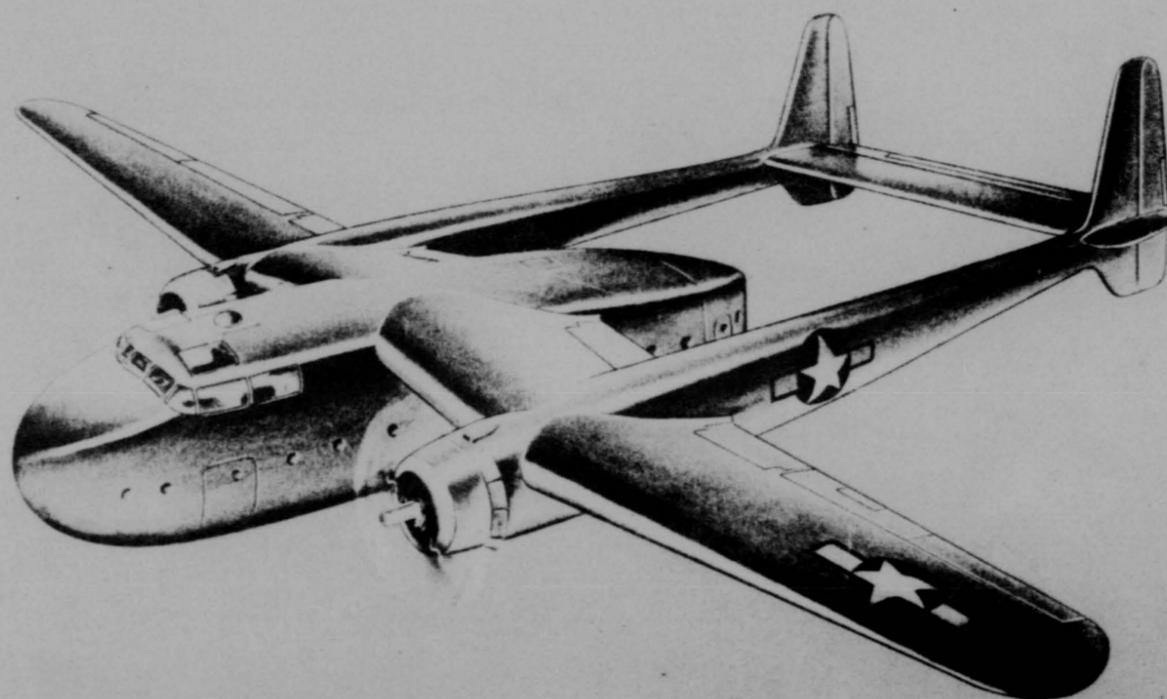
MODEL & BLOCK-NO.	ENGINE & PROP.		SIZE		WEIGHT		COMBAT CREW	FUEL & OIL		CARGO OR PASSENGERS			RADIO	REMARKS & REFERENCE									
	NUMBER	ENGINE MFR. MODEL SUPERCHARGER PROP MFR. TYPE	B.H.P./ALT.	S-SPAN L-LENGTH H-HEIGHT T-TREAD W-WING AREA	B-BASIC C-COMBAT W-WAR MAX. M-MAX LAND	NOTES		FUEL TANKAGE		STATION	CAPACITY	MAX. LOAD			STATION	PLACES	MAX. LOAD						
								TYPE OR LOCATION ALSO (MAX OIL)	NO AND CAPACITY (MAX FUEL)														
C-54A-DO -1 thru -5	4	PRATT-WHITNEY R-2000-7 2 SPEED HAMILTON STD. 13' 1" DIA. - 3 BL. F.F., HYDROMATIC	T 1350/ 5 L W NONE M 1100/ 3000 C 1000/ 14000	S 117' 6" L 93' 10" H 27' 5" T 24' 5" W 1450 Sq. Ft.	B 37000 C 73000 W 87000* M 73000	NONE	1980 - WINGS 2 x 430 OUTRD. - WINGS 2 x 420 FUSEL. (REMOVE) 4 x 450 (138) (3520)	CABIN - WING 27100 BELLY - FWD 1900 BELLY - AFT 2100 BAGGAGE 1500	32500	TROOPS 50 LITTERS 36 ATTENDANTS 4	12000	SCR-211 SCR-269 SCR-274 SCR-287 SCR-515 SCR-522 SCR-578 SCR-595 SCR-595 SCR-718 RC-32 RC-36 RC-43 RC-103 RC-198 HW-25 AN/AIC-3 AN/APN-4 AN/APN-9 AN/ARS-10 AN/ARC-3 AN/ARC-8 AN/ARC-9 AN/ARN-5 AN/ARN-7 AN/ARN-11 AN/ASA-1 AN/ASA-3 BC-1033	C-54A TECH. ORDER 01-40NM-1 (REV. 15 MAR. 1945) C-54B, DAE TECH. ORDER 01-40NS-1 (REV. 10 JAN. 1945) C-54G TECH. ORDER 01-40NU-1 (20 MAY 1945) (x) 50 TROOPS OR 36 LITTERS WITHOUT FUSELAGE TANKS. (y) 30 TROOPS OR 28 LITTERS WITH FUSELAGE TANKS. (z) 40 TROOPS OR 36 LITTERS WITHOUT FUSELAGE TANKS. (4) TROOPS OR 28 LITTERS WITH FUSELAGE TANKS.										
C-54B-DO -1 thru -20														4	AS ABOVE	AS ABOVE	B 39400 C 73000 W 87000* M 73000	NONE	1980 - WINGS 2 x 490 OUTRD. - WINGS 2 x 510 AUX. - WINGS 2 x 420 FUSEL. (REMOVE) 2 x 450 (138) (3740)	AS ABOVE	TROOPS 48 LITTERS 36 ATTENDANTS 4	11760	AS ABOVE
C-54B-DC -1																							
C-54E-DO -1 thru -20	4	PRATT-WHITNEY R-2000-11 2 SPEED HAMILTON STD. 13' 1" DIA. - 3 BL. F.F., HYDROMATIC	T 1350/ 3 L W NONE M 1100/ 16000 C 1100/ 7500 1000/ 17000	S 117' 6" L 93' 10" H 27' 5" T 24' 5" W 1450 Sq. Ft.	B 39400 C 73000 W 87000* M 73000	NONE	1980 - WINGS 2 x 500 OUTRD. - WINGS 2 x 420 INRD. W. (AUX.) 2 x 490 (138) (3520)	WITH REMOVAL OF CABIN SEATS CARGO LOADING SAME AS FOR C-54A.	PERSONNEL 44	5800	C-54G - WITH REMOVAL OF CANVAS TROOP BENCHES OR WEB TYPE LITTERS CARGO LOADING AS FOR C-54G. C-54J - WITH REMOVAL OF CABIN SEATS CARGO LOADING AS FOR C-54J.												
C-54D-DC -1 thru -15												4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE	C-54G - WITH REMOVAL OF CANVAS TROOP BENCHES OR WEB TYPE LITTERS CARGO LOADING AS FOR C-54G. C-54J - WITH REMOVAL OF CABIN SEATS CARGO LOADING AS FOR C-54J.					
C-54G (DO & DC) -1 thru -20																			4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE
C-54J-DO -1	4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE																	
C-54K-DO -1							4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE											
C-54L-DO -1	4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE																	
C-54M-DO -1							4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE											
C-54N-DO -1	4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE																	
C-54O-DO -1							4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE											
C-54P-DO -1	4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE																	
C-54Q-DO -1							4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE											
C-54R-DO -1	4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE																	
C-54S-DO -1							4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE											
C-54T-DO -1	4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE																	
C-54U-DO -1							4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE											
C-54V-DO -1	4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE																	
C-54W-DO -1							4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE											
C-54X-DO -1	4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE																	
C-54Y-DO -1							4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE											
C-54Z-DO -1	4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE																	
C-54AA-DO -1							4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE											
C-54AB-DO -1	4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE																	
C-54AC-DO -1							4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE											
C-54AD-DO -1	4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE																	
C-54AE-DO -1							4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE											
C-54AF-DO -1	4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE																	
C-54AG-DO -1							4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE											
C-54AH-DO -1	4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE																	
C-54AI-DO -1							4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE											
C-54AJ-DO -1	4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE																	
C-54AK-DO -1							4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE											
C-54AL-DO -1	4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE																	
C-54AM-DO -1							4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE											
C-54AN-DO -1	4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE																	
C-54AO-DO -1							4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE											
C-54AP-DO -1	4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE																	
C-54AQ-DO -1							4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE											
C-54AR-DO -1	4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE																	
C-54AS-DO -1							4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE											
C-54AT-DO -1	4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE																	
C-54AU-DO -1							4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE											
C-54AV-DO -1	4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE																	
C-54AW-DO -1							4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE											
C-54AX-DO -1	4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE																	
C-54AY-DO -1							4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE											
C-54AZ-DO -1	4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE																	
C-54BA-DO -1							4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE											
C-54BB-DO -1	4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE																	
C-54BC-DO -1							4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE											
C-54BD-DO -1	4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE																	
C-54BE-DO -1							4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE											
C-54BF-DO -1	4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE																	
C-54BG-DO -1							4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE											
C-54BH-DO -1	4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE																	
C-54BI-DO -1							4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE											
C-54BJ-DO -1	4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE																	
C-54BK-DO -1							4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE											
C-54BL-DO -1	4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE																	
C-54BM-DO -1							4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE											
C-54BN-DO -1	4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE																	
C-54BO-DO -1							4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE											
C-54BP-DO -1	4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE																	
C-54BQ-DO -1							4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE											
C-54BR-DO -1	4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE																	
C-54BS-DO -1							4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE											
C-54BT-DO -1	4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE																	
C-54BU-DO -1							4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE											
C-54BV-DO -1	4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE																	
C-54BW-DO -1							4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE											
C-54BX-DO -1	4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE																	
C-54BY-DO -1							4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE											
C-54BZ-DO -1	4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE																	
C-54CA-DO -1							4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE											
C-54CB-DO -1	4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE																	
C-54CC-DO -1							4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE											
C-54CD-DO -1	4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE																	
C-54CE-DO -1							4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE											
C-54CF-DO -1	4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE																	
C-54CG-DO -1							4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE											
C-54CH-DO -1	4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE																	
C-54CI-DO -1							4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE											
C-54CJ-DO -1	4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE																	
C-54CK-DO -1							4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE	AS ABOVE											
C-54CL-DO -1	4	AS ABOVE	AS ABOVE	B 40000 C 73000 W 87000* M 73000	NONE																		

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PAGE 30
AS OF:
29 JUNE 1945



PACKET
FAIRCHILD C-82

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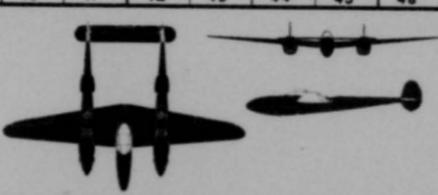
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TACTICAL PLANNING CHARACTERISTICS & PERFORMANCE CHART																		
MODEL & BLOCK-NO.	ENGINE & PROP.			SIZE		WEIGHT		COMBAT CREW	FUEL & OIL		ARMAMENT			BOMBS		RADIO	REMARKS & REFERENCE	
	NUMBER	ENGINE MFR. MODEL SUPERCHARGER PROP. MFR. TYPE	B.H.P./ALT.	S-SPAN L-LENGTH H-HEIGHT W-WING AREA	B-BASIC C-COMBAT W-WAR MAX. M-MAX LAND.	NOTES	FUEL TANKAGE		(NO. ROCKETS-SIZE-INSTALLATION)			NO. AND SIZE		MAX. LOAD	TYPICAL MODELS CARRIED (SEE A)			
							TYPE OR LOCATION ALSO (MAX. OIL)		NO. AND CAPACITY ALSO (MAX. FUEL)	NO. GUNS AND SIZE	RDS PER GUN	LOCATION AND TYPE	INTERNAL			EXTERNAL		
P-38G-L0 -1 thru -15	2	ALLISON V-1710-51 & -55 TURBO CURTISS 11'6"DIA.-3 BL. F.F., ELECTRIC	T 1225/ S L W NONE M 1150/27000 C 1100/24000	S 52' 0" L 37' 10" H 12' 10" W 328 Sq. Ft.	B 13500 C 15800 W - M -		1	MAIN - WINGS 2 x 90 RES. - WINGS 2 x 60 DROPP. - WINGS 2 x 75or 2 x 150 b (26)	2 x 90 2 x 60 2 x 75or 2 x 150 b (900)	4-.50 1-20mm	500 150	NOSE - FIXED NOSE-MI CANNON	NONE	2-1000 2-500 2-325 2-250 2-100	2000	SCR-274 SCR-522 SCR-595 NM-28 AR/AP3-13 AR/ARC-3 BC-1204	<p>REFER TO PAGE 4 FOR "FOREWORD"</p> <p>REFER TO PAGE 3 FOR SECURITY CLASSIFICATION</p> <p>REFER TO PAGE 5 FOR DEFINITIONS</p> <p>REFER TO T.O. LISTED FOR DETAILED PLANNING.</p> <p>P-38G TECH. ORDER 01-75F-1 (REV. 30 SEPT. 1944)</p> <p>P-38H, J, L TECH. ORDER 01-75-1 (REV. 15 FEB. 1945)</p> <p>(a) LIMITED TO 1150 H.P. BECAUSE OF INADEQUATE COOLING.</p> <p>(b) LAST (200) P-38G-10's HAVE MAX. FUEL AS LISTED FOR P-38H MODELS.</p> <p>(c) LAST (200) P-38G-10's HAVE BOMB LOAD AS LISTED FOR P-38H MODELS.</p> <p>(d) LIMITED TO 1240 H.P. BECAUSE OF INADEQUATE COOLING. P-38J's HAVE CORE TYPE COOLER GIVING INCREASED EX-GAME RATINGS AS LISTED.</p> <p>(e) ROCKETS INSTALLED ON P-38L-5 MODELS ONLY: CLUSTER OF 3 LOADED ROCKET TUBES (KITS ISSUED FOR EARLY MODELS) UNDER EACH WING, REDUCES TOP SPEED 20 MPH. AT 15000 FT.</p>	
P-38H-L0 -1 & -5	2	ALLISON V-1710-89 & 91 TURBO CURTISS 11'6"DIA.-3 BL. F.F., ELECTRIC	T 1425/ S L W NONE M 1240/25000 C 1100/34000	S L H AS ABOVE T W	B 13700 C 16300 W - M -		1	MAIN - WINGS 2 x 90 RES. - WINGS 2 x 90 DROPP. - WINGS 2 x 75or 2 x 150or 2 x 185 or 2 x 300 (26)	2 x 90 2 x 90 2 x 75or 2 x 150or 2 x 185 or 2 x 300 (900)	4-.50 1-20mm	500 150	NOSE - FIXED NOSE-AN - M2C	NONE	2-1600 2-1000 2-500 2-325 2-250 2-100	3200			
P-38J-L0 -5 thru -25	2	AS ABOVE	T 1425/ S L W 1600/26500 M 1425/26500 C 1100/32500	S L H AS ABOVE T W	B 14100 C 17500 W - M -		1	MAIN - WINGS 2 x 90 RES. - WINGS 2 x 90 L.L. - WINGS 2 x 55 DROPP. - WINGS 2 x 75 or 2 x 150 or 2 x 185 or 2 x 300 (1010)	2 x 90 2 x 90 2 x 55 2 x 75 or 2 x 150 or 2 x 185 or 2 x 300 (1010)									
P-38L-L0 -1 & -5	2	ALLISON V-1710-111 & -113 TURBO CURTISS 11'6"DIA.-3 BL. F.F., ELECTRIC	T 1425/ S L W 1800/28700 M 1425/29000 C 1100/33800	S L H AS ABOVE T W	B AS ABOVE C AS ABOVE W - M -		1	AS ABOVE	AS ABOVE	4-.50 1-20mm	500 150	NOSE - FIXED NOSE-AN - M2C	AS ABOVE	AS ABOVE				
P-38L-VN -5			T W M C	S L H T W	B C W M													
COLUMN	1	2	3	4	5	6	7	8	9	10	11	12	13	14	15	16	17	18

MODEL & BLOCK-NO.	TAKE OFF & LANDING HARD SURFACE-NO WIND					HIGH SPEED & CLIMB CLEAN AIRPLANE AT NORMAL COMBAT WEIGHT										RANGE AND ENDURANCE STATUTE AIR MILES-NO WIND- NO ALLOWANCE FOR RESERVE										REMARKS					
	GROSS WEIGHT LB	TO CLEAR 50'		LANDING SPEED MPH	GROUND RUN		WEIGHT LB	STD. ALT. FT	WAR EMERG.		MIL. POWER		MAX. CONTIN. POWER		LOADING			MAX. CONTINUOUS POWER				MAX. CRUISE POWER					LONG RANGE				
		T.O. DIST. FT	LAND DIST. FT		T.O. DIST. FT	LAND DIST. FT			HIGH SPEED MPH	RATE OF CLIMB FT/MIN	HIGH SPEED MPH	RATE OF CLIMB FT/MIN	HIGH SPEED MPH	RATE OF CLIMB FT/MIN	HIGH SPEED MPH	RATE OF CLIMB FT/MIN	TIME TO CLIMB MIN	TAKE OFF WEIGHT LB	BOMBS CARGO BASE LB	TOTAL FUEL U.S. GAL	at 10000 FT.		at 25000 FT.		at 10,000 FT.		at 25,000 FT.		at 10,000 FT.		
P-38G-L0 -1 thru -15	14000 16000 18000 20000	- 2800 3100 4100	85 90 2000 2500	- 1500 1700 -	15000 18000 19000 5000	30000 25000 20000 15000 10000 5000		NONE	400 390 375 360 345	1700 2300 2500 2600 2800	390 375 360 348 332	950 1600 1900 2200 2.0	11.0 8.5 6.0 4.0 2.0	19500 17500 15500 16500 17800	NONE NONE NONE 300 2000	900 600 300 300 300	1100 700 340 320 310	3.3 2.2 1.0 1.0 1.0	1100 750 350 320 310	3.4 2.2 1.0 1.0 1.0	- - 600 550 500	1.9 1.9 1.7 1.6 1.6	1700 1200 570 510 480	6.2 4.0 1.8 1.7 1.6	2200 1670 850 760 740	10.8 7.9 3.9 3.5 3.5	<p>P-38J & L MODELS HAVE INCREASED DRAG DUE TO EXTERNAL CORE TYPE COOLERS.</p> <p>WITH ONE PROP. FEATHERED, LOSS IN LOW RANGE (CLEAN AIRPLANE) IS APPROX. 15% FOR P-38G; APPROX. 25% FOR P-38L; LOSS IN MAX. CONT. POWER RANGE IS APPROX. 25% FOR P-38G; APPROX. 30% FOR P-38L.</p> <p>LOSS IN SPEED FOR P-38L AT MILITARY POWER, CAUSED BY INSTALLATION OF (10) ZERO LENGTH ROCKET RAILS, IS 17 MPH AT 10,000'. SPEED LOSS WITH (10) 5" AR OR HYAR IS 21 MPH AT 10,000' AND 24 MPH AT 25,000'.</p> <p>P-38L FLIGHT TEST IN PROGRESS</p>				
P-38H-L0 -1 & -5	14000 15000 18000 20000	- 2000 2100 2400	88 95 1300 1700	- 1400 1800 -	18300 15000 10000 5000	30000 25000 20000 15000 10000 5000		NONE	402 387 372 360 345	2500 2850 3200 3400 3500	384 370 353 336 318	11.0 8.0 6.5 5.0 2.0	20300 18300 16300 18300 19500	NONE NONE NONE 300 3200	900 600 300 300 300	1180 520 350 320 300	4.3 2.1 1.2 1.2 1.2	1150 590 370 320 300	3.5 2.0 1.2 1.2 1.2	1800 1180 500 530 500	6.5 4.3 1.7 1.6 1.6	1700 1180 570 440 450	5.7 2.8 1.8 1.8 1.8	2200 1540 850 750 700	10.2 7.2 4.2 3.6 3.4						
P-38J-L0 -5 thru -25	14000 16000 18000 20000 21000	- 2200 2600 3500 -	102 109 1300 1700 2000	- 1400 1800 -	17300 25000 20000 15000 10000 5000	30000 25000 20000 15000 10000 5000			405 414 403 390 377 360	1625 2425 2925 3275 3525 3675	385 368 351 335 317	11.0 9.0 7.0 5.0 4.0 2.0	21600 20200 17500 19500 20700	NONE NONE NONE 2000 3200	1010 740 410 410 410	1050 820 400 420 400	2.8 2.8 1.4 1.4 1.4	1070 500 470 420 410	3.8 2.8 1.4 1.4 1.4	1820 1360 900 710 580	6.8 4.7 2.8 2.7 2.8	1880 1460 840 730 720	9.8 6.9 2.8 2.4 2.4	2260 1810 1170 870 800	12.1 10.8 6.0 4.8 4.8						
P-38L-L0 -1 & -5	14500 16000 18000 20000	- 2200 2600 -	103 109 1300 1700	- 1400 1800 -	17300 25000 20000 15000 10000 5000	30000 25000 20000 15000 10000 5000																									
P-38L-VN -5	20000 22000	2800 3700	- -	1700 2200	- -	17300 25000 20000 15000 10000 5000																									
19	20	21	22	23	24	25	26	27	28	29	30	31	32	33	34	35	36	37	38	39	40	41	42	43	44	45	46	47	48		

NOTES:	(A) SCR-274 COMMAND SET SCR-522 COMMAND SET SCR-595 IFF NM-28 RADIO COMPASS AR/AP3-13 TAIL WARNING RADAR AR/ARC-3 COMMAND SET BC-1204 BEACON RECEIVER	RED FIGURES ARE PRELIMINARY; SUBJECT TO REVISION AFTER FLIGHT CHECK. * FIRST 5 MINUTES WITH MILITARY POWER. # TIME TO CLIMB FROM S.L. BASED ON MILITARY POWER.
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P-38
"LIGHTNING"

MANUFACTURED BY LOCKHEED (BURBANK) "LO" & VOLTEC (MADISONVILLE) "VM"

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TACTICAL PLANNING CHARACTERISTICS & PERFORMANCE CHART																		
MODEL & BLOCK-NO.	ENGINE & PROP.			SIZE		WEIGHT		COMBAT CREW	FUEL & OIL		ARMAMENT			BOMBS		RADIO	REMARKS & REFERENCE	
	NUMBER	ENGINE MFR. MODEL SUPERCHARGER PROP. MFR. TYPE	B.H.P./ALT.	S-SPAN L-LENGTH H-HEIGHT T-TREAD W-WING AREA	B-BASIC C-COMBAT W-WAR MAX. M-MAX.LAND.	NOTES	FUEL TANKAGE		NO GUNS AND SIZE	RDS. PER GUN	LOCATION AND TYPE	NO. AND SIZE		MAX. LOAD	TYPICAL MODELS CARRIED (SEE A)			
							TYPE OR LOCATION ALSO (MAX. OIL)					NO. AND CAPACITY ALSO (MAX. FUEL)	INTERNAL			EXTERNAL		
P-39N-BE -0, -1 & -5	1	ALLISON V-1710-B5 SINGLE SPEED AERO PRODUCTS 11'7" DIA. - 3 BL. CONST. SPD. - HYDRD.	T 1200/ 3 L W 1430/ 9600 M 1125/ 16400 C 1000/ 14000	S 34' 0" L 30' 2" H 12' 5" T 11' 4" W 213 Sq. Ft.	B 6400 C 7700 W - M -		WING - CELLS HELLY - DROP. (14)	12 - 120 1 x 75 gal 1 x 158 gal 1 x 175	295	6 (4.5") M-10 ROCKETS IN CLUSTERS OF 3 PROCURED IN SERVICE KITS FOR SOME MODELS	4 - 30 2 - 50 1 - 37mm	300 200 30	WINGS - FIXED NOSE - FIXED PROP. - TYPE M4	NONE	1-500 1-225 1-250 1-100	500	SCR-274 SCR-522 SCR-535 SCR-695	P-39N SERIES - TECH. ORDER 01-110FM-1 (REV. 25 OCT. 1944) P-39Q SERIES - TECH. ORDER 01-110FM-1 (20 MARCH 1944) (a) P-39Q-30 HAS PROP. DIA. OF 11' 0". (b) P-39Q-20, -25 & -30 DO NOT HAVE WING GUNS.
P-39Q-BE -1 thru -30	1	AS ABOVE	AS ABOVE	AS ABOVE	B 6400 C 7800 W - M -		AS ABOVE	AS ABOVE	295	6 (4.5") M-10 ROCKETS IN CLUSTERS OF 3 PROCURED IN SERVICE KITS FOR SOME MODELS	2 - 50 2 - 50 1 - 37mm	300 200 30	WINGS - FIXED NOSE - FIXED PROP. - TYPE M4	AS ABOVE	AS ABOVE			

MODEL & BLOCK-NO.	TAKE OFF & LANDING HARD SURFACE - NO WIND					HIGH SPEED & CLIMB CLEAN AIRPLANE AT NORMAL COMBAT WEIGHT										RANGE AND ENDURANCE STATUTE AIR MILES - NO WIND - NO ALLOWANCE FOR RESERVE										REMARKS
	GROSS WEIGHT	TO CLEAR 50'		GROUND RUN		WEIGHT	STD. ALT.	WAR EMERG.		MIL. POWER		MAX. CONTIN. POWER		LOADING			MAX. CONTINUOUS POWER		MAX. CRUISE POWER		LONG RANGE					
		LB	T.O. DIST. FT.	LAND DIST. FT.	LANDING SPEED M.P.H.			T.O. DIST. FT.	LAND DIST. FT.	HIGH SPEED M.P.H.	RATE OF CLIMB FT. MIN.	HIGH SPEED M.P.H.	RATE OF CLIMB FT. MIN.	HIGH SPEED M.P.H.	RATE OF CLIMB FT. MIN.	TIME TO CLIMB MIN.	TAKE OFF WEIGHT LB.	BOMBS CAPAC. LB.	TOTAL FUEL U.S. GAL.	at 10000 FT. RANGE MILES	at 10000 FT. ENDUR. HOURS	at 25000 FT. RANGE MILES	at 25000 FT. ENDUR. HOURS	at 10000 FT. RANGE MILES	at 10000 FT. ENDUR. HOURS	
P-39N-BE -0, -1 & -5	5800 7500 8000 8500 9000	- 2500 3000 3500 4200	- - - - -	80 - - - -	1500 - - - -	1200 - - - -	25000 20000 15000 10000 5000	7700	NO DATA	328 375 376 357 325	1175 1825 2450 2600 2600	390 385 345 320 320	11.5 8.1 5.8 3.8 1.9	9900 8800 8200 7700 8200	NONE NONE NONE NONE 500	295 278 195 120 120	550 520 370 270 220	2.0 1.9 1.4 0.8 0.9	950 900 650 480 350	3.5 3.3 2.5 1.4 1.3	- - 570 350 270	2.5 2.5 1.5 1.5	ABOVE CEILING OF AIRPLANE IN MAX. CRUISE POWER	1100 1050 750 570 480	5.4 5.3 2.9 2.2 2.1	
P-39Q-BE -1 thru -30	AS ABOVE	AS ABOVE	AS ABOVE	AS ABOVE	AS ABOVE	AS ABOVE	7800	AS ABOVE	AS ABOVE	AS ABOVE	AS ABOVE	AS ABOVE	AS ABOVE	AS ABOVE	AS ABOVE	AS ABOVE	AS ABOVE	AS ABOVE	AS ABOVE	AS ABOVE	AS ABOVE	AS ABOVE	AS ABOVE	AS ABOVE		

NOTES: (A) SCR-274 COMMAND SET
SCR-522 COMMAND SET
SCR-535 IFF
SCR-695 IFF

RED FIGURES ARE PRELIMINARY; SUBJECT TO REVISION AFTER FLIGHT CHECK.
* TIME TO CLIMB FROM S.L. BASED ON MILITARY POWER.



P-39
"AIRACOBRA"

MANUFACTURED BY BELL (BUFFALO) "NE"

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TACTICAL PLANNING CHARACTERISTICS & PERFORMANCE CHART

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AS OF 28 JULY 1945

MODEL & BLOCK-NO.	ENGINE & PROP.		SIZE	WEIGHT		COMBAT CREW	FUEL & OIL		ARMAMENT			BOMBS		RADIO	REMARKS & REFERENCE
	NUMBER	ENGINE MFR. MODEL SUPERCHARGER PROP. MFR. TYPE		B.H.P./ALT.	S-SPAN L-LENGTH H-HEIGHT T-TREAD W-WING AREA		B-BASIC C-COMBAT W-WAR MAX. M-MAX. LAND	FUEL TANKAGE	NO. AND CAPACITY ALSO (MAX. OIL) (MAX. FUEL)	(NO. ROCKETS-SIZE-INSTALLATION)			NO. AND SIZE		
			T-TAKE OFF W-WAR EMERG M-MILITARY C-CONTINUOUS	NO. GUNS AND SIZE			RDS. PER GUN			LOCATION AND TYPE	INTERNAL	EXTERNAL			
P-51B-NA -1, -5, -7, -10 P-51C-NT -1 & -3	1	PACKARD V-1650-3 2 SPEED-2 STAGE HAMILTON STD. 11" DIA. - 4 BL. CONST. SPD., HYDRO.	T-1450/1800 W-1295/25200 M-1450/19800 C-1150/31000 940/34000	S-37' 0" L-32' 3" H-13' 8" T-11' 10" W-233 Sq. Ft.	B-7400 C-9800	1	MAIN - R. WING MAIN - L. WING FUSELAGE DROPP. - WINGS (2)	1 x 92 1 x 92 1 x 85 2 x 75 or 2 x 110 or 2 x 150 (569)	8 (5") ROCKETS OR 4 (5") M-10 ROCKETS PROCURED IN SERVICE KITS FOR SOME MODELS.	280 350	OUTRD. - WINGS INRD. - WINGS	NONE 2-1000 2-500 2-325 2-250 2-100	2000	SCR-274 SCR-515 SCR-522 SCR-535 SCR-595 SCR-595 AN-ARA-8 AN-ARC-3 AN-APS-13 SC-1206 NW-26	P-51B & C TECH. ORDER 01-80JD-1 (REV. 25 MAY 1945) P-51D & K TECH. ORDER 01-80JE-1 (REV. 25 MAY 1945) P-51H TECH. ORDER 01-80JF-1 (REV. 25 APR. 1945) (a) P-51C-NT HAS 11" DIA. AEROPRODUCTS PROPELLER INSTALLED. GIVING APPROX. SAME PERFORMANCE AS SHOWN WITH HAMILTON STD. PROP. (b) ROCKETS INSTALLED ON LAST (1100) P-51D-25-NA; P-51D-10-NT; P-51K-10 & -15 AND SUBSEQUENT MODELS. KITS PROCURED FOR SERVICE INSTALLATION ON SOME OF OTHER MODELS.
P-51B-NA -15 P-51C-NT -5 & -10	1	PACKARD V-1650-7 2 SPEED-2 STAGE HAMILTON STD. 11" DIA. - 4 BL. CONST. SPD., HYDRO.	T-1450/1800 W-1295/25200 M-1450/19800 C-1150/31000 940/34000	AS ABOVE	AS ABOVE	1	AS ABOVE	AS ABOVE	AS ABOVE	AS ABOVE	AS ABOVE	AS ABOVE	AS ABOVE		
P-51D-NA -5 thru -30 P-51D-NT -5 thru -25 P-51E-NT -1 thru -15	1	AS ABOVE EXCEPT P-51D-25-NA V-1650-9 ENGINE INSTALLED	AS ABOVE	AS ABOVE	AS ABOVE	1	AS ABOVE	AS ABOVE	AS ABOVE	AS ABOVE	AS ABOVE	AS ABOVE	AS ABOVE		
P-51H-NT	1	PACKARD V-1650-9 2 SPEED-2 STAGE AEROPRODUCTS 11" DIA. - 4 BL. CONST. SPD., HYDRO.	T-1840/22000 W-2220/10200 M-1800/23400 C-1470/21800 1190/34900 1080/22100 950/35000	S-37' 0" L-33' 4" H-13' 8" T-11' 1" W-235 Sq. Ft.	B-7050 C-10100	1	AS ABOVE	AS ABOVE	AS ABOVE	AS ABOVE	AS ABOVE	AS ABOVE	AS ABOVE		

MODEL & BLOCK-NO.	TAKE OFF & LANDING HARD SURFACE-NO WIND						HIGH SPEED & CLIMB CLEAN AIRPLANE AT NORMAL COMBAT WEIGHT						RANGE AND ENDURANCE STATUTE AIR MILES-NO WIND-NO ALLOWANCE FOR RESERVE										REMARKS					
	GROSS WEIGHT LB.	TO CLEAR 50'		LANDING SPEED MPH	GROUND RUN		STD. ALT. FT.	WAR EMERG.		MIL. POWER		MAX. INTIN. POWER		LOADING		MAX. CONTINUOUS POWER		MAX. CRUISE POWER		LONG RANGE								
		T.O. DIST. FT.	LAND DIST. FT.		T.O. DIST. FT.	LAND DIST. FT.		HIGH SPEED MPH	RATE OF CLIMB FT. MIN.	HIGH SPEED MPH	RATE OF CLIMB FT. MIN.	HIGH SPEED MPH	RATE OF CLIMB FT. MIN.	TAKE OFF WEIGHT LB.	BOMBS TOTAL LB.	TOTAL FUEL U.S. GAL.	RANGE MILES	ENDUR. HOURS	RANGE MILES	ENDUR. HOURS	RANGE MILES	ENDUR. HOURS		RANGE MILES	ENDUR. HOURS			
P-51B & C SERIES WITH (V-1650-3)	8000 9000 10000 11000 12000	- 2200 2500 3000 3500	2100 2300 2500 - - 2400	190 106 111 - - 2400	1100 1200 1300 - - 2400	9800	30000 25000 20000 15000 10000 5000	440 430 427 424 406 388	1825 2425 2525 3075 3150 3050	435 418 427 414 392 374	1425 1825 2175 2475 2700 2600	415 391 390 405 373 335	1125 1200 11.5 8.5 1800 1770	11800 11900 9800 10800 11800	2000 2000 1000 1000 2000	269 569 489 259 269	860 1520 1270 1100 740 660	2.2 4.7 4.0 3.4 2.1 2.2	720 1880 1440 1240 810 720	2.2 5.2 4.1 3.5 2.2 2.2	1000 2000 1670 1570 1050 1000	3.6 7.3 6.8 6.0 4.2 3.7	1150 2100 1720 1720 1150 1070	3.9 7.1 6.0 6.0 3.9 3.6	1180 2540 1840 1280 1180 1070	4.9 10.8 7.7 5.1 4.9 4.3		
P-51B & C SERIES WITH (V-1650-7)	AS ABOVE						9800	30000 25000 20000 15000 10000 8000	485 439 426 414 417 325	1925 2100 2825 2900 3075 3550	435 428 414 414 416 281	1425 1420 1950 2450 1800 1750	415 391 390 405 373 335	1125 1200 11.5 8.5 1800 1770	11800 11900 9800 10800 11800	2000 2000 1000 1000 2000	269 569 489 259 269	860 1520 1270 1100 740 660	2.2 5.4 4.6 3.9 2.4 2.4	720 1950 1450 1430 950 870	2.2 5.4 4.6 4.0 2.4 2.4	1000 2200 1700 1550 1100 950	3.6 7.6 6.1 5.0 3.3 3.6	1150 2100 1850 1670 1050 1020	3.9 7.1 6.1 6.0 2.8 3.1	1180 2440 1850 1300 1150 1080	4.9 9.8 7.7 5.0 4.8 4.3	
P-51D & K SERIES	AS ABOVE						10100	25000 20000 15000 10000 5000	437 424 413 416 395	1940 2575 2775 2925 3475	426 412 413 401 381	1860 1960 2300 2825 3190	404 388 387 360 340	11900 11100 10100 11100 12100	NONE NONE NONE NONE 2000	489 419 259 259 269	1500 1270 870 790 770	4.6 3.9 2.4 2.4 2.4	1650 1410 950 870 840	4.6 3.9 2.4 2.4 2.5	1620 1550 1050 1000 950	5.4 4.9 3.2 3.7 3.5	1770 1570 1050 950 900	5.4 4.9 2.8 3.0 2.8	2080 1830 1300 1150 1070	8.5 7.7 5.0 4.8 4.3		
P-51H-NT	AS ABOVE						9300	30000 25000 20000 15000 10000 5000	485 467 463 463 389	1925 1900 1400 1475 5350	435 425 415 395 370	1425 1425 1825 1790 1750	415 391 387 360 340	1125 1125 9.0 8.0 1750	11800 10500 9500 10500 11500	NONE NONE NONE NONE 2000	475 405 255 255 255	1820 1150 750 870 850	4.0 3.4 2.1 2.1 2.1	1470 1270 800 720 670	4.2 3.5 2.1 2.1 2.1	1670 1400 900 820 800	5.0 5.0 2.7 2.8 2.8	1750 1570 970 860 800	6.2 7.2 4.9 4.3 4.0			

NOTES: (A) THREE OR MORE OF THE FOLLOWING:
 SCR-274 COMMAND SET
 SCR-515 IFF
 SCR-522 COMMAND SET
 SCR-535 IFF
 SCR-595 IFF
 SCR-595 IFF
 AN-ARA-8 HOMING ADAPTER
 AN-ARC-3 COMMAND (W/F)
 AN-APS-13 TAIL WARNING RADAR
 SC-1206 BEACON RECEIVER
 NW-26 RADIO COMPASS

RED FIGURES ARE PRELIMINARY: SUBJECT TO REVISION AFTER FLIGHT CHECK.
 * WAR EMERGENCY HIGH SPEED AND RATE OF CLIMB BASED ON 90° HD. (WATER INJECTION) USING AN-F-33 FUEL.

LOSSES DUE TO ROCKET INSTALLATIONS:
 (1) COMBAT RADIUS OF ACTION DECREASED BY 12%.
 (2) (10") HVAR ON POST TYPE LAUNCHERS INCREASE TAKE-OFF GROUND RUN BY 20%.
 (3) 4.5 (HS) ROCKET WITH ADAPTER HAS SAME DRAG AS 5" HVAR.

TEST RESULTS BASED ON 87° HD (DRY ENGINE)

WEIGHT LB.	STD. ALT. FT.	WAR EMERG.	
		HIGH SPEED MPH	RATE OF CLIMB FT. MIN.
30000	450	450	2250
25000	435	435	2750
20000	430	430	3050
15000	430	430	3220
10000	405	405	3350
5000	380	380	3300

MANUFACTURED BY NORTH AMERICAN (INGLEWOOD) "NA" & (DALLAS) "NT"



P-51 "MUSTANG"

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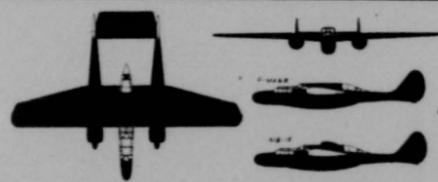
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TACTICAL PLANNING CHARACTERISTICS & PERFORMANCE CHART																		
MODEL & BLOCK-NO.	ENGINE & PROP.		SIZE	WEIGHT		COMBAT CREW	FUEL & OIL		ARMAMENT			BOMBS			RADIO	REMARKS & REFERENCE		
	NUMBER	ENGINE MFR. MODEL SUPERCHARGER PROP MFR. TYPE	B.H.P./ALT.	S-SPAN L-LENGTH H-HEIGHT T-TREAD W-WING AREA	B-BASIC C-COMBAT W-WAR MAX. M-MAX.LAND.		NOTES	FUEL TANKAGE		NO. AND SIZE			NO. AND SIZE			TYPICAL MODELS CARRIED (SEE A)	REFER TO PAGE 4 FOR "FOREWORD." REFER TO PAGE 3 FOR SECURITY CLASSIFICATION. REFER TO PAGE 5 FOR DEFINITIONS. REFER TO T.O. LISTED FOR DETAILED PLANNING.	
			T-TAKE OFF W-WAR EMERG. M-MILITARY C-CONTINUOUS					TYPE OR LOCATION ALSO (MAX. OIL)	NO. AND CAPACITY ALSO (MAX. FUEL)	NO. GUNS AND SIZE	RDS. PER GUN	LOCATION AND TYPE	INTERNAL	EXTERNAL	MAX. LOAD			
P-61A-NO -1	2	PRATT-WHITNEY R-2800-10 2 SPD.-2 STD. CURTISS 12" 2" DIA. -4 BL F.F., ELECTRIC	T-2000/4000 W-1800/1600 M-1675/1800 C-1550/22700	S 65' 0" L 98' 11" H 14' 8" T 17' 2" W 664 Sq. Ft.	B 22300 C 27500 W - M -	2 OF 3	OUTRD.-WINGS 2 x 205 INBD.-WINGS 2 x 115 (44)	4-.50 560 4-20mm 200	UPPER TURRET FORE-FUSELAGE	N O N E				SCR-522 (2) SCR-695 SCR-718 SCR-720 SCR-729 RC-32 RC-36 RC-198 MW-26 AN/APG-1 AN/APM-1 AN/APM-13 AN/APC-3 AN/ARM-7 AN/ASA-3 BC-1206	P-61A & B TECH. ORDER 01-15FB-1 (REV. 10 MAY 1945) (a) INSTALLED ON FIRST 37 (P-61A-1) ONLY. (b) THE (2) 310 GAL. DROPPABLE WING TANKS OR 2-1600 LB. BOMBS INSTALLED ONLY ON P-61A-11; P-61B-2, -6 & -11 MODELS; OTHERS AS ABOVE.			
P-61A-NO -5, -10, -11 P-61B-NO -1, -2, -5, -6, -11	2	AS ABOVE EXCEPT CHANGE TO R-2800-65 ENGINE	T-2000/4000 W-2250/1000 M-2040/12800 C-1930/17000	AS ABOVE EXCEPT P-61B HAS 48' 7" LENGTH	B 22300 C 28000 W - M -	2	OUTRD.-WINGS 2 x 205 INBD.-WINGS 2 x 115 DROPP.-WINGS 2 x 165 or 2 x 310 (84)	4-20mm 200	FORE-FUSELAGE	NONE			2-1500 lb 2-1000 2-500 2-325 2-250 2-100					
P-61B-NO -10	2	AS ABOVE	M-2000/4000 1800/16000 1650/20900	AS ABOVE	B 22650 C 28000 W - M -	2	OUTRD.-WINGS 2 x 205 INBD.-WINGS 2 x 115 DROPP.-WINGS 4 x 165 or 4 x 310 (84)	AS ABOVE	AS ABOVE	NONE			4-1600 4-1000 4-500 4-325 4-250 4-100					
P-61B-NO -15, -16, -20, -25	2	AS ABOVE	C-1675/18400 1625/17800 1550/22700	AS ABOVE	B 23450 C 29700 W - M -	3	AS ABOVE EXCEPT P-61B-16 CARRIES ONLY (2) DROPPABLE WING TANKS.	4-.50 560 4-20mm 200	UPPER TURRET FORE-FUSELAGE	AS ABOVE EXCEPT P-61B-16 HAS ONLY (2) EXTERNAL BOMB STATIONS IN PLACE OF (4)								
COLUMN	1	2	3	4	5	6	7	8	9	10	11	12	13	14	15	16	17	18

MODEL & BLOCK-NO.	TAKE OFF & LANDING HARD SURFACE-NO WIND				HIGH SPEED & CLIMB CLEAN AIRPLANE AT NORMAL COMBAT WEIGHT										RANGE AND ENDURANCE STATUTE AIR MILES-NO WIND-NO ALLOWANCE FOR RESERVE										REMARKS			
	GROSS WEIGHT LB	TO CLEAR 50'		LANDING SPEED MPH	GROUND RUN		STD. ALT. FT	WAR EMERG.		MIL. POWER		MAX. CONTIN. POWER		LOADING			MAX. CONTINUOUS POWER		MAX. CRUISE POWER		LONG RANGE							
		T.O. DIST. FT	LAND DIST. FT		T.O. DIST. FT	LAND DIST. FT		WEIGHT LB	HIGH SPEED MPH	RATE OF CLIMB FT/MIN	HIGH SPEED MPH	RATE OF CLIMB FT/MIN	HIGH SPEED MPH	RATE OF CLIMB FT/MIN	TIME TO CLIMB MIN	TAKE OFF WEIGHT LB	BOMBS GAL	TOTAL FUEL U.S. GAL	at 10000 FT. RANGE MILES	at 10000 FT. ENDUR. HOURS	at 25000 FT. RANGE MILES	at 25000 FT. ENDUR. HOURS	at 10000 FT. RANGE MILES	at 10000 FT. ENDUR. HOURS				
P-61A-NO -1	24000	-	2100	-	1900	27000	30000	-	-	350	1050	350	500	18.1	27500	NONE	640	370	1.2	420	1.3	810	2.7	790	2.9	1020	4.5	PERFORMANCE SHOWN WITH-OUT DE-ICERS.
P-61A-NO -5, -10, -11 P-61B-NO -1, -2, -5, -6, -11	24000	2600	2200	-	1900	28000	30000	358	1025	350	1025	350	475	18.3	32400	NONE	1280	750	2.4	-	-	-	-	-	-	1900	8.6	(c) CRUISE AT 15000' UNTIL EXTERNAL TANKS ARE EMPTY; DROP EXTERNAL TANKS; CLIMB TO 25000' AND COMPLETE FLIGHT.
P-61B-NO -10	28000	2900	2300	-	1900	28000	30000	369	1625	358	1550	355	475	10.3	30400	NONE	970	560	1.8	620	1.8	1140	4.5	1030	4.1	1425	6.3	P-61B FLIGHT TEST IN PROGRESS
P-61B-NO -15, -16, -20, -25	30000	3300	-	2200	-	29700	30000	364	2250	350	1900	343	475	7.2	28000	NONE	640	370	1.2	420	1.3	810	2.7	790	2.9	1020	4.5	

- NOTES:
(A) SEVER OR MORE OF FOLLOWING:
SCR-274 COMMAND SET
SCR-522 COMMAND SET
SCR-695 IFF
SCR-718 RADAR ALTIMETER
SCR-720 RADAR SET IN NOSE
SCR-729 INTERROGATOR
RC-32 FILTER EQUIPMENT
RC-36 INTERPHONE AMPLIFIER
RC-198 FILTER EQUIPMENT
MW-26 RADIO COMPASS
AN/APG-1 GUN LAYING RADAR
AN/APM-1 LOW ALT. ALTIMETER
AN/APD-13 TAIL WARNING RADAR
AN/ARC-5 COMMAND UNIT
AN/ARM-7 RADIO COMPASS
AN/ASA-3 STATIC DISCHARGE
BC-1206 BEACON RECEIVER

RED FIGURES ARE PRELIMINARY; SUBJECT TO REVISION AFTER FLIGHT CHECK.
* TIME TO CLIMB FROM S.L. BASED ON MILITARY POWER.



P-61
"BLACK WIDOW"

MANUFACTURED BY NORTHROP (NORTHROP) "NO"

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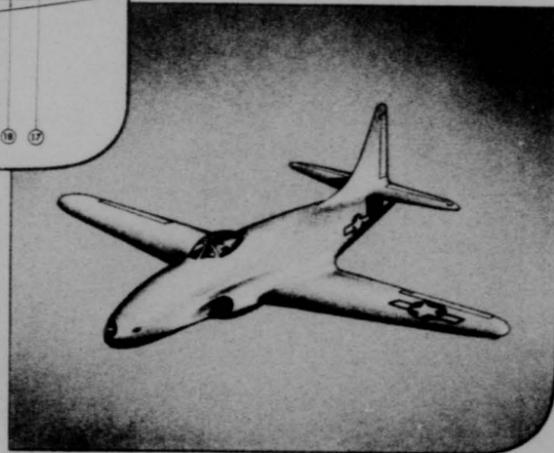
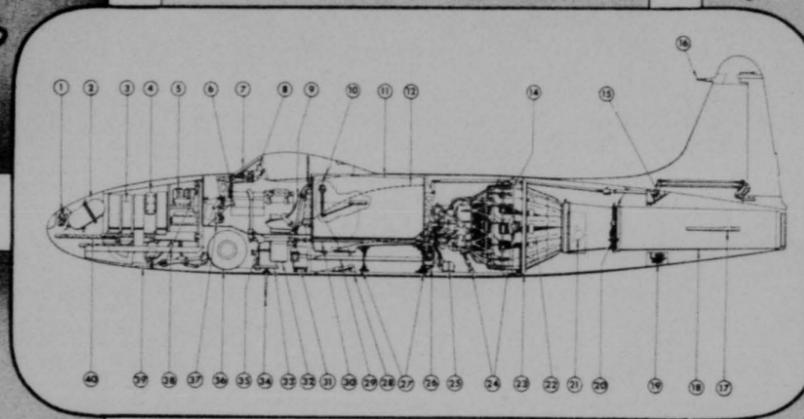
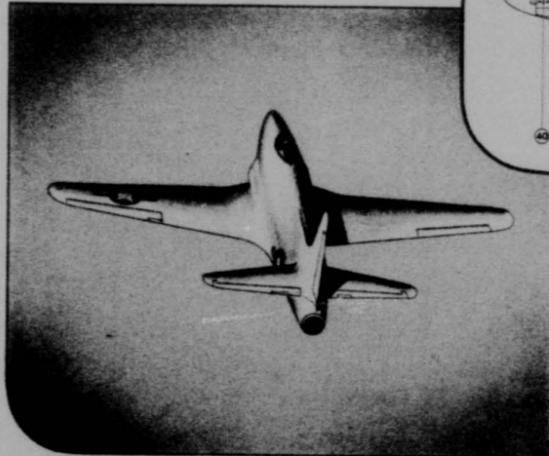
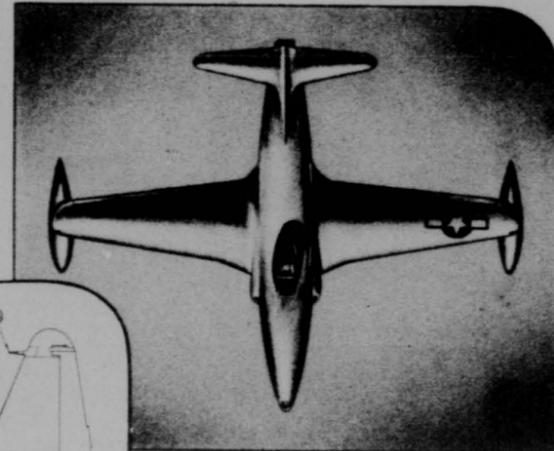
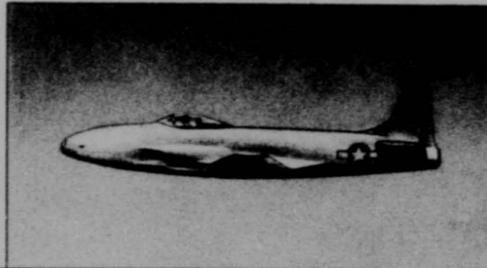
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PAGE 38
AS OF:
29 JUNE 1945

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LOCKHEED P-80A

Shooting Star



- | | |
|-----------------------------|----------------------------------|
| 1. Landing light | 21. Elevator tab motor |
| 2. Oxygen cylinder | 22. Engine |
| 3. Ammunition box | 23. Air intake seal |
| 4. Armament junction box | 24. Engine mounts |
| 5. Command radio | 25. Fuel filter |
| 6. Instrument panel | 26. Aileron booster |
| 7. Bullet-proof windshield | 27. Wing spars |
| 8. Gun sight | 28. Dive flaps |
| 9. Pilot's seat | 29. Aileron torque tube |
| 10. Fuel gage transmitter | 30. Elevator push-pull tube |
| 11. Main fuel tank | 31. Identification radio |
| 12. Air intake duct | 32. Electrical junction box |
| 14. Aft fuselage joint | 33. Battery |
| 15. Elevator spring | 34. Identification radio antenna |
| 16. Air speed pitot | 35. Elevator-aileron control |
| 17. Tail pipe support track | 36. Nose wheel |
| 18. Tail pipe | 37. Rudder pedals |
| 19. Remote compass | 38. Nose section joint |
| 20. Tail pipe clamp | 39. Shell case ejection doors |
| | 40. Machine guns-.50 cal. |

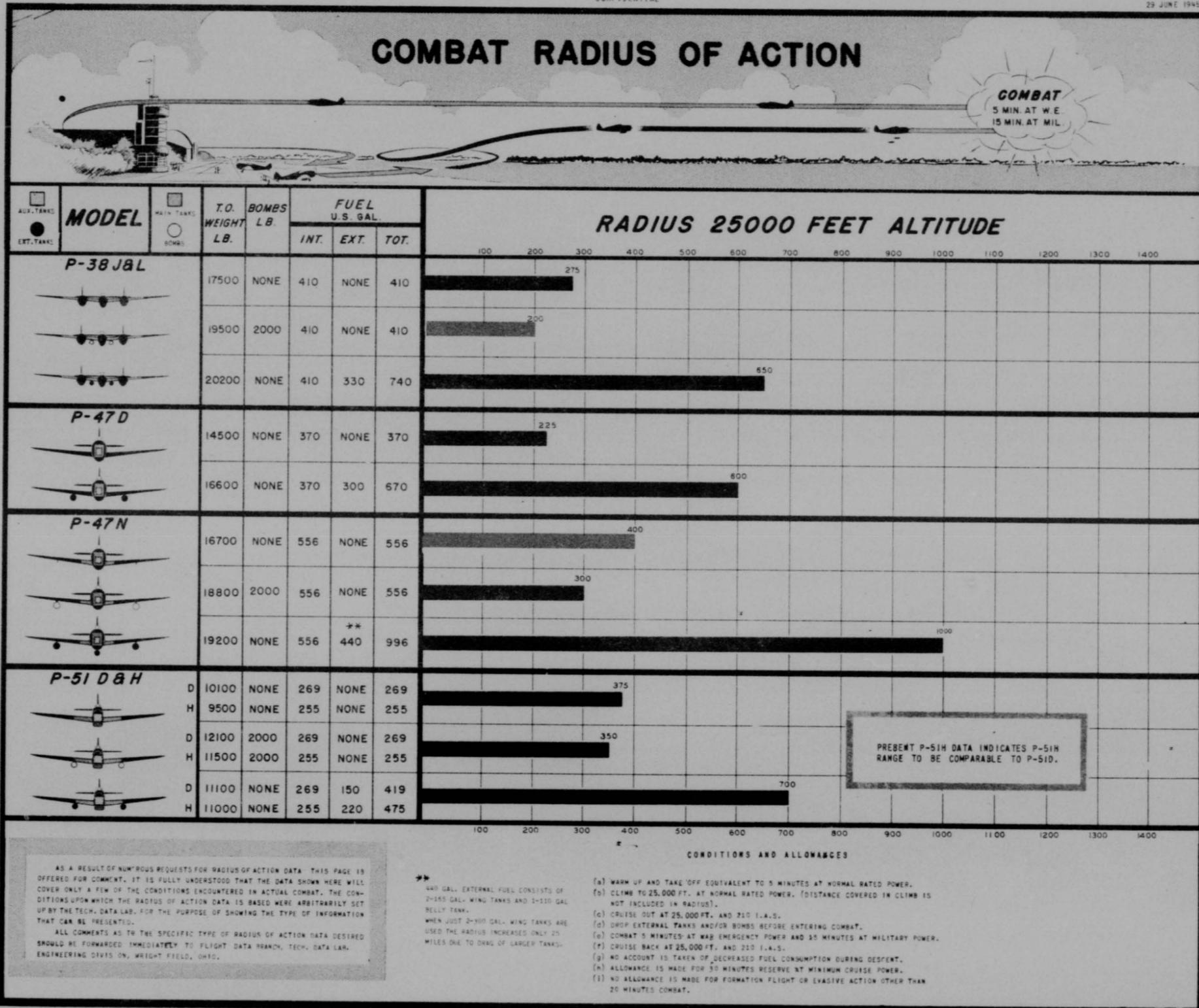
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29 JUNE 1945



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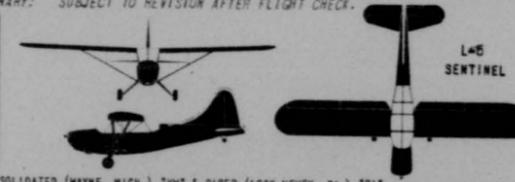
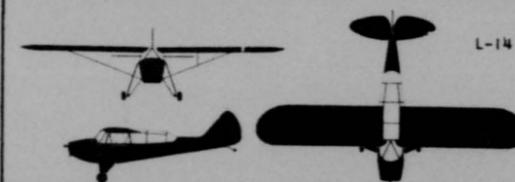
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TACTICAL PLANNING CHARACTERISTICS & PERFORMANCE CHART																	
MODEL & BLOCK-NO.	ENGINE & PROP.		SIZE	WEIGHT	COMBAT CREW	FUEL & OIL		CARGO OR PASSENGERS			RADIO	REMARKS & REFERENCE					
	NUMBER	ENGINE MFR. MODEL SUPERCHARGER PROP. MFR. TYPE				B.H.P./ALT.	S-SPAN	B-BASIC	TYPE OR LOCATION (MAX. OIL)	NO. AND CAPACITY (MAX. FUEL)		STATION	CAPACITY	MAX. LOAD	STATION	PLACES	MAX. LOAD
			T - TAKE OFF W - WAR EMERG. M - MILITARY C - CONTINUOUS	L-LENGTH H-HEIGHT T-TREAD W-WING AREA		G-COMBAT W-WAR MAX. M-MAX LAND.	NO. OF	LB			LB						
L-5 L-5B L-5C VW L-5E L-5E-1	1	LYCOMING D-435-1 NONE SENSEICH 7' 1" DIA., -2 9L. FIXED-WOOD	T 185/ 5 L W NONE M NONE C 185/ 5 L	S 34' 0" L 24' 1" H 7' 11" T 7' 5" W 155 Sq. Ft.	B 1550 C 2200 W 2200 M 2200	2 or 1	WINGS (3) FUSELAGE (2)	2 x 12 (36) 1 x 2 (30)	L-5 CAN CARRY PASSENGER OR 200 LB. BAGGAGE IN LIEU OF OBSERVER. L-5B AND SUBSEQUENT CAN CARRY CASUALTY IN LITTER OR 200 LB. BAGGAGE IN LIEU OF OBSERVER.	SCR-274K AVT-112 AFB-204 AN/APR-13 SCR-510 GT-110 AN/ARC-3	L-5 SERIES L-14	TECH. ORDER 01-500B-1 (10 AUGUST 1945) TECH. ORDER 01-1400D-1 (25 JULY 1945)					
L-5G-VW	1	LYCOMING D-435-11 NONE ANNESLEY-2 BLADE 7' 1" METAL CONTROLLABLE PITCH	T 185/ 5 L W NONE M NONE C 185/ 5 L	S L H T W ABOVE	B 1525 C 2300 W 2300 M 2300				A B O V E								
L-14-PI	1	LYCOMING D-290-3 NONE SENSEICH 6' 4" DIA.-2BL. FIXED-WOOD	T 130/ 5 L W NONE M NONE C 125/ 5 L	S 35' 10" L 23' 0" H 7' 0" T 8' 4" W 181 Sq. Ft.	B 1100 C 1800 W 1800 M -	2 or 1	WINGS FUSELAGE (2)	2 x 14 (30)	N O N E	CABIN	1 or 2	170 or 340	SCR-510A				

MODEL & BLOCK-NO.	TAKE OFF & LANDING				HIGH SPEED & CLIMB						RANGE AND ENDURANCE										REMARKS						
	HARD SURFACE-NO WIND				CLEAN AIRPLANE AT NORMAL COMBAT WEIGHT						STATUTE AIR MILES-NO WIND- NO ALLOWANCE FOR RESERVE																
	GROSS WEIGHT	TO CLEAR 50'		LANDING SPEED	GROUND RUN		WEIGHT	STD. ALT.	WAR EMERG.		MIL POWER		MAX CONTIN. POWER		LOADING		MAX. CONTINUOUS POWER		MAX. CRUISE POWER			LONG RANGE					
T.O. DIST.		LAND DIST.	T.O. DIST.		LAND DIST.	HIGH SPEED			RATE OF CLIMB	HIGH SPEED	RATE OF CLIMB	HIGH SPEED	RATE OF CLIMB	TIME TO CLIMB	TAKE OFF WEIGHT	BOMBS CARGO PASS.	TOTAL FUEL	at 5000FT.	at 15000FT.	at 5000FT.	at 15000FT.	at 10000FT.					
L-5 L-5B L-5C VW L-5E L-5E-1	1800 1900 2000 2100 2200	- - 1050 1150 1300	875 925 950 1000 -	- - - 43 -	475 500 525 550 -	2200 2000 2000 2000 2000	15000 10000 5000 5000 5 L	- - - - -	NOT APPLICABLE	NOT APPLICABLE	115 118 122 130	150 400 650 900	35.5 18.0 8.4 0.0	2200 ^b 2200 2200 2200	200 ^b NONE NONE NONE	36 36 36 36	250 250 250 250	1.8 1.9 - -	NOT APPLICABLE	380 350 350 350	3.4 3.2 - -	NOT APPLICABLE	420 ^c 400 ^c -				
L-5G-VW	1900 2000 2100 2200 2300	950 1050 1150 1300 -	750 800 825 850 -	- - 43 -	400 400 425 450 -	2300 2300 2300 2300 2300	10000 5000 5000 5000 5 L	- - - -	NOT APPLICABLE	NOT APPLICABLE	118 122 130	370 620 875	21.5 8.5 0.0	- -	- -	- -	- -	- -	- -	- -	- -	- -	- -	- -	- -	- -	- -
L-14-PI	1300 1400 1400	- 875 1300	650 675 750	40 43 45	275 300 350	1800 1800 1800	10000 5000 5000	- - -	NOT APPLICABLE	NOT APPLICABLE	- -	210 400 600	28.0 10.0 0.0	- -	- -	- -	- -	- -	- -	- -	- -	- -	- -	- -	- -	- -	- -

NOTES: RED FIGURES ARE PRELIMINARY: SUBJECT TO REVISION AFTER FLIGHT CHECK.

MANUFACTURED BY VULTEE-CONSOLIDATED (WAYNE, MICH.) "VW" & PIPER (LOCK HAVEN, PA.) "PI"

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TACTICAL PLANNING CHARACTERISTICS & PERFORMANCE CHART

PAGE 43
AS OF 31 AUGUST 1945

MODEL & BLOCK-NO.	ENGINE & PROP.			SIZE		WEIGHT		COMBAT CREW	FUEL & OIL		CARGO OR PASSENGERS			RADIO	REMARKS & REFERENCE			
	NUMBER	ENGINE MFR. MODEL SUPERCHARGER PROP. MFR. TYPE	B.H.P./ALT. T - TAKE OFF W - WAR EMERG. M - MILITARY C - CONTINUOUS	S-SPAN L-LENGTH H-HEIGHT T-TREAD W-WING AREA	B-BASIC C-COMBAT W-WAR MAX. M-MAX LAND.	NOTES	FUEL TANKAGE		STATION	CAPACITY	MAX. LOAD LB.	STATION	PLACES			MAX. LOAD LB.	TYPICAL MODELS CARRIED	
							TYPE OR LOCATION ALSO (MAX. OIL)											NO AND CAPACITY ALSO (MAX. FUEL)
C-45F-BH	2	PRATT-WHITNEY R-985 AR-1 OF -3 INTEGRAL HAMILTON STD. 9'2" DIA-2 BL. F.F., CONST. SPEED	T 450/5 L W NONE M NONE C 450/5 L	S 47' 8" L 34' 3" H 9' 8" T 12' 11" W 348 Sq. Ft.	B 5800 C 7850 W 8725 M 7850		MAIN - WINGS 2 x 78 AUX. - WINGS 2 x 25 (16)	2 x 78 2 x 25 (206)	N O N E			CABIN	5	1000	SCR-274H SCR-299G RC-428 RC-35 SCR-895	C-45F TECH. ORDER 01-900D-1 (REV. 15 FEB. 1945) C-54A TECH. ORDER 01-1550B-1 (REV. 25 FEB. 1945) UC-78 SERIES TECH. ORDER 01-125-1 (REV. 20 NOV. 1944)		
C-64A-ND	1	PRATT-WHITNEY R-1340 AR-1 INTEGRAL 9'1" or 9'7" DIA. 2 BL., CONST. SPD.	T 600/5 L W NONE M NONE C 550/5000	S 51' 6" L 32' 4" H 10' 1" T 9' 1" W 325 Sq. Ft.	B 4800 C 7400 W 8000 M 7400		WINGS FRONT BELLY REAR BELLY AUX. - CABIN (20)	2 x 60 1 x 45 1 x 77 1 x 38 (280)	CABIN	1500	1500	CABIN	4 LITTERS or 6 PASS.	1200	SCR-274H SCR-578 SCR-595 SCR-895 RC-198			
UC-78 UC-78B CE UC-78C	2	JACOBS R-755-9 NONE HAMILTON STD. 7'9" DIA.-2 BL. CONST. SPEED	T 245/5 L W NONE M NONE C 225/5 L	S 41' 11" L 32' 9" H 9' 11" T 12' 7" W 295 Sq. Ft.	B 4200 C 5700 W 5700 M 5700		WINGS AUX. - FUSEL. (10)	2 x 60 1 x 30 (150)	N O N E			CABIN	3	600	SCR-283			

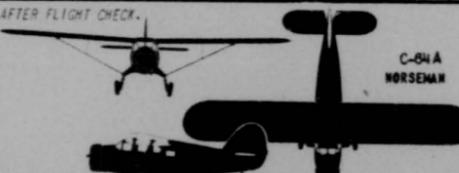
COLUMN 1 2 3 4 5 6 7 8 9 10 11 12 13 14 15 16 17 18

MODEL & BLOCK-NO.	TAKE OFF & LANDING HARD SURFACE - NO WIND					HIGH SPEED & CLIMB CLEAN AIRPLANE AT NORMAL COMBAT WEIGHT								RANGE AND ENDURANCE STATUTE AIR MILES - NO WIND - NO ALLOWANCE FOR RESERVE										REMARKS							
	GROSS WEIGHT LB.	TO CLEAR 50'		GROUND RUN LANDING SPEED MPH	T.O. DIST. FT.	LAND DIST. FT.	WEIGHT LB.	STD. ALT. FT.	WAR EMERG.		MIL. POWER		MAX. CONTIN. POWER		LOADING			MAX. CONTINUOUS POWER		MAX. CRUISE POWER		LONG RANGE									
		T.O. FT.	LAND FT.						HIGH SPEED MPH	RATE OF CLIMB FT. MIN.	HIGH SPEED MPH	RATE OF CLIMB FT. MIN.	HIGH SPEED MPH	RATE OF CLIMB FT. MIN.	TIME TO CLIMB MIN.	TAKE OFF WEIGHT LB.	BOMB CARGO PASS. LB.	TOTAL FUEL U.S. GAL.	at 5000FT. RANGE MILES	at 5000FT. ENDUR. HOURS	at 15000FT. RANGE MILES	at 15000FT. ENDUR. HOURS	at 5000FT. RANGE MILES		at 5000FT. ENDUR. HOURS	at 15000FT. RANGE MILES	at 15000FT. ENDUR. HOURS	at 10000FT. RANGE MILES	at 10000FT. ENDUR. HOURS		
C-45F-BH	5500 7000 8000 8700	- 1100 1600 2100	1700 1800 1950 -	56 58 62 -	1200 1250 1400 -	7850 15000 10000 5000 S L	25000 29000 15000 10000 5000	- -	NOT APPLICABLE						188 206 210 215 206	250 550 900 1250 1200	29.0 15.8 8.4 4.1 0.0	7850 8725	NONE 1000	206 206	500 500	2.4 2.4	475 ^a 475 ^a	2.3 2.3	675 650	3.5 3.4	650 ^a 625 ^a	3.4 3.5	850 750	4.8 4.4	(a) SPECIAL OXYGEN PROVISIONS MUST BE MADE FOR ALTITUDE FLIGHTS. (b) AT 9000 FT. ALTITUDE.
C-64A-ND	5500 6000 7000 8000	- 1250 1800 2800	1150 1200 1450 -	- - 64 -	575 625 750 -	7400 15000 10000 5000 S L	20000 15000 10000 5000	- -	NOT APPLICABLE						- 200 400 500 145	- 200 45.0 19.0 9.0 600	- -	7400 7400	400 1500	280 120	750 250	5.0 1.7	900 ^a 350 ^a	5.0 2.5	1000 450	7.7 3.5	NOT APPLICABLE	1275 500	10.7 4.2		
UC-78 UC-78B CE UC-78C	4500 5000 5500 5700	- 1325 1700 2000	- - - -	- - - -	- 625 650 725	5700 15000 10000 5000 S L	15000 10000 5000	- -	NOT APPLICABLE						144 181 171 175	220 520 850 950	30.0 15.0 6.0 0.0	5700 5700	400 600	150 120	600 475	3.6 2.8	675 500	2.9 2.3	775 575	4.8 3.6	NOT APPLICABLE	850 ^b 650 ^b	6.0 4.7		

NOTES: RED FIGURES ARE PRELIMINARY. SUBJECT TO REVISION AFTER FLIGHT CHECK.



C-45F
EXPEDITOR



C-64A
HORSEMAN



UC-78
BOBCAT

MANUFACTURED BY BEECH (WICHITA) "AN" & "ORNDORFF" (MONTREAL) "CO" & "CESSNA" (WICHITA) "CE".

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**GLIDER AND
TOWPLANE SECTION**

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FOREWORD

WARNING

THESE CHARTS CONTAIN CONSERVATIVE AVERAGES FOR TACTICAL PLANNING AND ARE NOT SUITABLE FOR AERODYNAMIC ANALYSIS.

Data contained herein are not to be considered guaranteed performance nor optimum performance as established by test flight but are to be considered official "Practical" characteristics and performance to be used for planning purposes for the average pilot.

All data includes service allowances based on theater experience. Individual towplanes may vary appreciably due to age and service changes. Data printed in red are preliminary and subject to revision after flight check. Data printed in black have been derived from information obtained in flight but are not actual flight test results. For detailed planning see Technical Orders listed as references.

RANGE and ENDURANCE are based on the following assumptions:

- (a) Allowance for warm up, taxi, run up, take off, and landing. (equal to ten minutes max. contin. at S.L.)
- (b) Allowances for fuel consumed in climb. Distance and time to climb are included in range and endurance.
- (c) Allowance for carrying bombs and droppable tanks entire flight.
- (d) Allowance of 10% net ideal range and endurance for miscellaneous differences in airplanes, equipment, pilot technique, atmospheric conditions other than wind, unusable fuel, weight and similar variables. (For example, range varies day and night due to fuel expansion prior to take off.)
- (e) Other than the above, no allowances for wind, assured reserve, combat, nor formation flight.

TYPE OF RELEASE: A number of combinations shown are not eligible for clear releases, for complete information on release characteristics see T.O. 01-1-143, dated 5 Mar. 1945.

RED PRINT: Preliminary estimates or calculated data.

RESTRICTED

DEFINITIONS

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LOADING:	Typical loadings of glider and towplane for tactical missions.
Note:	Weight and fuel quantities shown apply to all performances values to the right on the same line.
Max. tow speed:	Maximum permissible calibrated indicated air speed for towing glider at weight shown.
TAKE OFF: (AT S.L.) (To clear 50' object)	Distances representative of minimum airport requirements. (No allowance made for tow rope glider length).
Take off speed:	Normal for average pilot. (Towplane I.A.S.)
Ground run:	Distances representative of minimum runway requirements. (No allowance made for tow rope glider length).
CLIMB (I.A.S. for climb)	Minimum towplane indicated air speed for good control & satisfactory engine cooling.
Initial rate of climb:	Sea level rate of climb at minimum towplane indicated air speed.
Note:	For some combinations rate of climb tends to increase up to critical altitude of towplane.
Service ceiling:	Altitude at which rate of climb is 100'/min.
Time to service ceiling:	Time to climb in minutes from sea level to service ceiling based on I.A.S. for climb.
RANGE & ENDURANCE:	(See discussion under foreword)
Radius of action:	Distance glider can be towed and released with towplane having enough fuel to return to base alone.
Time out:	Time from take off to release of glider, (At recommended cruising altitude).
Time back:	Time required for towplane to return to base alone at recommended cruising speed and altitude.
Max. range & endurance:	Practical maximum range which glider can be towed under conditions set forth and corresponding time required.

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TOWPLANE & GLIDER TACTICAL PLANNING PERFORMANCE CHART

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AS OF:
28 JULY 1945

BOMBER TOWPLANE AND CG-4A GLIDER

TOWPLANE & GLIDER COMBINATION	LOADING						TAKE OFF (WITH GLIDER ATTACHED) S.L.-HARD SURFACE-NO WIND			CLIMB (AT MINIMUM TOW SPEED)				RANGE & ENDURANCE MAXIMUM STATUTE AIR MILES-NO WIND-NO RESERVE						REMARKS
	GLIDERS		TOWPLANE				TO CLEAR 50' OBJ. DISTANCE (FT.)	TAKE OFF SPEED I.A.S.	GROUND RUN DISTANCE (FT.)	I.A.S. FOR CLIMB (M.P.H.)	INITIAL RATE OF CLIMB (FT./MIN.)	SERVICE CEILING (FT.)	TIME TO SERVICE CEIL. (MIN.)	RADIUS OF ACTION (GLIDER RELEASED AT RADIUS)			RECOMM CRUISING ALTITUDE (FT.)	MAX. RANGE (WITH GLIDER)		
	TAKE-OFF WEIGHT (LB.)	CARGO OR PASS. (LB.)	TAKE-OFF WEIGHT (LB.)	BOMBS CARGO OR PASS. (LB.)	FUEL (U.S. GAL.)	MAX. TOW SPEED (M.P.H.)								RADIUS	TIME OUT (HRS.)	TIME BACK (HRS.)		RANGE	ENDUR.	
	(LB.)	(LB.)	(LB.)	(LB.)	(U.S. GAL.)	(M.P.H.)	(MI.)	(HRS.)	(HRS.)	(MI.)	(HRS.)									
B-17F & (1) CG-4A	7500	3800	59100	NONE	2780	150	6000	110	4000	135	300	12000*	37	(990)	6.8	6.0	5000	1380	9.1	* LIMITED- NO OXYGEN IN GLIDER
														(980)	6.3	5.5	10000	1370	8.8	
B-17F & (2) CG-4A	7500	3800	52500	NONE	1700	150	5750	110	3750	135	300	11000	57	510	3.4	3.1	5000	720	4.8	NOTE: THE 59100 LB. LOADING WITH (2) GLIDERS IS FOR EMERGENCY RE- LEASE ONLY.
	7500	3800	59100	NONE	2780	150	6600	110	4300	135	210	8000	44	860	6.0	5.2	5000	1140	7.9	
B-24D & (1) CG-4A	7500	3800	52500	NONE	2344	150	3750	115	3000	140	550	12000*	24	(870)	5.7	4.7	5000	1360	8.8	* LIMITED- NO OXYGEN IN GLIDER
	7500	3800	59000	6000	2344	150	4600	115	3650	140	325	12000*	41	(830)	5.6	4.1	10000	1280	7.8	
B-25C & (1) CG-4A	7500	3800	30000	2000	972	150	4500	120	3700	140	390	12000*	38	540	3.4	3.0	5000	700	3.8	* LIMITED- NO OXYGEN IN GLIDER NOTE: LOADINGS SHOWN ARE FOR PRO- VISIONAL RELEASE ONLY.

COLUMN— 1 2 3 4 5 6 7 8 9 10 11 12 13 14 15 16 17 18 19 20 21

NOTES:

- (1) REFERENCE MATERIAL FROM TECHNICAL ORDER 01-1-143 dated 5 MARCH 1945
- (2) ALL PERFORMANCE SHOWN WITH GLIDER TRAINING GEAR ATTACHED.

RED FIGURES ARE PRELIMINARY; SUBJECT TO
REVISION AFTER FLIGHT CHECK.

TEMPORARY TOW RELEASE MECHANISM ON B-17F CAN BE
INSTALLED QUICKLY WITH NO STRUCTURAL MODIFICATIONS REQUIRED.

TOW PLANE PERFORMANCE WITH CG-4D TO
BE SAME AS WITH CG-4A.

GLIDER CHARACTERISTICS

WEIGHT	GLIDER CHARACTERISTICS	SIZE
B-BASIC		S- SPAN
C-COMBAT		L- LENGTH
W-WAR MAX		H- HEIGHT
		T- TREAD
		W- WING AREA
	TYPICAL GLIDER CAPACITIES	
B- (3700 to 4500)	(a) TROOP CARRIER: CREW(2) & TROOPS(13).	S- 8' 5"
C- 7500	(b) CARGO CARRIER: CREW(2) & TROOPS(4) & JEEP CAR(1).	L- 48' 1/2"
W- 9000	(c) CARGO CARRIER: CREW(2) & TROOPS(3) & 75MM. HOWITZER(1) & 75MM. AMMUNITION(18 RB.)	H- 12' 7"
		T- 9' 1/2"
		W- (tab. gear) 852 sq. ft.

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TOWPLANE & GLIDER TACTICAL PLANNING PERFORMANCE CHART

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CARGO TOWPLANE AND CG-4A GLIDER

TOWPLANE & GLIDER COMBINATION	LOADING						TAKE OFF (WITH GLIDER ATTACHED) SL-HARD SURFACE-NO WIND			CLIMB (AT MINIMUM TOW SPEED)				RANGE & ENDURANCE STATUTE AIR MILES-NO WIND-NO RESERVE MAXIMUM						REMARKS	
	GLIDERS		TOWPLANE				MAX. TOW SPEED (M.P.H.)	TO CLEAR 50' OBJ. DISTANCE (FT.)	TAKE OFF SPEED I.A.S. (M.P.H.)	GROUND RUN DISTANCE (FT.)	I.A.S. FOR CLIMB (M.P.H.)	INITIAL RATE OF CLIMB (FT/MIN)	SERVICE CEILING (FT.)	TIME TO SERVICE CEIL. (MIN.)	RADIUS OF ACTION (GLIDER RELEASED AT RADIUS)			RECOMM. CRUISING ALTITUDE (FT.)	MAX. RANGE (WITH GLIDER)		
	TAKE-OFF WEIGHT (LB.)	CARGO OR PASS (LB.)	TAKE-OFF WEIGHT (LB.)	BOMBS CARGO OR PASS (LB.)	FUEL (U.S. GAL.)	RADIUS (MI.)									TIME OUT (HRS.)	TIME BACK (HRS.)	RANGE (MI.)		ENDUR. (HRS.)		
	(LB.)	(LB.)	(LB.)	(LB.)	(U.S. GAL.)	(M.P.H.)	(FT.)	(M.P.H.)	(FT/MIN)	(FT.)	(M.P.H.)	(FT/MIN)	(FT.)	(MIN.)	(MI.)	(HRS.)	(HRS.)	(MI.)	(HRS.)		
C-46A & (1)CG-4A	7500	3800	41000	NONE	1400	150	2900	85	2150	120	640	12000*	22	770	5.2	4.7	5000	1255	8.5	* LIMITED - NO OXYGEN IN GLIDER.	
	7500	3800	46000	5000	1400	150	3950	90	2900	120	490	12000*	30	715	5.3	4.3	5000	1115	8.2		
C-46A & (2)CG-4A	7500	3800	41000	NONE	1400	150	4100	85	2950	120	330	12000*	50	605	4.5	3.7	5000	850	6.3	* LIMITED - NO OXYGEN IN GLIDER. NOTE: THE 46000 LB. LOADING WITH (2) GLIDERS IS FOR EMERGENCY RELEASE ONLY.	
	7500	3800	46000	5000	1400	150	6200	90	4200	120	240	8700	43	555	4.1	3.4	5000	755	5.7		
C-47, 47A & 53 (PADDLE BLADE PROPS) & (1)CG-4A	7500	3800	26000	3000	804	150	4150	80	2550	105	515	12000*	31	575	4.8	3.6	5000	955	7.9	* LIMITED - NO OXYGEN IN GLIDER.	
	7500	3800	29000	6000	804	150	4900	85	2750	105	410	11300	37	545	4.6	3.4	5000	870	7.3		
	7500	3800	31000	8000	804	150	5450	85	2950	105	350	10400	39	520	4.4	3.1	5000	820	6.9		
C-47, 47A & 53 (PADDLE BLADE PROPS) & (1)CG-4A (TOWPLANE EQUIPPED WITH PARA-PACK RACKS)	7500	3800	26000	3000	804	150	4280	80	2650	105	465	11800	35	525	4.4	3.6	5000	870	7.3		
	7500	3800	29000	6000	804	150	5550	80	2900	105	370	10600	38	485	4.1	3.4	5000	830	6.9		
C-47, 47A & 53 (PADDLE BLADE PROPS) & (2)CG-4A	7500	3800	24200	1200	804	150	5700	80	3050	105	220	7800	46	450	4.0	2.8	2000	665	5.8	LOADINGS SHOWN ARE FOR EMERGENCY RELEASE ONLY.	
	7500	3800	26000	3000	804	150	6360	80	3250	105	180	7300	50	435	3.8	2.7	2000	630	5.5		

COLUMN— 1 2 3 4 5 6 7 8 9 10 11 12 13 14 15 16 17 18 19 20 21

NOTES:
 (1) REFERENCE MATERIAL FROM TECHNICAL ORDER 01-1-113 dated 5 MARCH 1945
 (2) ALL PERFORMANCE SHOWN WITH GLIDER TRAINING GEAR ATTACHED.
 TOW PLANE PERFORMANCE WITH CG-4A TO BE SAME AS WITH CG-4A

GLIDER CHARACTERISTICS		
WEIGHT		SIZE
B - BASIC		S - SPAN
C - COMBAT		L - LENGTH
W - WAR MAX.		H - HEIGHT
		T - TREAD
		W - WING AREA
TYPICAL GLIDER CAPACITIES		
B - (3700 to 14500)	(a) TROOP CARRIER: CREW(2) & TROOPS(13).	S - 83'8"
C - 7500	(b) CARGO CARRIER: CREW(2) & TROOPS(4) & JEEP CAR(1).	L - 48'4"
W - 9000	(c) CARGO CARRIER: CREW(2) & TROOPS(3) & 75MM. HOWITZER(1) & 75MM. AMMUNITION(18 RDS.)	H - 12'7"
		T - 9'4"
		W (Tact. gear) 852 SQ. FT.

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TOWPLANE & GLIDER TACTICAL PLANNING PERFORMANCE CHART																					
CARGO TOWPLANE AND CG-13A GLIDER																					
TOWPLANE & GLIDER COMBINATION	LOADING						TAKE OFF (WITH GLIDER ATTACHED) SL-HARD SURFACE-NO WIND			CLIMB (AT MINIMUM TOW SPEED)				RANGE & ENDURANCE MAXIMUM STATUTE AIR MILES-NO WIND-NO RESERVE						REMARKS	
	GLIDERS			TOWPLANE			MAX. TOW SPEED (M.P.H.)	TO CLEAR 50' OBJ. DISTANCE (FT.)	TAKE OFF SPEED I.A.S.	GROUND RUN DISTANCE (FT.)	I.A.S. FOR CLIMB (M.P.H.)	INITIAL RATE OF CLIMB (FT./MIN.)	SERVICE CEILING (FT.)	TIME TO SERVICE CEIL. (MIN.)	RADIUS OF ACTION (GLIDER RELEASED AT RADIUS)			RECOMM. CRUISING ALTITUDE (FT.)	MAX. RANGE (WITH GLIDER)		
	TAKE-OFF WEIGHT (LB.)	CARGO OR PASS. (LB.)		TAKE-OFF WEIGHT (LB.)	BOMBS CARGO OR PASS. (LB.)	FUEL (U.S. GAL.)									RADIUS (MI.)	TIME OUT (HRS.)	TIME BACK (HRS.)		RANGE (MI.)		ENDUR. (HRS.)
C-47, 47A & 53 PADDLE BLADE PROPS & (1) CG-13A	10000	1300		24200	1200	804	195	4100	85	2600	105	460	12000*	36	555	4.8	3.6	2000	855	7.4	
	15700	7000		24200	1200	804	195	5150	90	3100	105	300	9500	41	485	4.2	3.2	2000	725	6.3	
	17700 [†]	9000		24200	1200	804	195	5850	90	3350	105	215	8500	45	460	4.0	3.0	2000	675	5.9	
C-46A & (1) CG-13A	10000	1300		41000	NONE	1100	195	3950	85	2750	120	465	12000*	29	700	5.2	4.2	5000	1095	8.1	
	18900	10200		41000	NONE	1100	195	4650	90	3250	120	320	12000*	45	610	4.6	3.7	5000	895	6.7	

NOTES:

- (1) REFERENCE MATERIAL FROM TECHNICAL ORDER 01-1-113 dated 5 MARCH 1945 (A) TYPICAL GLIDER CONDITIONS
- (2) ALL PERFORMANCE SHOWN WITH GLIDER TRAINING GEAR ATTACHED.
- ATTENTION: WEIGHT OF GLIDER MAY BE LIMITED BY TOWPLANE PERFORMANCE; SEE T.O. 01-1-113 DEC. 1944 FOR ALLOWABLE GLIDER WEIGHT WITH CORRESPONDING TOWPLANE.
- (a) JEEPS (2) & TROOPS (8) & CREW (2)
 - (b) JEEP (1) & TROOPS (8) & CREW (2) & 75mm. HOWITZER (1) & AMMUNITION (156 RDS.)
 - (c) TROOPS (8) & CREW (2) & 75mm. HOWITZER (2) & AMMUNITION (165 RDS.)
 - (d) TROOPS (8) & CREW (2) & STD. 105mm. HOWITZER (1) (M-2) & AMMUNITION (65 RDS.)
 - (e) TROOPS (10) & CREW (2) & AERIAL DELIVERY CONTAINERS (4)
 - (f) WEAPONS CARRIER, 6x6, 1 1/2 TON (1) & CREW (2)
 - (g) JEEP (1) & TROOPS (6) & CREW (2) & 105mm. HOWITZER (1) & AMMUNITION (27 RDS.)

GLIDER CHARACTERISTICS

WEIGHT		SIZE
B - BASIC		S - SPAN
C - COMBAT		L - LENGTH
W - WAR MAX.		H - HEIGHT
		T - TREAD
		W - WING AREA
B - 8700		S - 85' 7"
C - 18900		L - 54' 3"
W - 18900		H - 20' 3"
		T - 9' 11"
		W - 873 SQ.'

TYPICAL GLIDER CAPACITIES

SEE NOTE "A" AT LEFT

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TOWPLANE & GLIDER TACTICAL PLANNING PERFORMANCE CHART

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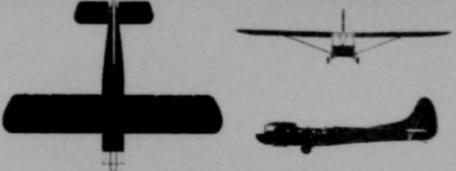
CARGO TOWPLANE AND CG-15A GLIDER

TOWPLANE & GLIDER COMBINATION	LOADING						TAKE OFF (WITH GLIDER ATTACHED) S.L.-HARD SURFACE-NO WIND			CLIMB (AT MINIMUM TOW SPEED)				RANGE & ENDURANCE MAXIMUM STATUTE AIR MILES-NO WIND-NO RESERVE						REMARKS	
	GLIDERS			TOWPLANE			MAX TOW SPEED (M.P.H.)	TO CLEAR 50' OBJ. DISTANCE (FT.)	TAKE OFF SPEED (I.A.S.)	GROUND RUN DISTANCE (FT.)	I.A.S. FOR CLIMB (M.P.H.)	INITIAL RATE OF CLIMB (FT./MIN.)	SERVICE CEILING (FT.)	TIME TO SERVICE CEIL. (MIN.)	RADIUS OF ACTION (GLIDER RELEASED AT RADIUS)			RECOMM. CRUISING ALTITUDE (FT.)	MAX. RANGE (WITH GLIDER)		
	TAKE-OFF WEIGHT (LB.)	CARGO OR PASS. (LB.)	TAKE-OFF WEIGHT (LB.)	BOMBS CARGO OR PASS. (LB.)	FUEL (U.S. GAL.)	RADIUS (MI.)									TIME OUT (HRS.)	TIME BACK (HRS.)	RANGE (MI.)		ENDUR. (HRS.)		
	(LB.)	(LB.)	(LB.)	(LB.)	(U.S. GAL.)	(M.P.H.)	(FT.)	(I.A.S.)	(FT.)	(M.P.H.)	(FT./MIN.)	(FT.)	(MIN.)	(MI.)	(HRS.)	(HRS.)	(FT.)	(MI.)	(HRS.)		
C-47 47A&53 MIDDLE BLADE PROPS & (1) CG-15A	8000	3950	26000	3000	804	180	4200	80	2700	105	485	* 12000	37	555	4.4	3.8	5000	850	6.8	* LIMITED - NO OXYGEN IN GLIDER.	
	8000	3950	29000	6000	804	180	5150	85	3200	105	365	10500	45	520	4.2	3.5	5000	775	6.2		
	8000	3950	31000	8000	804	180	5700	85	3600	105	320	9500	45	490	3.9	3.3	5000	720	5.7		
C-46A & (1) CG-15A	8000	3950	41000	NONE	1400	180	3400	85	2400	120	620	12000*	23	755	5.3	4.6	5000	1215	8.4	* LIMITED - NO OXYGEN IN GLIDER.	
	8000	3950	46000	5000	1400	180	4550	90	3200	120	480	12000*	30	715	5.3	4.3	5000	1115	8.3		

NOTES:

- (1) REFERENCE MATERIAL FROM TECHNICAL ORDER 01-1-143, DATED 5 MARCH 1945
- (2) ALL PERFORMANCE SHOWN WITH GLIDER TRAINING GEAR ATTACHED.

GLIDER CHARACTERISTICS

WEIGHT		SIZE
B-BASIC		S-SPAN
C-COMBAT		L-LENGTH
W-WAR MAX		H-HEIGHT
		T-TREAD
		W-WING AREA
TYPICAL GLIDER CAPACITIES		S- 62'3"
B- 4050		L- 48'9"
C- 8000		H- 12'7"
W- 9000	(a) TROOP CARRIER: CREW(2) & TROOPS(13). (b) CARGO CARRIER: CREW(2) & TROOPS(4) & JEEP CAR(1). (c) CARGO CARRIER: CREW(2) & TROOPS(3) & 75MM. HOWITZER(1) & 75MM. AMMUNITION(18 RUG.)	T- -
		W- 556 SQ. FT.

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